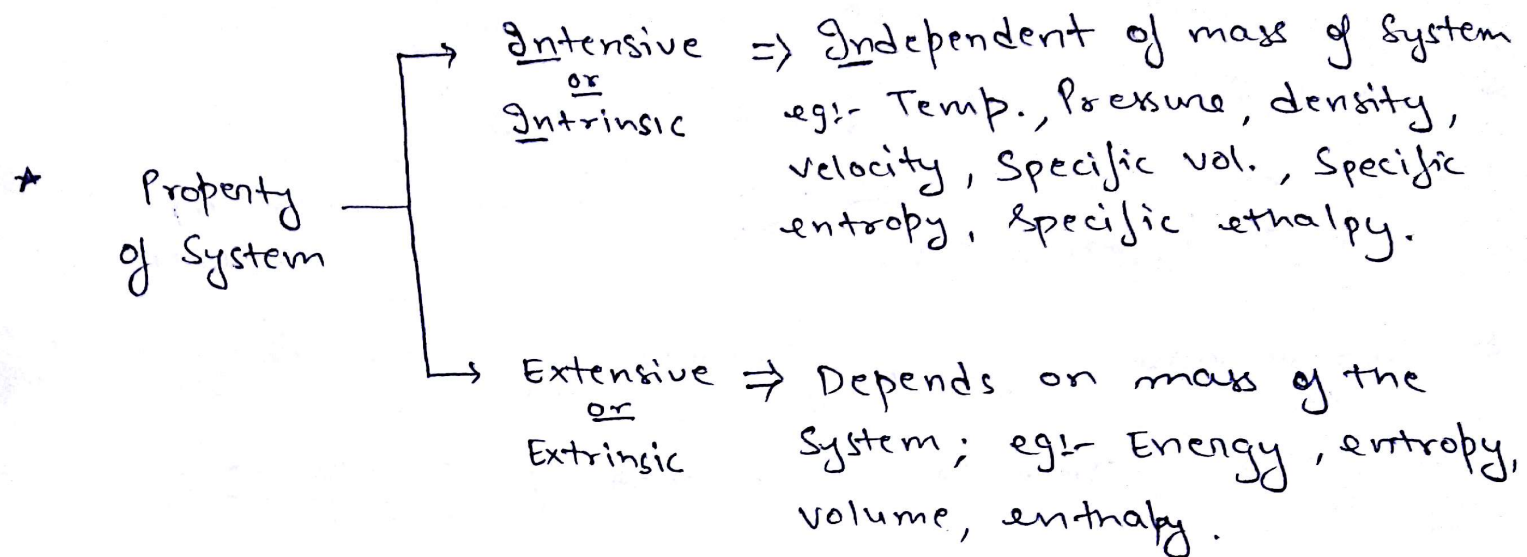


THERMODYNAMICS

* root mean square velocity = $C_{rms} = \sqrt{\frac{3RT}{M}}$ → in Kelvin
 (M) → molecular mass

Also, Pressure = $P = f(T, \rho)$



- * Property :- Point function
- State function
 - Independent of Past history
 - Exact differential

work or Heat :- Path function

- Boundary phenom
- Transient pheno.
- Depends on past history
- Inexact differential

* Continuum concept :- It is valid Till mean free Path is much much lesser than system dimension.
 Mean free Path: It is the average distance travelled by the molecule b/w two successive collisions.

* Thermodynamic equilibrium: Thermal, mechanical, Phase and chemical equilibrium.

⇒ Quasistatic compression or expansion of a gas will be considered as reversible.

⇒ Ideal flow through nozzle & diffuser etc. will be considered as reversible.

★ Pure Substance: Substance is said to be pure if it is of homogenous chemical composition throughout.

eg:- O_2 & N_2 mixture

- water + water vapour

- Ethyle Alcohol + water

★ Azeotrope: The mixture of liquids which behaves as pure substance. [R-500 Series]

★ Gibb's Phase rule: $P + F = C + 2$

$\begin{matrix} \text{no. of phase} \\ \text{[Solid, liquid, gas]} \end{matrix} \quad \leftarrow \quad \downarrow \quad \text{DOF} \quad \rightarrow \quad \begin{matrix} \text{no. of chemical} \\ \text{component} \end{matrix}$

★ Zeroth law: If $A \xrightleftharpoons[\text{equi.}]{\text{Thermal}} B \xrightleftharpoons[\text{equi.}]{\text{Thermal}} C$

Then it is sure that $A \xrightleftharpoons[\text{equi.}]{\text{Thermal}} C$

★ Thermometers

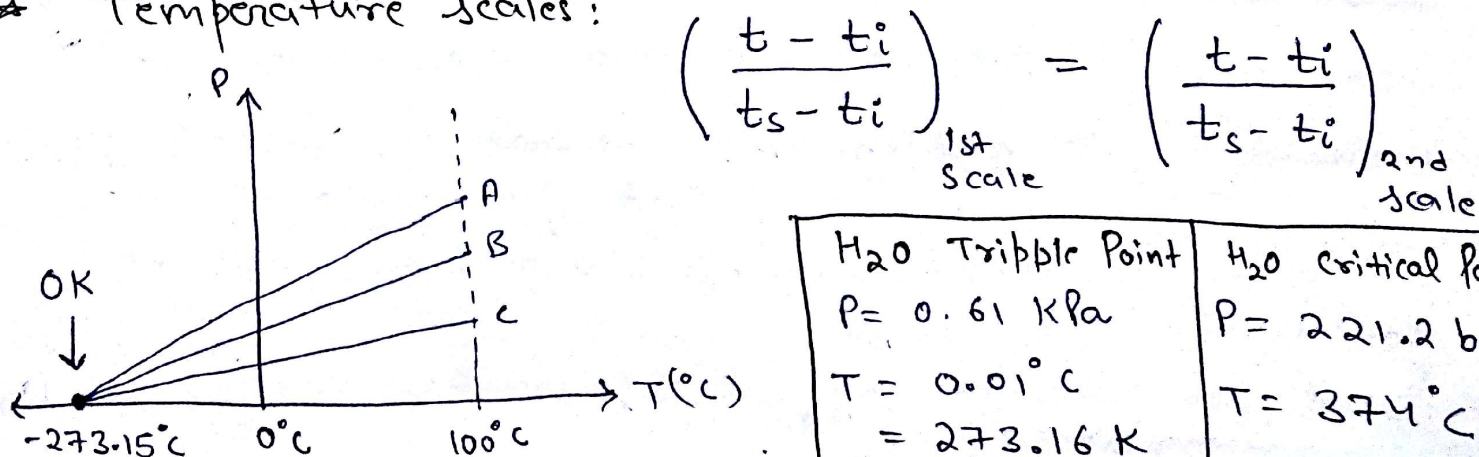
→ Resistance thermo-meter or thermister	⇒ Resistance	}	Thermo metric Property.
→ Thermocouple	⇒ Voltage, Emf		
→ Const. vol. gas Thermo.	⇒ Pressure		
→ Const. Pr. gas Thermo.	⇒ Volume		

⇒ Const. Vol. & Const. Pr. gas thermometer

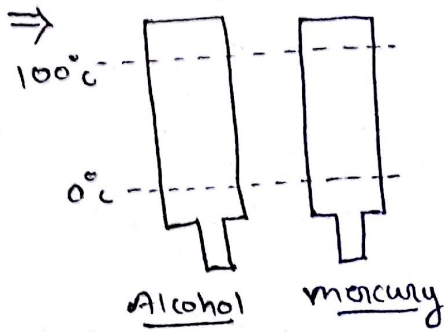
- Both are called ideal gas thermometer as Ideal gas is the material of construction.

