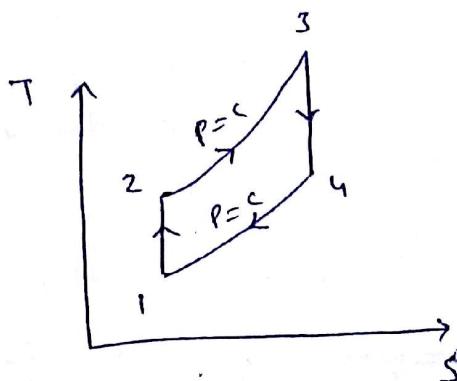
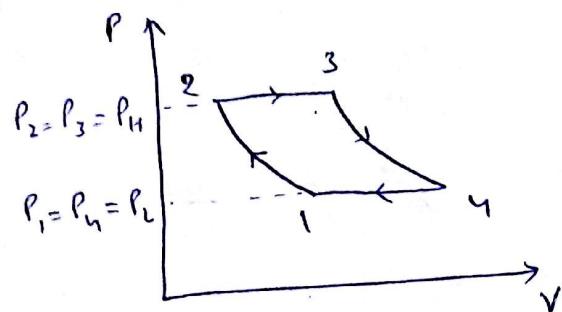


POWER PLANT

CHAPTER-1 [SIMPLE GAS TURBINE CYCLE, BRAYTON CYCLE]

- Brayton cycle



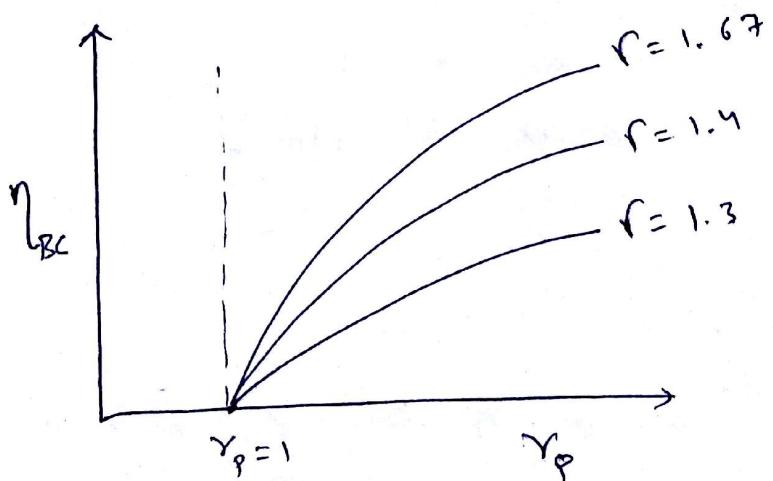
$$T_1, T_3 = T_2, T_4$$

Always

- Back work ratio (γ_{bw}) = $\frac{-ve \text{ work}}{+ve \text{ work}} = \frac{W_c \text{ or } W_p}{W_T}$

$$\text{work ratio} = 1 - \gamma_{bw} = \frac{W_T - W_c}{W_T}$$

- $\eta_{\text{Brayton}} = 1 - \frac{T_1}{T_2} = 1 - \frac{1}{\frac{T_2}{T_1}} = 1 - \frac{1}{(\gamma_p)} \frac{r-1}{r}$



- Optimum pressure ratio for max. work

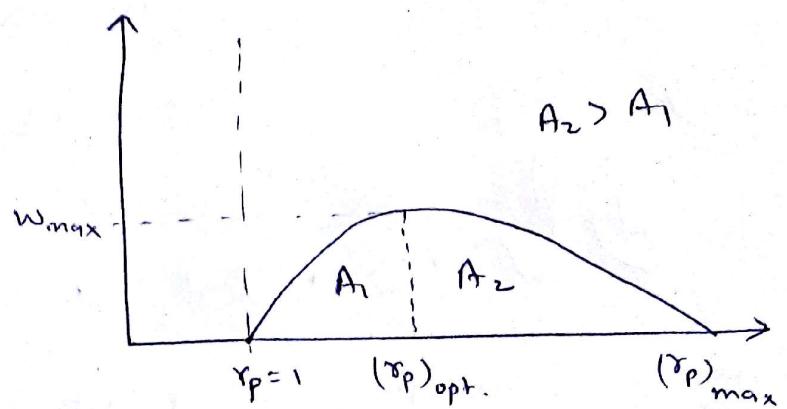
$$T_2 = T_4 = \sqrt{T_1, T_3}$$

$$W_{\max} = C_p \left[\sqrt{T_{\max}} - \sqrt{T_{\min}} \right]^2$$

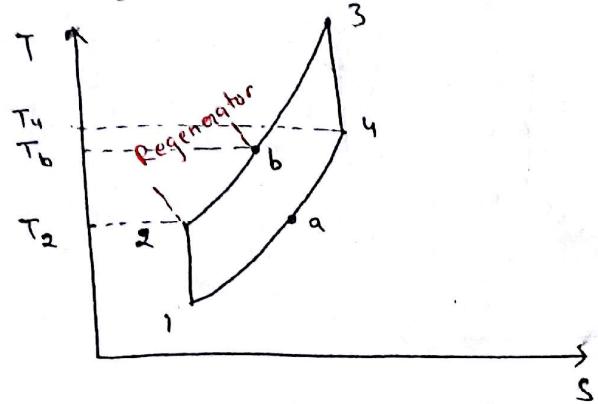
here $T_{\max} = T_3$
 $T_{\min} = T_1$

$$(\gamma_p)_{\text{opt}} = \left(\frac{T_{\max}}{T_{\min}} \right)^{\frac{1}{2}(r-1)}$$