- Both are independent of material of construction

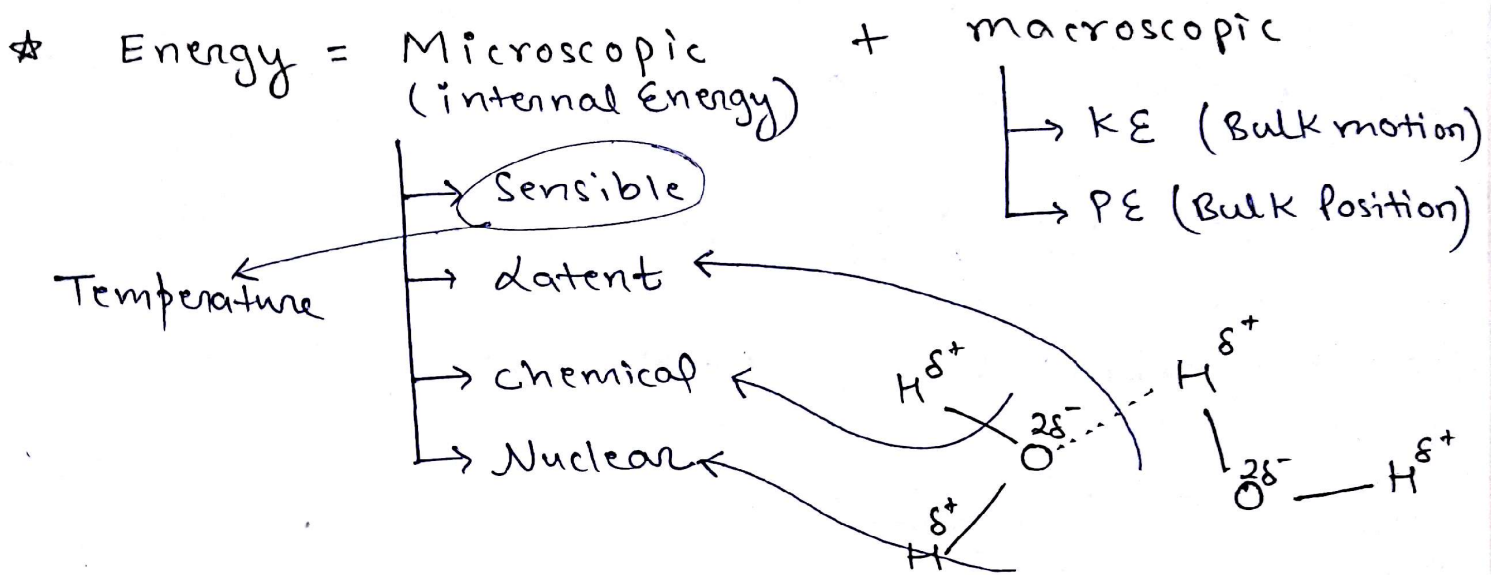
★ Temperature scales:



$\Rightarrow 1 K = \left(\frac{1}{273.16} \right)^{th}$ of Triple point of water.



$\Rightarrow$ if Both are calibrated (graded) at ice point and steam point & Distance b/w the points are divided into 100 equal parts then they are not guaranteed to give same reading b/w point of calibration, However they will give same reading at point of calibration.



★ methods of energy transfer

- Heat interaction
- work interaction
- mass interaction

★ non flow work or Boundary work or closed system work

$$\delta W = P \cdot dv$$

T & C :- closed system

- Quasistatic (Reversible)

$\Rightarrow$ Area under P-V diagram represents work interaction for a closed system.

★ non-flow work for various Process

1. constant volume or Isochoric or isometric

$$\delta W = P dv \rightarrow 0 = 0$$

2. Constant Pressure or Isobaric Process

$$\delta w = P dv \Rightarrow w = \int P dv$$

$$\Rightarrow w = P(v_2 - v_1)$$

3. Isothermal Process ($T = \text{const.}$, ^{Rectangular} hyperbolic curve)

$$w = C \ln\left(\frac{v_2}{v_1}\right) = C \ln\left(\frac{P_1}{P_2}\right) ; \text{ T \& C :- closed sys.}$$

$$C = P_1 v_1 = P_2 v_2 = mRT$$

- Reversible

- Ideal gas

4. Adiabatic Process ($Q=0$) [no electric or shaft work]

$$w = \frac{P_1 v_1 - P_2 v_2}{\gamma - 1} \quad \therefore P v^\gamma = \text{const.}$$

5. Polytropic process ($P v^n = \text{const.}$)

$$w = \frac{P_1 v_1 - P_2 v_2}{n - 1}$$

$\Rightarrow$ Isothermal process can be carried out

- in very slow process

- or when phase transfer (fast process)

$\Rightarrow$ Adiabatic - very fast

- insulate

} Two ways to conduct adiabatic process.

$\Rightarrow$ Area enclosed by cycle on P-v diagram is net work done.

- clockwise cycle $\rightarrow$ work done by system

- anticlockwise cycle $\rightarrow$ work done on system

$\Rightarrow$ Substance which exists as gases normally considered as ideal gas (eg:- O_2 , N_2 , CO_2 , He, Air, etc.)

$\Rightarrow$ water vapour should be considered ideal gas.

$\Rightarrow$ steam should not be considered as ideal gas unless mentioned

$\Rightarrow$ Vapour is special name given to gaseous form of substance when it is close to condensation.

⇒ Steam is special name given to gaseous form of only³ water, generally at high temperature.

$$\Rightarrow PV = n\bar{R}T \quad ; \quad PV = mRT$$

$$n\bar{R} = mR \quad ; \quad R = \frac{\bar{R}}{\bar{M}} \rightarrow \text{molecular mass}$$

$$1 \text{ mol of } O_2 = 32 \text{ g}$$

$$1 \text{ K-mol of } O_2 = 32 \text{ Kg}$$

$$R_c = \frac{\sum mR}{\sum m}$$

$$\bar{R} = 8.314 \text{ KJ/mol.K}$$

⇒ Const. Volume.

$$\frac{P_1}{P_2} = \frac{T_1}{T_2}$$

★ Const. Pressure

$$\frac{V_1}{V_2} = \frac{T_1}{T_2}$$

★ Isothermal $P_1V_1 = P_2V_2$

★ Adiabatic

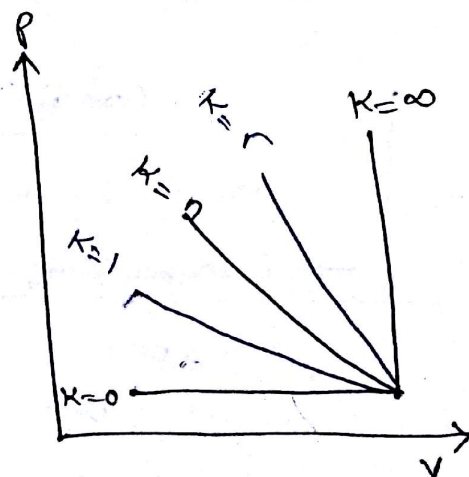
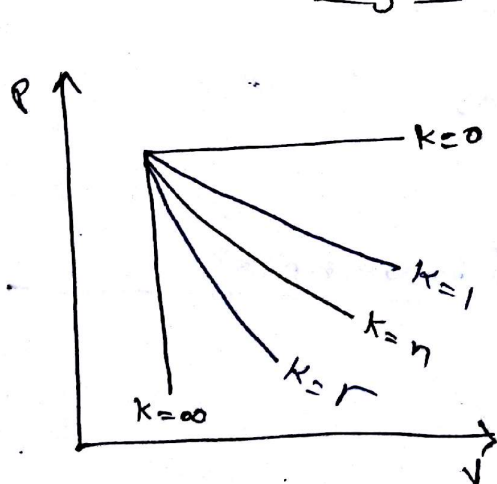
$$\frac{T_2}{T_1} = \left(\frac{P_2}{P_1}\right)^{\frac{\gamma-1}{\gamma}} \quad ; \quad \frac{T_2}{T_1} = \left(\frac{V_1}{V_2}\right)^{\gamma-1}$$

★ Polytropic

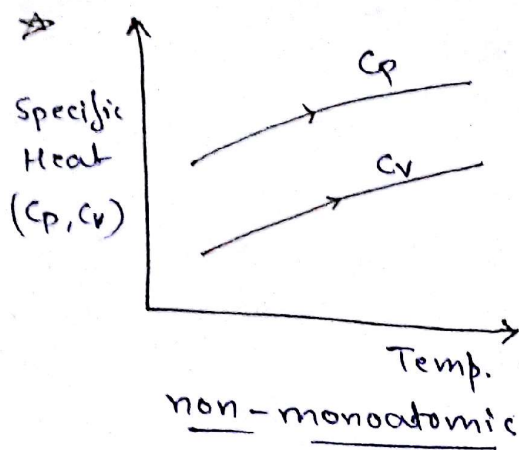
$$\frac{T_2}{T_1} = \left(\frac{P_2}{P_1}\right)^{\frac{n-1}{n}} \quad ; \quad \frac{T_2}{T_1} = \left(\frac{V_1}{V_2}\right)^{n-1}$$

⇒ $P_1V_1^n = P_2V_2^n$ should not be used for calculating 'n' for a process.

★ P-V diagram

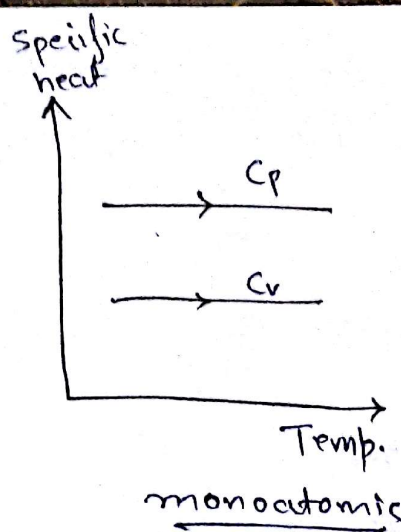


K ⇓	Process ⇓
0	Const. Pr.
1	isothermal
n	Polytropic
γ	adiabatic
∞	const. vol.



$$C_p \uparrow, C_v \uparrow$$

$$\Rightarrow \gamma \downarrow$$



$$C_p \Rightarrow \text{const.} \Rightarrow \gamma = \text{const.}$$

$$C_v = \text{const}$$

$$Q \propto m$$

$$Q \propto dT$$

$$Q = m C dT$$

↓
Specific heat
 KJ/kg-K

★ Heat supplied at const pressure is more than specific heat supplied at const. volume because C_p includes both internal energy as well as boundary work, C_v includes only internal energy.

★ First law of thermodynamics

- law of conservation of energy, Joule's law

$$\oint \delta Q = \oint \delta W \Rightarrow \text{for a closed cycle process}$$

$$- \delta Q = dE + \delta W$$

- 1st law of thermodynamics
- valid only for closed system
- undergoing process as well as cycle
- Both reversible & irreversible.

S.no.	Formula	Conditions
1.	$\delta W = P dv$	- closed system - Quasistatic (Reversible)
2.	$PV^\gamma = \text{const.}$ $\frac{T_2}{T_1} = \left(\frac{P_2}{P_1}\right)^{\frac{\gamma-1}{\gamma}}$ $\frac{T_2}{T_1} = \left(\frac{V_1}{V_2}\right)^{\gamma-1}$	- Adiabatic - Reversible - Ideal gas.

3.	$\delta Q = dE + \delta W$	- closed system
4.	$\delta Q = dU + \delta W$	- closed system & $\Delta KE = \Delta PE = 0$
5.	$\delta Q = dU + P dv$	- closed system - $\Delta KE = 0 = \Delta PE$ - Quasistatic (Reversible)

* Consequences of 1st law of thermodynamics

- Heat transfer is a Path function
- $\delta Q - \delta W = dE \Rightarrow dE$ is a property [Internal Energy]
- Energy of an isolated system is constant
- PMM-I is impossible (m/c working without E_{input})

* $1 \text{ atm} = 101.325 \text{ kPa} = 1.01325 \text{ bar}$

$1 \text{ bar} = 100 \text{ kPa}$

$\Rightarrow$ Pressure is not Energy, it only tells energy per unit volume (energy concentration)

* Heat transfer in various process

1. Const volume

$$\delta Q = dU + P dv \rightarrow 0$$

$$\delta Q = dU$$

$$dU = m c_v dT$$

$\rightarrow$ Const. vol., any Substance
 $\rightarrow$ ideal gas, any process

2. Const. Pressure

$$\delta Q = dU + P dv$$

$$\delta Q = dU + d(PV)$$

$$\delta Q = d(U + PV)$$

$$\delta Q = dH$$

$$dH = m c_p dT$$

$\rightarrow$ Const. Pr., any Substance
 $\rightarrow$ ideal gas, any process

3. Isothermal Process

$$\delta Q = \delta W + dU \rightarrow 0$$

$$\delta Q = \delta W = c \ln\left(\frac{P_1}{P_2}\right) = c \ln\left(\frac{V_2}{V_1}\right)$$

4. Adiabatic Process $\delta Q = 0$

5. Polytropic Process

$$(\delta Q)_{\text{poly}} = (W_{\text{poly}}) \times \left(\frac{\gamma - n}{\gamma - 1} \right)$$

⇒ For a constant Pressure process the maximum Boundary work for a given amount of heat is obtained when gas is monoatomic

$$\text{work ratio} = 1 - \frac{C_v}{C_p} = 1 - \frac{1}{\gamma}$$

$$\gamma = 1.67 \Rightarrow \text{monoatomic}$$

$$\gamma = 1.4 \Rightarrow \text{diatomic}$$

$$\gamma = 1.33 \Rightarrow \text{Polyatomic}$$

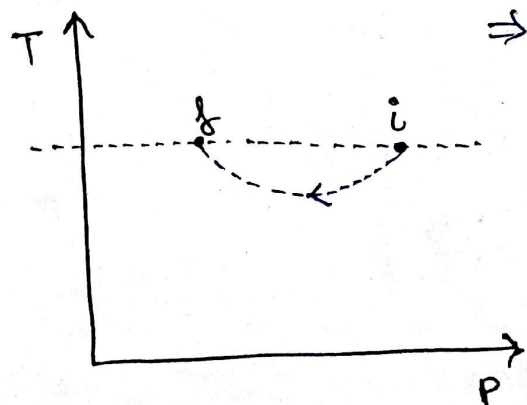
★ Meyer's equation $\Rightarrow C_p - C_v = R$

$$C_v = \frac{R}{\gamma - 1} ; C_p = \frac{\gamma R}{\gamma - 1}$$

★ Polytropic specific heat is -ve for, $1 < n < \gamma$, because even though heat is added to the gas temperature of gas decreases, since the work done by gas is more than the heat supplied.

$$\frac{Q_{\text{poly}}}{m(T_2 - T_1)} = -C_v \left(\frac{\gamma - n}{n - 1} \right)$$

★ Free expansion $\delta Q = dU + \delta W$



$$\Rightarrow dU = 0$$

$$mC dT = 0 \Rightarrow dT = 0 \Rightarrow T_i = T_f$$

$$P_i V_i = P_f V_f$$

⇒ Process is not isothermal as temperature decreases in between

★ Steady flow energy equation

$$h_1 + \frac{c_1^2}{2} + gz_1 + q = h_2 + \frac{c_2^2}{2} + gz_2 + w_{c.v.}$$

kJ/kg

J/kg [so multiplied by 10^{-3}]

→ 1st law of thermodynamics for an open system working under steady state.

★ For ideal gas

$U = mC_v T$ & $H = mC_p T$

★ SFEE applied on various device

1. nozzle: - steady flow
 - $\Delta PE = 0$
 - Adiabatic ($q=0$)
 - $w_{c.v.} = 0$

$h_1 + \frac{c_1^2}{2} = h_2 + \frac{c_2^2}{2}$

if - $c_1 \ll c_2 \Rightarrow c_2 = \sqrt{2(h_1 - h_2)}$

2. Diffuser: - steady flow
 - $\Delta PE = 0$
 - Adiabatic ($q=0$)
 - $w_{c.v.} = 0$

$h_1 + \frac{c_1^2}{2} = h_2 + \frac{c_2^2}{2}$

if - $c_2 \ll c_1 \Rightarrow c_1 = \sqrt{2(h_2 - h_1)}$

3. Turbine: - steady flow
 - $\Delta PE = 0 = \Delta KE$
 - $q = 0$

$w_{c.v.} = h_1 - h_2$

4. Compressor: - steady flow
 - $\Delta KE = 0$
 - $\Delta PE = 0$
 - $q = 0$

$w_{c.v.} = h_2 - h_1$

5. Throttling device :
- Steady flow
 - $\Delta KE = 0 = \Delta PE$
 - $W_{c.v.} = 0$
 - $Q = 0$ (adiabatic)
- $h_1 = h_2$