$$\circ (\gamma_p)_{\max} = (\gamma_p)_{\text{opti}}^2$$



• Regeneration in Gas turbine



$$W_{\text{net}} = \text{const}$$

$$Q_s \downarrow \Rightarrow \eta \uparrow$$

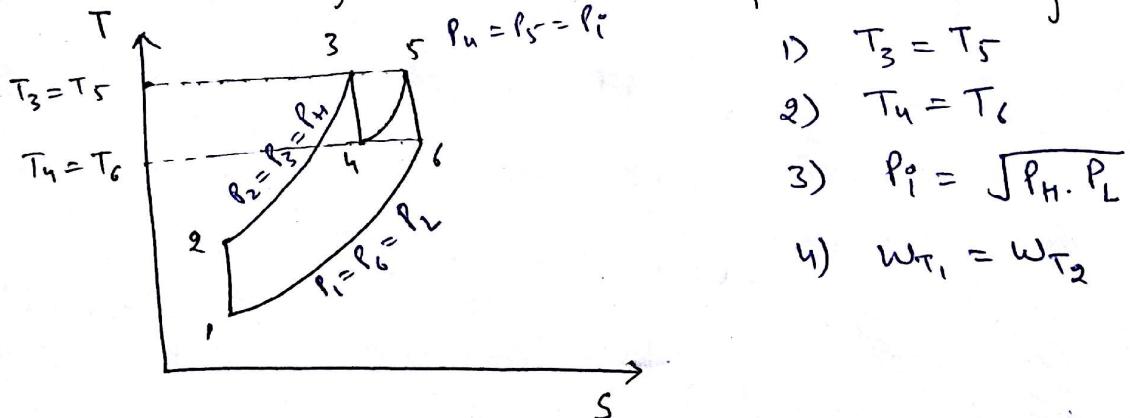
effectiveness of regenerator

$$\epsilon = \frac{T_b - T_a}{T_b - T_2}$$

for ideal regenerator $T_b = T_4 \ \& \ T_2 = T_a$

$$\eta_{\text{reg. cycle}} = 1 - \frac{T_1}{T_3} (\gamma_p)^{\frac{r-1}{r}}$$

• Condition for max work output with Perfect reheat



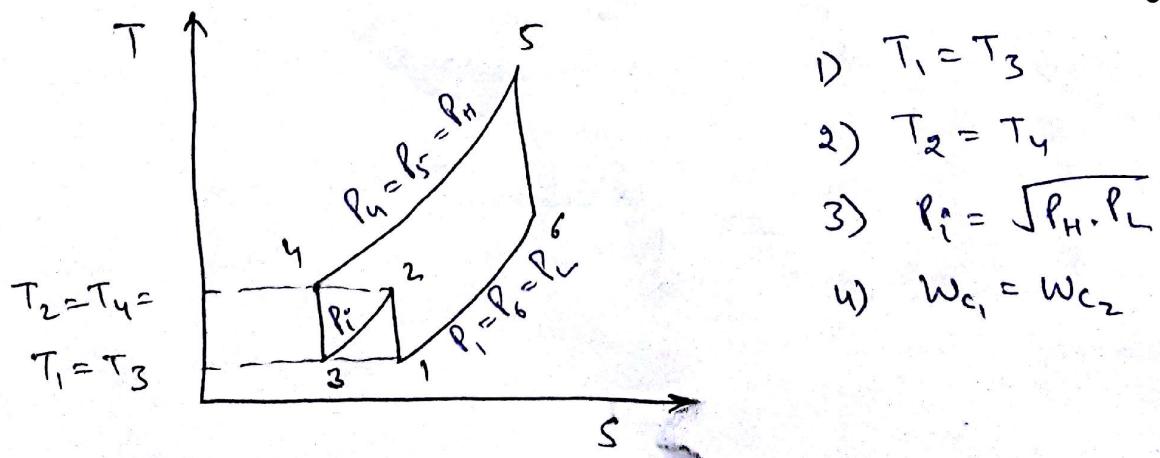
$$1) T_3 = T_5$$

$$2) T_4 = T_6$$

$$3) P_i = \sqrt{P_H \cdot P_L}$$

$$4) W_{T_1} = W_{T_2}$$

• Condition for mini. work input with Perfect intercooling



$$1) T_1 = T_3$$

$$2) T_2 = T_4$$

$$3) P_i = \sqrt{P_H \cdot P_L}$$

$$4) W_{c_1} = W_{c_2}$$

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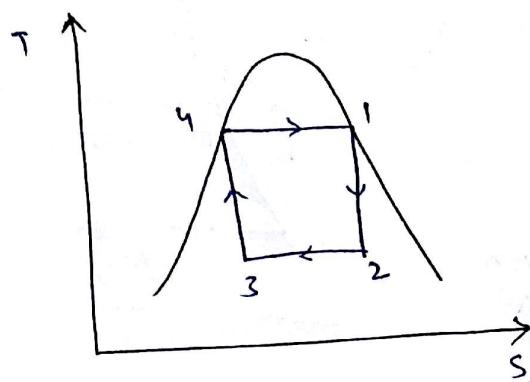
CHAPTER - 2 [RANKINE CYCLE] or steam Power cycle

- Specific steam consumption

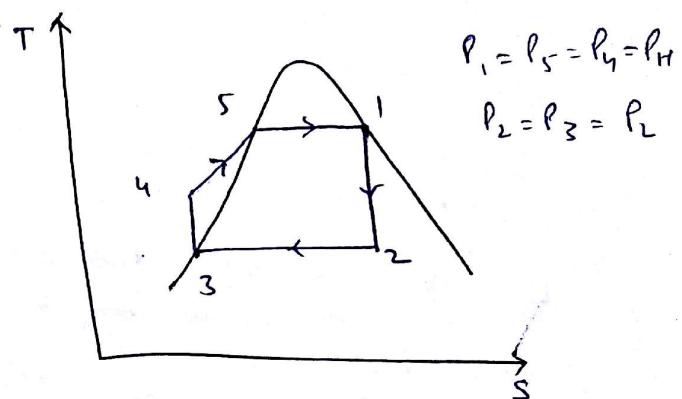
$$SSC = \frac{3600}{W_{net}} \frac{\text{kg}}{\text{kW hr}}$$

- Heat rate = $\frac{1}{\eta}$

- Cannot vapour cycle



- Rankine cycle



$1-2 \Rightarrow$ turbine
 $2-3 \Rightarrow$ condenser
 $3-4 \Rightarrow$ pump
 $4-1 \Rightarrow$ Boiler

- mean temp^o of Heat addition is higher in Carnot cycle so η_{Carnot} is more than $\eta_{Rankine}$.

- methods to improve performance of Rankine cycle

1) Decreasing condenser Pressure

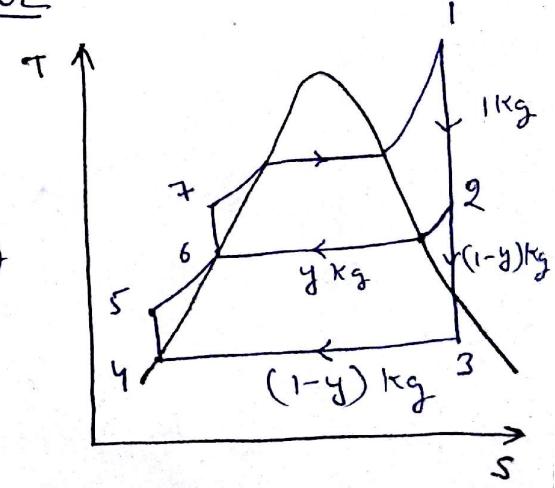
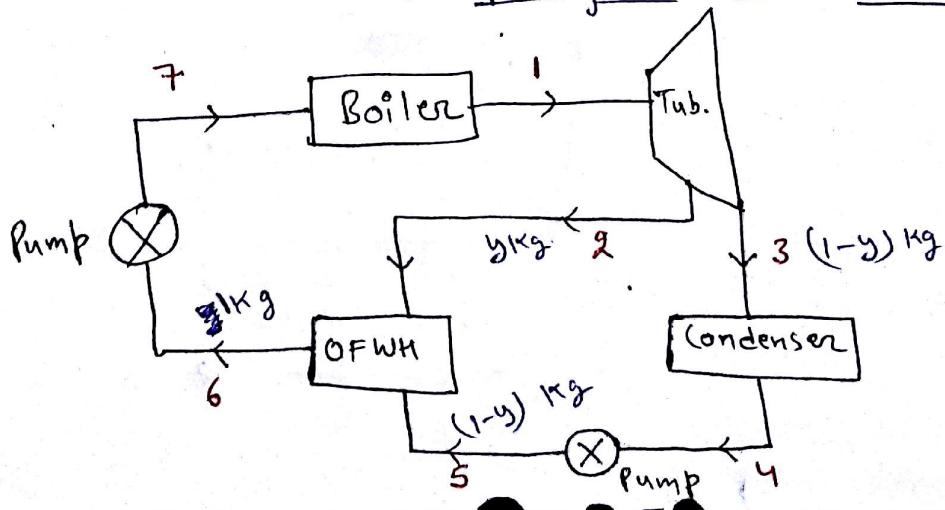
2) increasing boiler pressure