$\Rightarrow$ - isenthalpic
- highly irreversible

★ Unsteady state Analysis

1. Conservation of mass

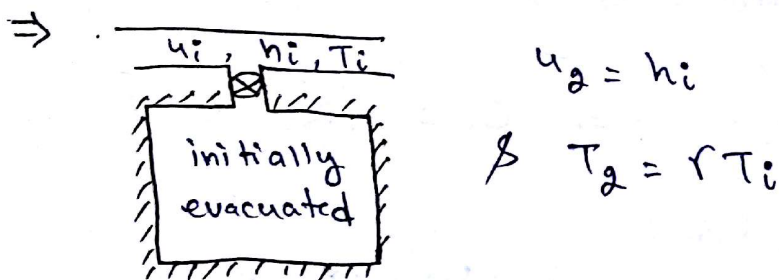
$$\left(\frac{dm}{dt}\right)_{c.v.} = \dot{m}_i - \dot{m}_e$$

2. Conservation of Energy

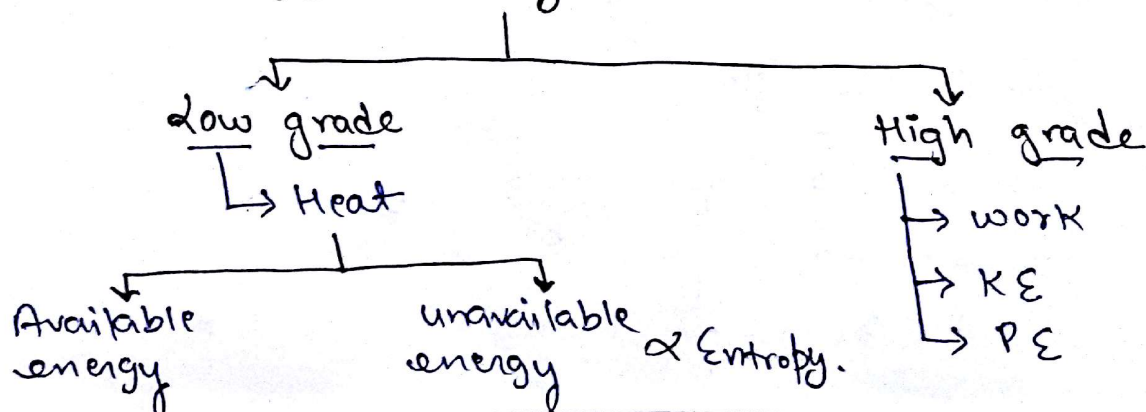
$$\left(\frac{dE}{dt}\right)_{c.v.} = \dot{m}_i \left(h_i + \frac{C_i^2}{2} + g z_i \right) + \dot{Q} - \dot{m}_e \left(h_e + \frac{C_e^2}{2} + g z_e \right) - \dot{W}_{c.v.}$$

if $\Delta KE = 0 = \Delta PE$

$$\left(\frac{dU}{dt}\right)_{c.v.} = \dot{m}_i h_i + \dot{Q} - \dot{m}_e h_e - \dot{W}_{c.v.}$$

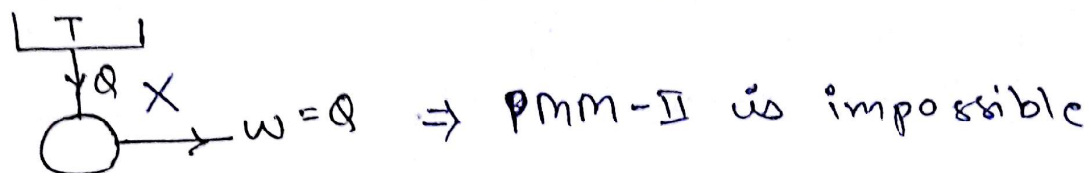


★ Energy Quality



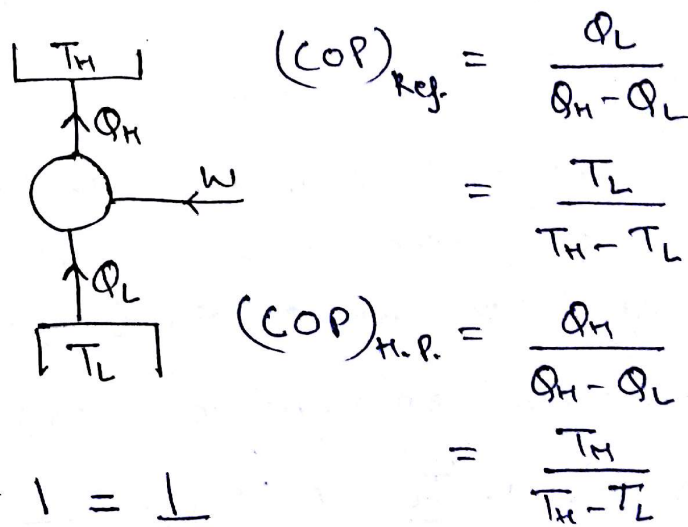
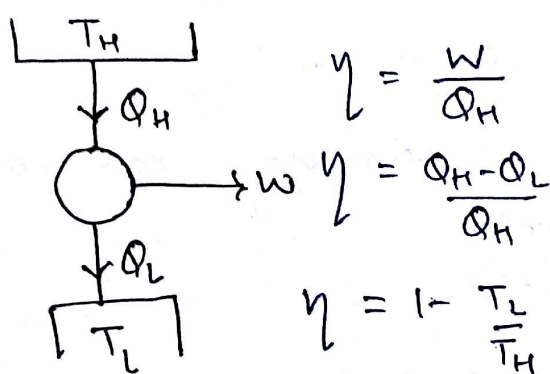
* 2nd law of thermodynamics

- Kelvin Planck: It is impossible to develop a cyclic device which develops work while exchanging energy with a single reservoir. Also an engine can never have 100% efficiency.



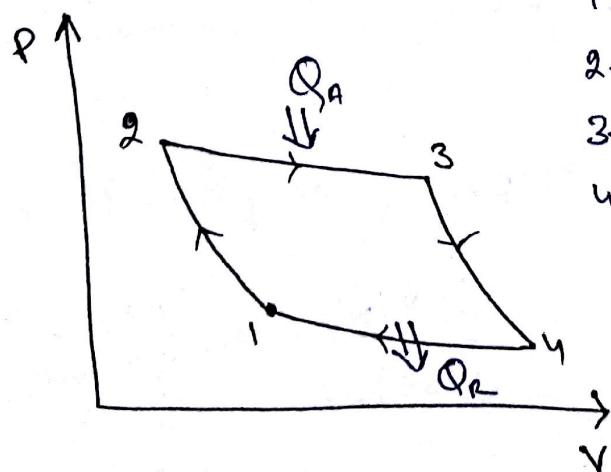
- Clausius: It is impossible to develop a cyclic device which transfer energy from lower temperature to higher temperature without any external energy input. Also, COP of a device can never be infinity.

* Violation of Kelvin plank statement leads to violation of Clausius statement and vice-versa. Hence these two statement are called parallel statement of 2nd law of thermodynamics.



$$(COP)_{HP} = (COP)_{Ref} + 1 = \frac{1}{\eta}$$

* Carnot Cycle



- 1-2 $\Rightarrow$ adiabatic compression
- 2-3 $\Rightarrow$ isothermal heat addition
- 3-4 $\Rightarrow$ adiabatic Expansion
- 4-1 $\Rightarrow$ isothermal heat rejection.

* All process are reversible

Carnot Principle

$$1^{st} : \eta_{\text{irreversible}} < \eta_{\text{reversible}}$$

$$2^{nd} : (\eta_1)_{\text{rev.}} = (\eta_2)_{\text{rev.}} = (\eta_3)_{\text{rev.}} \dots$$

- ★ above 2 principles are valid till the engines are working b/w same temp. limits.
- ★ $\eta_{\text{rev.}}$ is independent of working fluid.

★ Clausius Inequality

$$\oint \frac{\delta Q}{T} \leq 0 ;$$

$$\text{Reversible} \Rightarrow \oint \frac{\delta Q}{T} = 0$$

$$\text{irreversible} \Rightarrow \oint \frac{\delta Q}{T} < 0$$

$$\text{impossible} \Rightarrow \oint \frac{\delta Q}{T} > 0$$

$\Rightarrow$ whenever a device is shown exchanging heat from more than 2 reservoirs it can be a case of combination of cycles, hence clausius inequality should be used.