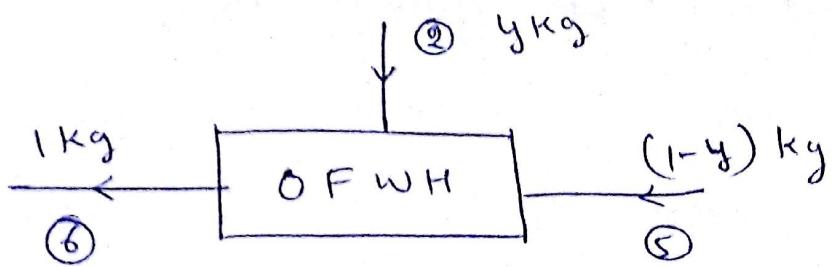
3) Super heating] the main aim is to increase dryness fraction

4) Reheating] at outlet of turbine

5) Regeneration

a) Open feed water Heater

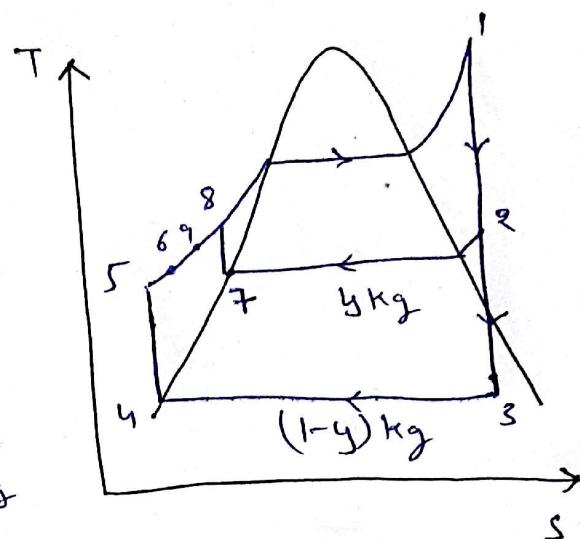
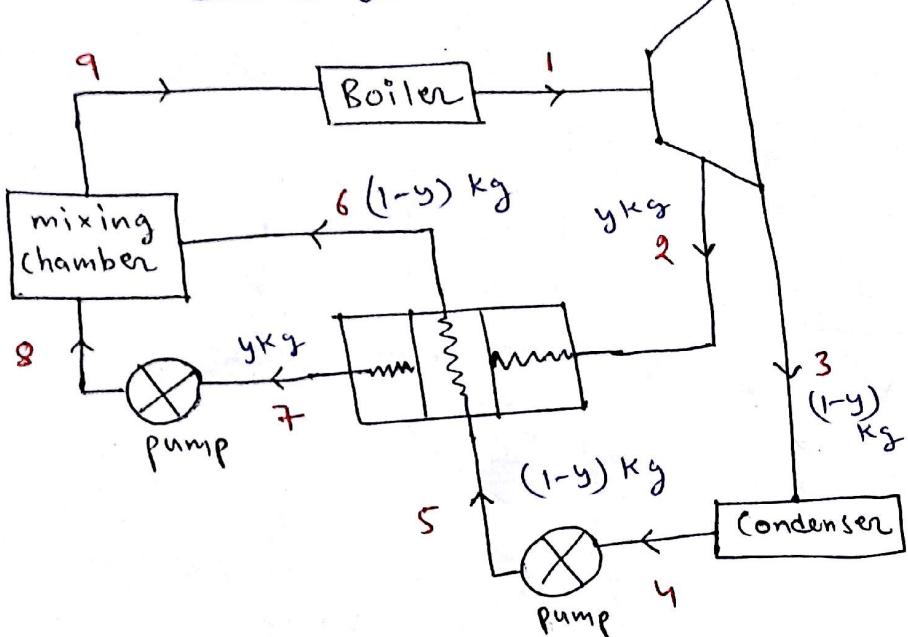




$$y h_2 + (1-y) h_5 = (1) h_6$$

$$m_2 + m_5 = m_6$$

b) Closed feed water heater



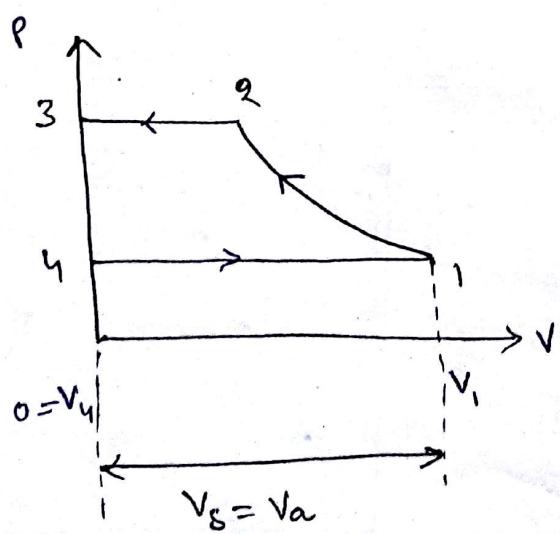
- Position of point ② will be decided by x_2 ; by equating $s_1 = s_2$

$$y h_2 + (1-y) h_5 = (1-y) h_6 + y h_7$$

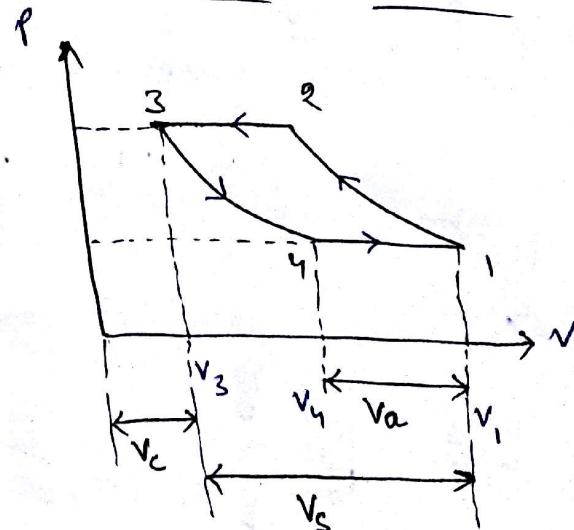
CHAPTER- 3 [RECIPROCATING COMPRESSOR]

- High Pressure upto 1000 bar
- Low Discharge, 5-8 m^3/min .

Work without clearance



Work with clearance



$$\Rightarrow W_{in} = \left(\frac{n}{n-1} \right) P_1 \times V_a \times \left[\left(\frac{P_2}{P_1} \right)^{\frac{1}{n}} - 1 \right] \text{ J/cycle}$$

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$V_a = V_i - V_u = V_i$ for comp. without clearance

= $V_i - V_u$ for comp. with clearance

= actual volume swept.

- Clearance ratio = $c = \frac{V_c}{V_s}$

- Volumetric efficiency

$$\eta_v = 1 + c - c \left(\frac{P_H}{P_L} \right)^{1/n}$$

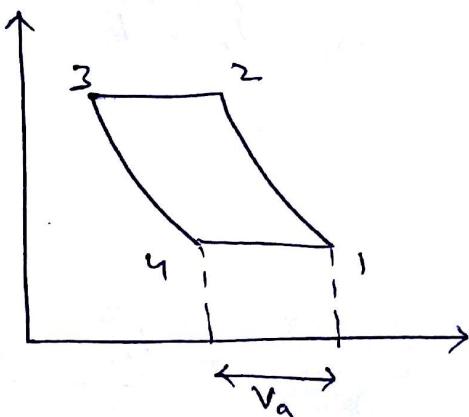
- Use of multistage compressor

1) overall Pressure ratio = $(\text{Pressure ratio})^N$
(in each stage)