$\Rightarrow$ Temp. Appearing in Expression are temp. of working fluid

$$\Rightarrow \eta = 1 - \frac{T_L}{T_H} \quad \& \quad (\text{COP})_R = \frac{T_L}{T_H - T_L} \quad \& \quad (\text{COP})_{HP} = \frac{T_H}{T_H - T_L} \quad \text{are only}$$

valid if - cycle is reversible

- Heat addition & rejection are isothermal.

★ 2nd law Efficiency (η_{II})

for engine

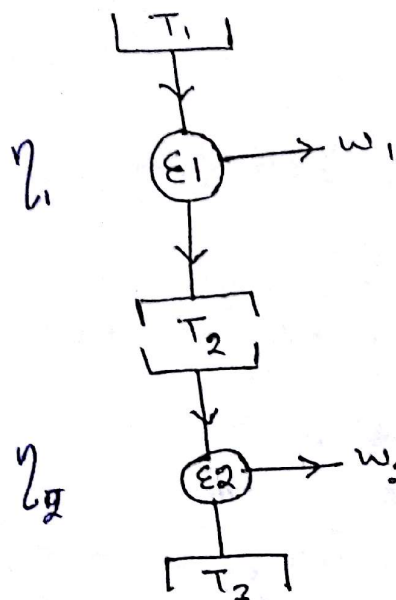
$$\eta_{II} = \frac{\eta_{\text{actual}}}{\eta_{\text{reversible}}} = \frac{W_{\text{actual}}}{W_{\text{max.}}}$$

for Refrigerator / Heat Pump

$$\eta_{II} = \frac{(\text{COP})_{\text{actual}}}{(\text{COP})_{\text{rev.}}} = \frac{W_{\text{min}}}{W_{\text{actual}}}$$

★ 2 Heat Engines in Series

7



⇒ overall efficiency of arrangement = η

$$\eta = \eta_1 + \eta_2 - \eta_1 \eta_2$$

⇒ If, work is same; $w_1 = w_2$

$$\text{then, } T_2 = \frac{T_1 + T_3}{2}$$

⇒ If, $\eta_1 = \eta_2$; then $T_2 = \sqrt{T_1 T_3}$

★ ENTROPY

- for a reversible process, $\left(\frac{\delta Q}{T}\right)_{\text{rev.}} = ds$
- for an irreversible process, $\left(\frac{\delta Q}{T}\right)_{\text{irrev.}} + (ds)_{\text{gen}} = ds$
- Entropy is a property
 - Point function
 - state function
 - Exact differential
- Entropy generation (ds_{gen}) is not a property.
- A process is said to be internally reversible if no entropy generation takes place inside the system during the process.
- Entropy generation represents degradation of energy.

★ for a cyclic device

$$\oint ds = 0 ; [\text{for both reversible \& irreversible}]$$

★ Entropy can change by 3 methods

1. Heat interaction [External interaction]
2. mass interaction [External interaction]
3. Entropy generation [internal interaction]

★ Fixed mass Entropy Analysis:

A) Reversible $[\delta s_{gen} = 0]$

$$ds = \left(\frac{\delta Q}{T}\right) + \cancel{\delta s_{gen}}$$

1. Heat addition

$$ds = \left(\frac{\delta Q}{T}\right)^{+ve} = +ve$$

2. Heat rejection

$$ds = \left(\frac{\delta Q}{T}\right)^{-ve} = -ve$$

3. Adiabatic

$$ds = \left(\frac{\delta Q}{T}\right)^0 = 0 \rightarrow \text{isentropic}$$

B) Irreversible

$$ds = \left(\frac{\delta Q}{T}\right) + (\delta s)_{gen}$$

1. Heat addition

$$ds = \left(\frac{\delta Q}{T}\right)^{+ve} + \cancel{\delta s_{gen}}^{+ve} = +ve$$

2. Heat rejection

$$ds = \left(\frac{\delta Q}{T}\right)^{-ve} + \cancel{\delta s_{gen}}^{+ve} = +ve, -ve, 0$$

isentropic $\leftarrow 0$

3. Adiabatic

$$ds = \left(\frac{\delta Q}{T}\right)^0 + \cancel{\delta s_{gen}}^{+ve} = +ve$$

★ Entropy of universe:

$\downarrow$
isolated;

$$\cancel{\delta Q}^0 = du + \cancel{\delta W}^0$$

$du = 0 \Rightarrow$ Energy remains constant.

$$ds = \left(\frac{\delta Q}{T}\right)^0 + \delta s_{gen} = +ve \Rightarrow \text{Entropy} \geq 0$$

- $\therefore$ entropy of universe can increase or remain constant but can never decrease. This is called as increase in entropy principle.

★ If we talk about same amount of energy then entropy associated is higher at lower temperature and lower at higher temperature.

- If mass is same then Entropy is higher at higher temp. & lower at lower temp. bcz total energy is also different.

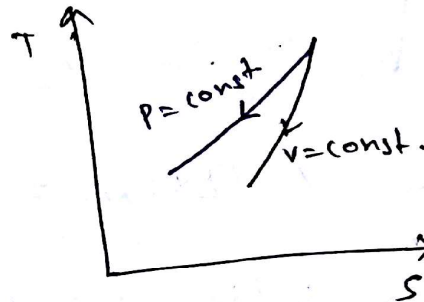
★ Area under curve on T-S diagram represents heat interaction for a reversible process.

- ★ Combined 1st law & 2nd law equations

Tds cautions

$$\textcircled{2} \quad Tds = dh - vdp$$

- * Const. Pr. $\&$ const. Vol. T-S diagram



$$\left(\frac{dT}{ds}\right)_v = \frac{T}{C_v}$$

$$\left(\frac{dT}{ds}\right)_P = \frac{T}{C_p}$$

- ★ Entropy calculation

$$ds = \frac{2}{T}$$

→ Solid or liquid of finite mass

$$\Delta S = S_2 - S_1 = mc \ln\left(\frac{T_2}{T_1}\right)$$

↳ Gas $\xrightarrow{\quad}$ Real gas (T.D. relations)

↳ Ideal gas ($Pv = RT$)

$$KJ/KgK \left\{ \begin{array}{l} 1. \Delta S = S_2 - S_1 = C_p \ln\left(\frac{T_2}{T_1}\right) - R \ln\left(\frac{P_2}{P_1}\right) \\ 2. \Delta S = S_2 - S_1 = C_v \ln\left(\frac{T_2}{T_1}\right) + R \ln\left(\frac{V_2}{V_1}\right) \\ 3. \Delta S = S_2 - S_1 = C_v \ln\left(\frac{P_2}{P_1}\right) + C_p \ln\left(\frac{V_2}{V_1}\right) \end{array} \right.$$

* Control volume Entropy Analysis:-

- Steady state

$$(\Delta E)_{c.v.} = 0$$

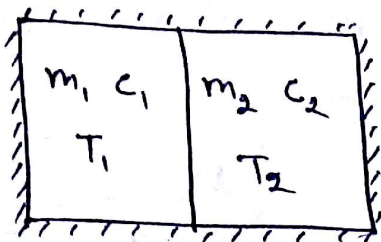
$$(\Delta m)_{c.v.} = 0$$

$$(\Delta S)_{c.v.} = 0$$

$$\begin{aligned} (\Delta S)_{c.v.}^{\circ} &= \dot{S}_{\text{entry}} - \dot{S}_{\text{exit}} + \delta \dot{S}_{\text{gen}} \\ 0 &= \dot{S}_{\text{entry}} - \dot{S}_{\text{exit}} + \delta \dot{S}_{\text{gen}} \end{aligned}$$

$$\Rightarrow \boxed{\delta \dot{S}_{\text{gen}} = \dot{S}_{\text{exit}} - \dot{S}_{\text{entry}}}$$