2) Total work input = $N \times (\text{work input in each stage})$

$N = \text{no. of stages}$

- Free Air delivery (FAD) (V_F)



$$\dot{m}_{\text{entering}} = \dot{m}_{\text{leaving}} = \dot{m}_F$$

$$\frac{P_1 (V_i - V_u)}{R T_1} = \frac{P_2 (V_2 - V_3)}{R T_2} = \frac{P_F V_F}{R T_F}$$

$$P_F = 1 \text{ bar}$$

$$T_F = 288 \text{ K} = 15^\circ\text{C}$$

\Rightarrow calculation of cylinder Dimensions

$$\frac{P_1 V_{a1}}{T_1} = \frac{P_2 V_{a2}}{T_2} = \frac{P_3 V_{a3}}{T_3}$$

$$\left[\eta_v = \frac{V_a}{V_s} \right]$$

$$\eta_{v1} P_1 V_{s1} = \eta_{v2} P_2 V_{s2} = \eta_{v3} P_3 V_{s3}$$

when $T_1 = T_2 = T_3$
Perfect intercooling

$$\Rightarrow \eta_{v1} P_1 D_1^2 = \eta_{v2} P_2 D_2^2 = \eta_{v3} P_3 D_3^2$$

when $l_1 = l_2 = l_3$

CHAPTER-4

[CENTRIFUGAL COMPRESSOR]

- General concepts for all type of compressors

$$\text{Enter's Power} = m(V_{w_2}u_2 - V_{w_1}u_1) \quad \text{J/kg} \times \frac{\text{kg}}{\text{s}} = \text{watt}$$

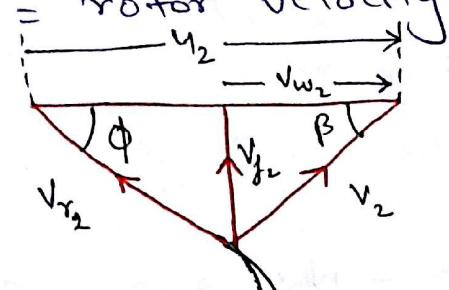
$$W_c = V_{w_2}u_2 - V_{w_1}u_1$$

$$W_T = V_{w_1}u_1 - V_{w_2}u_2$$

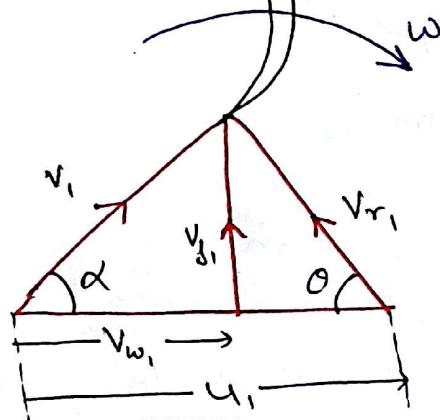
V_{w_1} = whirl velocity at inlet ($v_1 \cos \alpha$)

V_{w_2} = whirl velocity at outlet ($v_2 \cos \beta$)

u_1 & u_2 = rotor velocity at inlet & outlet



α & ϕ = blade angle at inlet & outlet

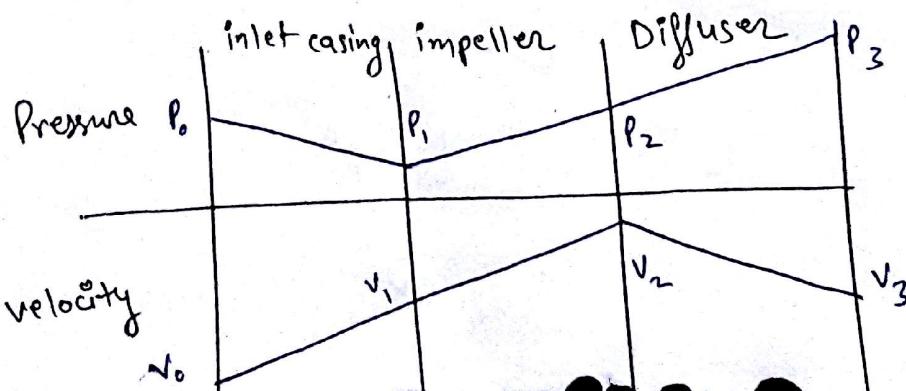


$$V^2 = V_w^2 + V_\theta^2$$

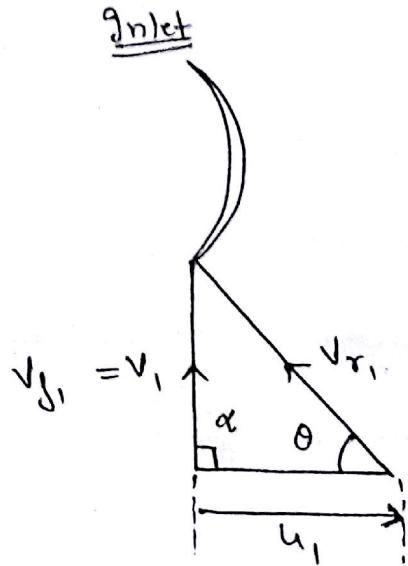
$$\vec{U} + \vec{V}_\theta = \vec{V}$$

$$V_{w_2}u_2 - V_{w_1}u_1 = \underbrace{\frac{v_2^2 - v_1^2}{2}}_{\text{impulse effect}} + \underbrace{\frac{u_2^2 - u_1^2}{2}}_{\text{centrifugal effect}} + \underbrace{\frac{v_{r_1}^2 - v_{r_2}^2}{2}}_{\text{reaction effect}}$$