* mixing of two incompressible fluids (liquid) or two solids:-



$$T_f = \frac{m_1 c_1 T_1 + m_2 c_2 T_2}{m_1 c_1 + m_2 c_2}$$

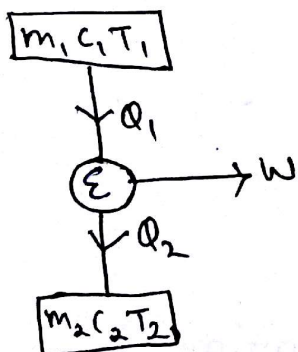
$$(\Delta S)_{\text{uni.}} = m_1 c_1 \ln\left(\frac{T_f}{T_1}\right) + m_2 c_2 \ln\left(\frac{T_f}{T_2}\right)$$

if $m_1 = m_2 = m$ & $c_1 = c_2 = c$

$$T_f = \frac{T_1 + T_2}{2} \quad \& \quad (\Delta S)_{\text{uni.}} = 2mc \ln\left(\frac{A_m}{G_m}\right) \begin{matrix} \xrightarrow{\text{of temp.}} \\ \xrightarrow{\text{of temp.}} \end{matrix}$$

for, n bodies $\Rightarrow (\Delta S)_{\text{uni.}} = nmc \ln\left(\frac{A_m}{G_m}\right)$

* Maximum obtainable work when two finite body are allowed to exchange heat through reversible engine.



$$(\Delta S)_1 + (\Delta S)_E^{\circ \text{ cyclic}} + (\Delta S)_2 = 0; \text{ [for max. work]}$$

$$m_1 c_1 \ln\left(\frac{T_f}{T_1}\right) + 0 + m_2 c_2 \ln\left(\frac{T_f}{T_2}\right) = 0$$

if $m_1 = m_2 = m$ & $c_1 = c_2 = c$

$$T_f = \sqrt{T_1 T_2}$$

$$W_{\text{max.}} = 2mc(A_m - G_m)$$

for n bodies, $W_{\text{max.}} = nmc(A_m - G_m)$

* Reversible work in a steady flow process
assumptions - 1. Steady flow

2. $\Delta KE = 0 = \Delta PE$

3. reversible

$$W_{c.v.} = - \int v dp$$

★ In a gas cycle for non adiabatic process ($Pv^n = c$) the

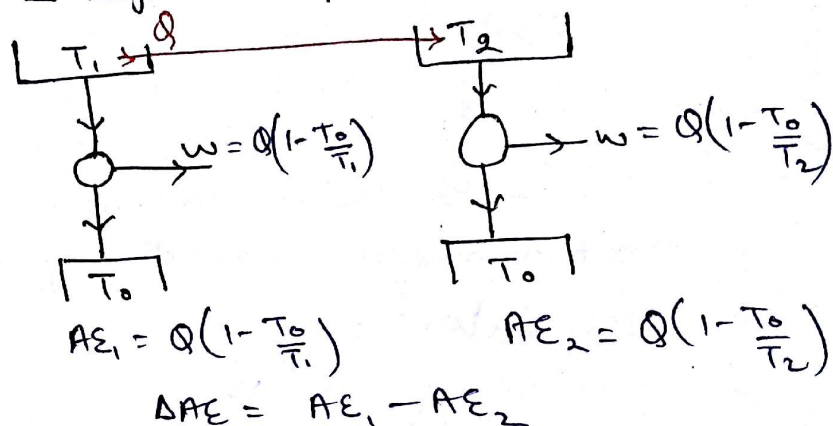
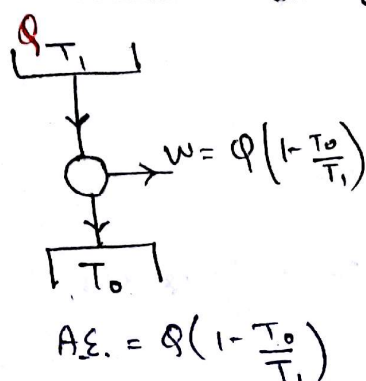
$$w_{c.v.} = - \int v dp \quad \text{or} \quad w_{c.v.} = \left(\frac{n}{n-1} \right) (P_1 v_1 - P_2 v_2)$$

⇒ Available Energy or Exergy or Availability

Available energy is the maximum useful work which can be obtained in a process in which system attains dead state.

→ Available energy is a function of - Energy of system
- state of system
- state of surroundings

① Thermal energy at a single temperature



② Thermal energy with a given mass of solid or liquid

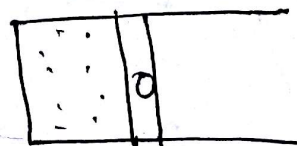
$$\boxed{\phi = u - T_0 s} \quad ; \quad \begin{aligned} \phi_1 &= u_1 - T_0 s_1 \\ \phi_2 &= u_2 - T_0 s_2 \end{aligned} \quad \left| \quad \begin{aligned} \Delta \phi &= \phi_1 - \phi_2 \\ &= (u_1 - u_2) - T_0 (s_1 - s_2) \end{aligned} \right.$$

③ Non-flow gases

$$\boxed{\phi = u - T_0 s + p_0 v}$$

$$\Delta \phi = \phi_1 - \phi_2 = \underbrace{(u_1 - u_2) - T_0 (s_1 - s_2)}_{\text{max. work}} + \underbrace{p_0 (v_1 - v_2)}_{\text{W}_{atm.}}$$

↓
max. useful work



④ flowing fluids

$$\boxed{\phi = h - T_0 s}$$

★ Since K.E & P.E. are totally available energy hence they can be directly added in the available energy expressions.

Irreversibility (I)

$$\begin{aligned} I &= \text{Loss of A.E.} \\ &= \text{increase in U.A.E} \\ &= W_{\text{max.}} - W_{\text{actual}} \end{aligned}$$

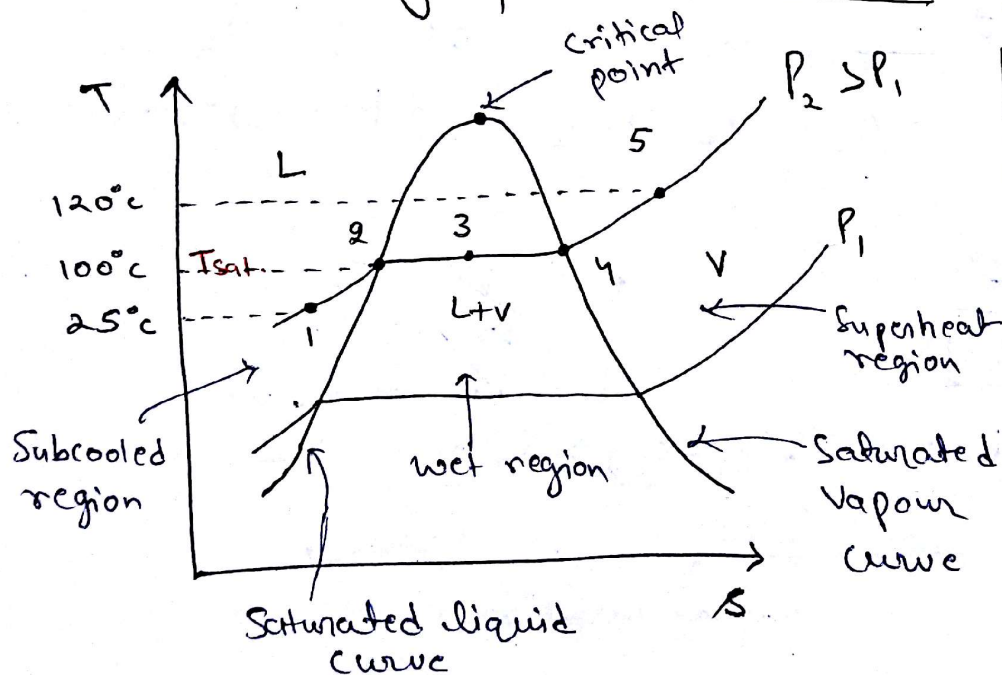
* Gouy - Stodala theorem

$$I_{\text{total}} = T_0 (\Delta S)_{\text{uni.}}$$

- Causes of irreversibility

- Conversion of high grade energy into low grade energy
 - friction
 - viscous forces
 - Electric resistance
 - Inelastic collision
- Heat transfer through finite temp. differences.
- Free Expansion
- mixing of fluids.

Properties of pure substance



$$\begin{aligned} \text{Degree of subcooling} \\ &= T_{\text{sat.}} - T_1 \end{aligned}$$

$$\begin{aligned} \text{Degree of superheating} \\ &= T_s - T_{\text{sat.}} \end{aligned}$$

$$\text{Dryness fraction (x)}$$

$$x = \frac{m_v}{m_l + m_v}$$

m_v = mass of vapour

m_l = mass of liquid

- * As the pressure increase latent heat of vapourisation decreases and boiling point increases. L.H. becomes zero at critical point
- * At critical point Dryness fraction is undefined & DOF = 0
- * only vapour exists above critical point.

- During phase change pressure & temp. are dependent property.

Expressions for wet region

$$h = h_f + x h_{fg}$$

$$v = v_f + x v_{fg}$$

$$s = s_f + x s_{fg}$$

$$u = u_f + x u_{fg}$$

$$\frac{1}{\beta} = \frac{1}{\beta_f} + x \left(\frac{1}{\beta_g} - \frac{1}{\beta_f} \right)$$

$$h_i = h_g - c_{p_e} (T_{sat.} - T_i)$$

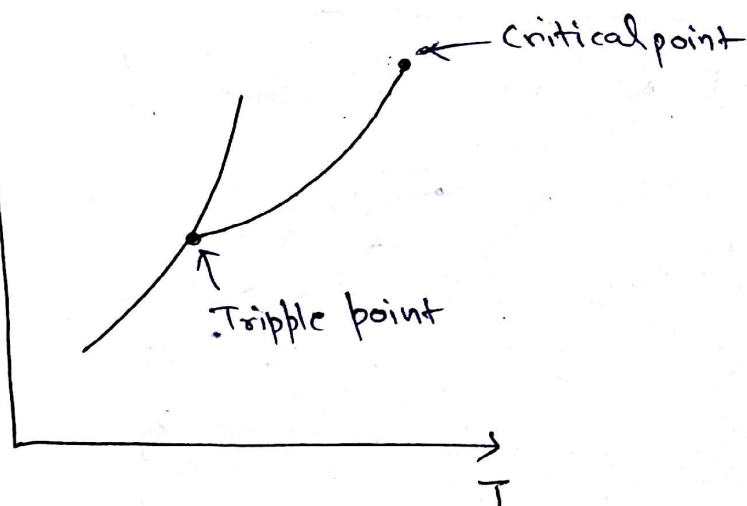
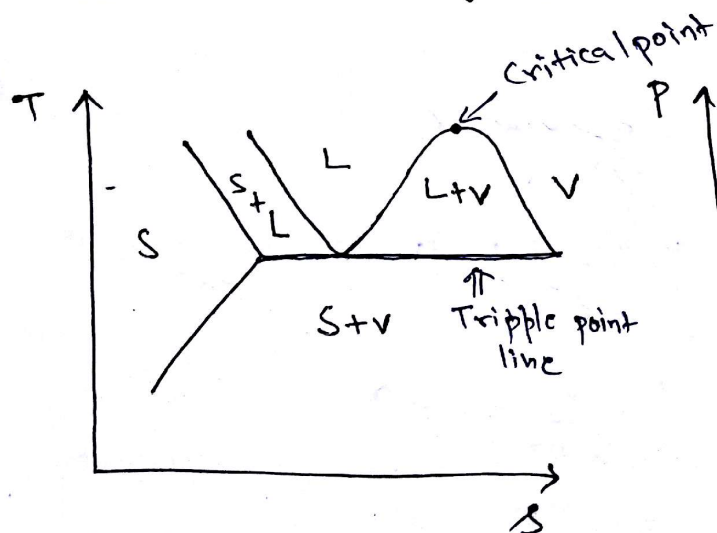
$$h_s = h_g + c_{p_v} (T_s - T_{sat.})$$

$$s_i = s_g - c_{p_e} \ln \left(\frac{T_{sat.}}{T_i} \right)$$

$$s_s = s_g + c_{p_v} \ln \left(\frac{T_s}{T_{sat.}} \right)$$

- ★ Heat supplied at const. Pressure is change in enthalpy. Heat supplied at const. volume is change in internal energy.
- ★ If steam is heated in rigid container then energy supplied is change in internal energy, not enthalpy.
- ★ A common mistake is not multiplying by mass. ($m = \frac{v}{v_g}$)
- ★ In steam table reference point is tripple point of water
 $\left. \begin{array}{l} u = 0 \\ s = 0 \end{array} \right\}$ Considered.

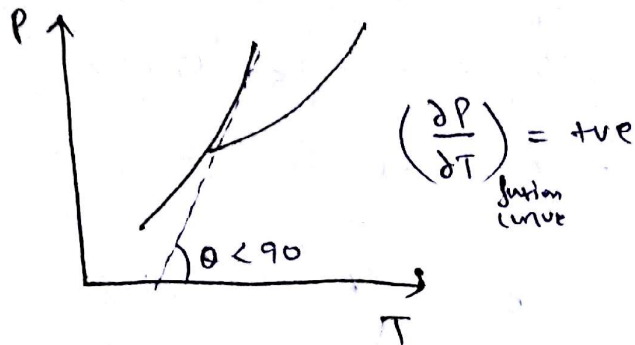
Pressure temp. diagram



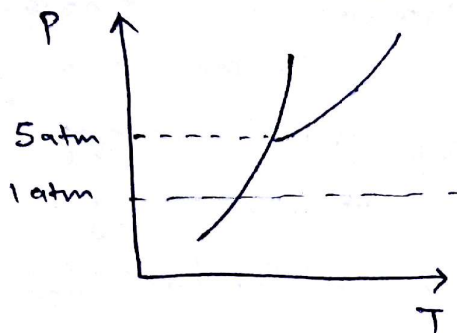
- out of 15 possible diagrams $[P, v, T, u, h, s; \phi_2 = 15]$ if diagram is P-T then phase change region is absent and tripple point is a point.
- on all other diagram phase change region is present and tripple point is line
- Critical point is a point on all diagram.

General Substance

$S \rightarrow L \Rightarrow \text{Expand}$
 $L \rightarrow S \Rightarrow \text{Contract}$

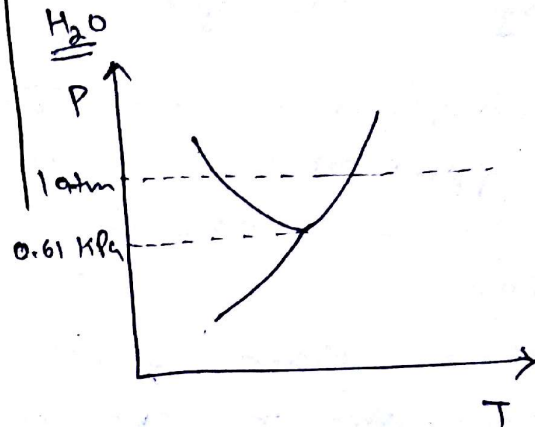
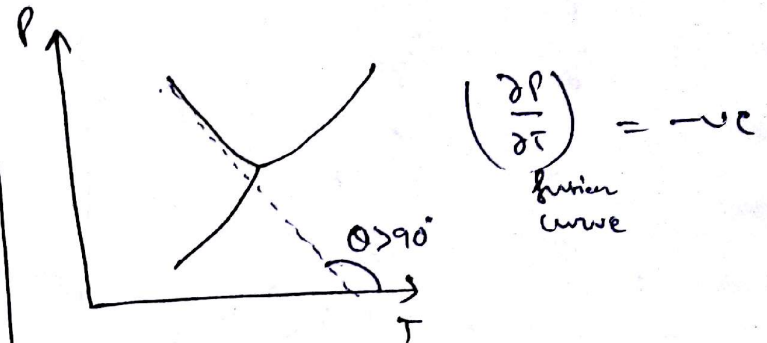


eg:- CO_2 [dry ice.]



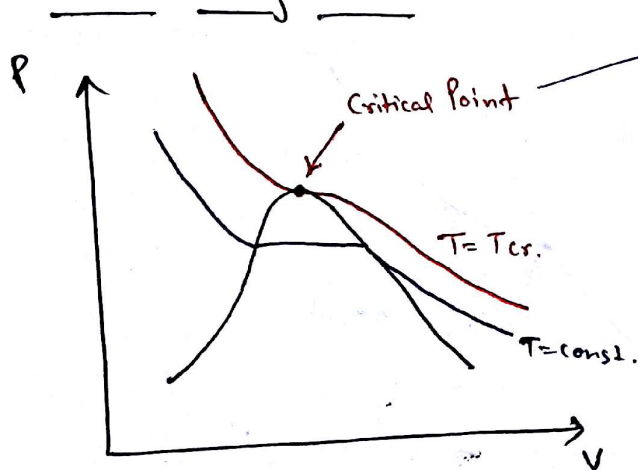
Water (H_2O)

$S \rightarrow L \Rightarrow \text{Contract}$
 $L \rightarrow S \Rightarrow \text{Expand}$



- If substance is kept below tripple point ~~temperature~~ Pressure then it will directly convert to vapour (from solid) on heating.