⇒ Centrifugal compressor



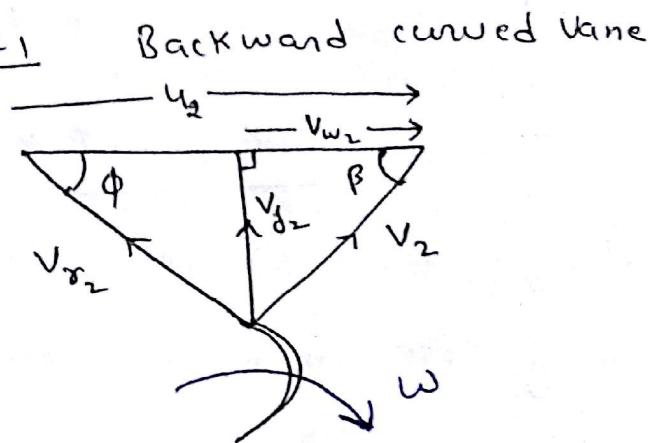
Velocity triangles for centrifugal compressor



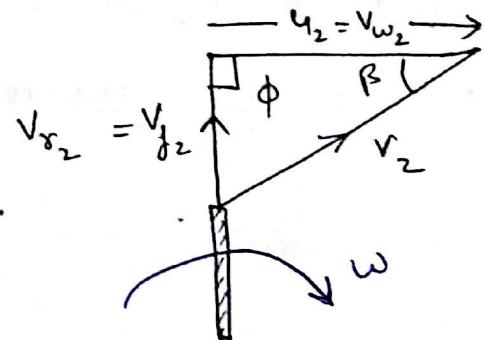
maximum efficiency

\leftarrow case-1

outlet

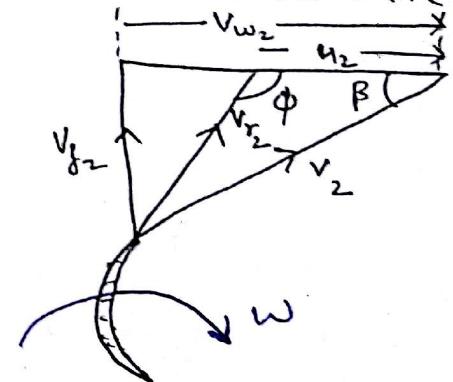


minimum \leftarrow Case-2 radially outward vane stressed



higher Pressure \leftarrow Case-3 ratio

Forward curved vane



$$W_{in} = V_{w2} u_2 - V_{w1} u_1 \text{ J/kg}$$

$V_{w1} = 0$ for centrifugal compressor

$$W_{in} = V_{w2} u_2 \text{ J/kg}$$

$$W_{in} = m V_{w2} u_2 \text{ J}$$

$$\text{slip factor } (\tau) = \frac{V_{w2}}{u_2}$$

$$\Rightarrow V_{w2} = \tau u_2$$

$$W_{in} = \tau \cdot u_2^2 \text{ J/kg}$$

- Power factor (Ψ) = $\frac{\text{Actual } w_{in}}{\text{theoretical } w_{in}} > 1$