P-V diagram



$$\left(\frac{\partial P}{\partial V}\right) = 0$$

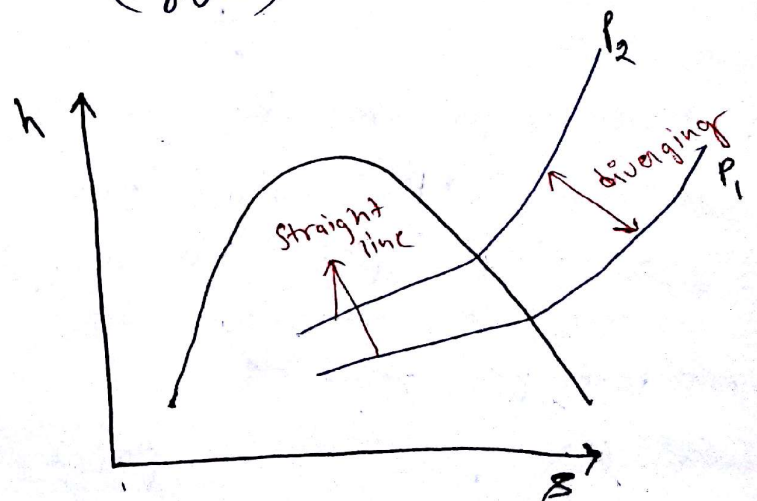
$$\left(\frac{\partial^2 P}{\partial V^2}\right) = 0$$

$$\left(\frac{\partial^3 P}{\partial V^3}\right) < 0$$

Mollier diagram

$$\left(\frac{dh}{ds}\right)_P = T$$

$$\text{Slope} = T$$



Real Gases

11

* compressibility factor = $z = \frac{V_a}{V_i}$

→ $z = 1 \Rightarrow$ real gas behaving as ideal gas

→ $z > 1 \Rightarrow V_a > V_i \Rightarrow$ repulsion dominate $\Rightarrow$ Difficult to compress as compared to ideal gas

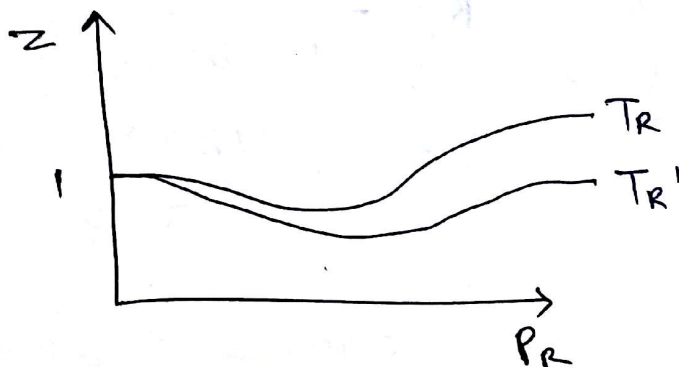
→ $z < 1 \Rightarrow V_a < V_i \Rightarrow$ attraction dominate $\Rightarrow$ easier to compress as compared to ideal gases.

* Different Gases behaves differently at same temperature and pressure but there behaviour is almost same at same Reduced Pressure & Reduced temperature.

$$P_R = \frac{P}{P_{cr.}}$$

$$T_R = \frac{T}{T_{cr.}}$$

$\Rightarrow$ compressibility chart



$$T_R > T_R'$$

$\Rightarrow$ Vander waal's equation

$$\left(P + \frac{n^2 a}{V^2}\right)(V - nb) = nRT$$

for $n=1$

$$\left(P + \frac{a}{V^2}\right)(V - b) = RT$$

$$V = \text{m}^3/\text{mol.}$$

$$\left(P + \frac{a}{V^2}\right)(V - b) = RT$$

$$V = \text{m}^3/\text{Kg}$$

$$a = \frac{27 R^2 T_{cr.}^2}{64 P_{cr.}}$$

$$b = \frac{RT_{cr.}}{8 P_{cr.}}$$

Thermodynamic Relations

- Gibbs's function $\Rightarrow G = H - TS$

- Helmholtz's function $\Rightarrow F = U - TS$

maxwell's equation

$$\begin{array}{cc|cc} T & P & \left(\frac{\partial P}{\partial S}\right)_V = -\left(\frac{\partial T}{\partial V}\right)_S & \left(\frac{\partial T}{\partial P}\right)_S = \left(\frac{\partial V}{\partial S}\right)_P \\ V & S & \left(\frac{\partial V}{\partial T}\right)_P = -\left(\frac{\partial S}{\partial P}\right)_T & \left(\frac{\partial P}{\partial T}\right)_V = \left(\frac{\partial S}{\partial V}\right)_T \end{array}$$

Tds Partial differential Equⁿ

1. $Tds = C_v dT + T\left(\frac{\partial P}{\partial T}\right)_V dV$
2. $Tds = C_p dT - T\left(\frac{\partial V}{\partial T}\right)_P dP$

Mayer's Equation

$$C_p - C_v = -T\left(\frac{\partial P}{\partial V}\right)_T \left[\left(\frac{\partial V}{\partial T}\right)_P\right]^2$$

$$C_p - C_v = \frac{TV\beta^2}{\alpha_T}$$

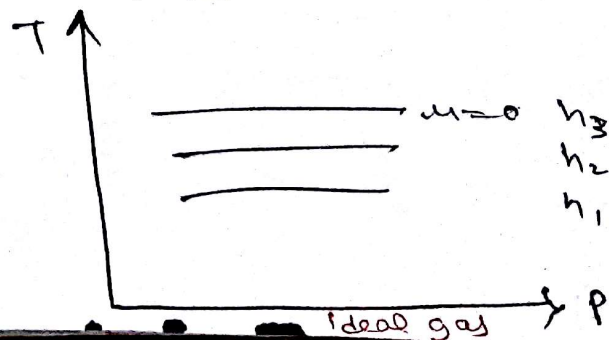
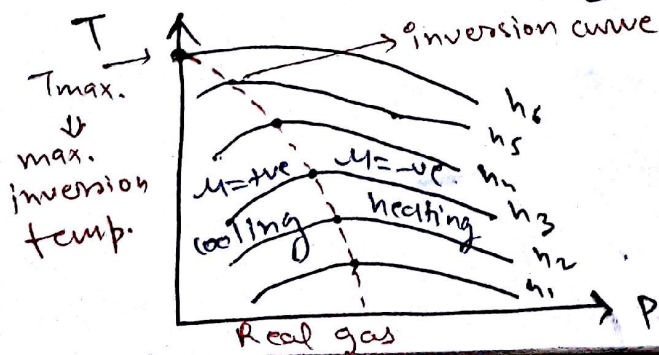
Energy equation

$$dU = C_v dT + \left[T\left(\frac{\partial P}{\partial T}\right)_V - P\right]dV$$

Joule Thomson coefficient (μ)

$$dh = C_p dT - \left[T\left(\frac{\partial V}{\partial T}\right)_P - V\right]dP$$

$$\mu = \left(\frac{\partial T}{\partial P}\right)_h = \frac{1}{C_p} \left[T\left(\frac{\partial V}{\partial T}\right)_P - V\right] = 0 \text{ for ideal gas}$$



Clapeyron equation

$$\frac{dP}{dT} = \frac{L.H.}{T_{sat.}(V_g - V_f)}$$

$$\alpha_T = -\frac{1}{V} \left(\frac{\partial V}{\partial P}\right)_T$$

$$\beta = \frac{1}{V} \left(\frac{\partial V}{\partial T}\right)_P$$

Clausius Clapeyron equⁿ

$$\frac{dP}{dT} = \frac{L.H. \cdot P_{sat}}{T_{sat}^2 \times R}$$

$$\int \frac{dP}{P} = \frac{LH}{R} \int \frac{dT}{T^2}$$