$$(w_{in})_{\text{actual}} = \Psi v_{w_2} u_2$$

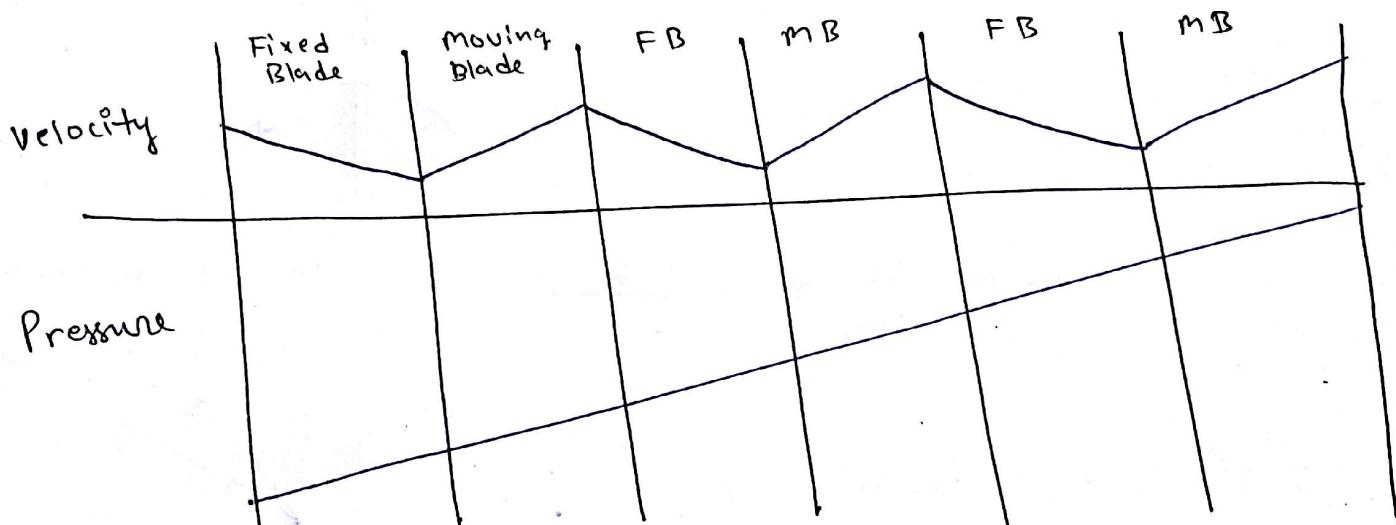
$$w_{in} = \Psi \tau u_2^2 \text{ J/kg}$$

- if Ψ & τ are not given, then assume it to be 1.

- if Prewhirl v_{in} in centrifugal compressor is given
then, $w_{in} = v_{w_2} u_2 - v_{w_1} u_1$

$$\Rightarrow \text{W.D.} = h_2 - h_1 = c_p(T_2 - T_1) = v_{w_2} u_2 = \tau u_2^2 = \Psi \tau u_2^2$$

CHAPTER-5 [AXIAL FLOW COMPRESSOR]



- Area of flow = $\frac{\pi}{4} (D_t^2 - D_h^2)$

D_t = tip / outer diameter

D_h = hub diameter

- $u_1 = u_2 = \frac{\pi D_m N}{60}$; D_m = mean dia = $\frac{D_t + D_h}{2}$

- $V_{g1} = V_{g2}$

- $\alpha_1 > \alpha_2 \Rightarrow$ absolute velocity angle made with V_g

- $\beta_1 > \beta_2 \Rightarrow$ blade angles made with V_g

- Degree of Reaction (R_D)

$$R_D = \frac{\Delta h_{mb}}{\Delta h_{stage}} = \frac{V_{r_1}^2 - V_{r_2}^2}{2 u (V_{w_2} - V_{w_1})} = \frac{V_f}{2u} (\tan \beta_1 + \tan \beta_2)$$

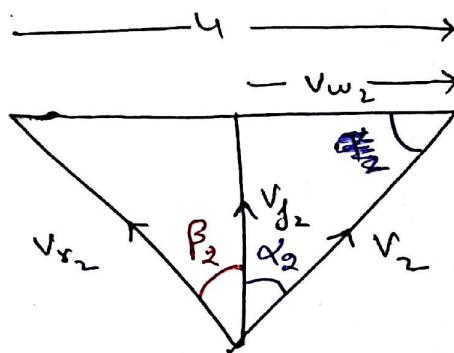
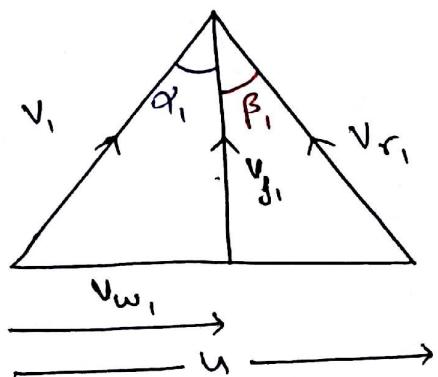
- Work input

$$W_{in} = (V_{w_2} - V_{w_1}) u = V_f \cdot u \cdot (\tan \beta_1 - \tan \beta_2)$$

$$W_{Total} = N \times (W_{in})_{\text{single stage}} \quad [\because N = \text{no. of stages}]$$

- W.D. = $c_p(T_2 - T_1) = N \times V_f \cdot u \cdot (\tan \beta_1 - \tan \beta_2)$

\Rightarrow Velocity triangles



\Rightarrow For 50% R_D

$$\alpha_1 = \beta_2$$

$$\beta_1 = \alpha_2$$

$$V_1 = V_{r_2}$$

$$V_2 = V_{r_1}$$

- Fluid deflection angle = $\beta_1 - \beta_2$