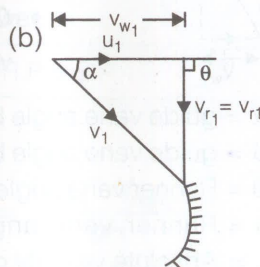
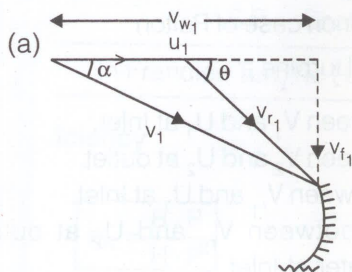


## VELOCITY TRIANGLE

### 1. At Inlet

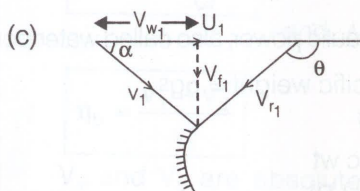


- $\theta < 90^\circ$
- Forward vanes
- Common case of
- Francis Turbine

- $\theta = 90^\circ$
- Radial vanes
- $U_1 = V_{w1}$
- $V_{F1} = V_{r1}$

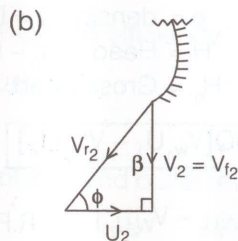
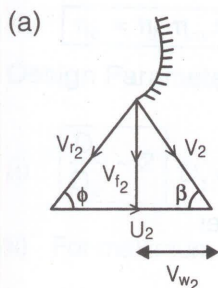
$$\tan \theta = \frac{V_{f1}}{V_{w1} - U_1}$$

$$\tan \theta = \frac{V_{f1}}{V_{w1}}$$

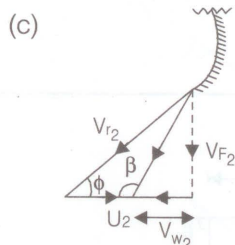


- $\theta > 90$
- Backward vanes
- $\tan (180^\circ - \theta) = \frac{V_{f1}}{U_1 - V_{w1}}$

### 2. At Outlet



- $\beta < 90^\circ$
- $V_{w2}$  is in the direction of  $U_2$ .
- $\beta = 90^\circ$
- Common case of Francis turbine
- Use fuel for maximum efficiency.



- $\beta > 90^\circ$
- Common case of Pelton wheel turbine

Here,  $\alpha$  = guide vane angle between  $V_1$  and  $U_1$  at Inlet.  
 $\beta$  = guide vane angle between  $V_2$  and  $U_2$  at outlet.  
 $\theta$  = Runner vane angle between  $V_{r1}$  and  $U_1$  at Inlet.  
 $\phi$  = Runner vane angle between  $V_{r2}$  and  $U_2$  at outlet.  
 $V_1$  = Absolute velocity of water at Inlet.  
 $V_2$  = Absolute velocity of water of outlet.

$u_1$  and  $u_2$  are relative velocities at inlet and outlet respectively.

$V_1$  and  $V_2$  are tangential also called circumferential velocity at inlet and outlet respectively.

## FRANCIS TURBINE

### 1. Powers

(i)  $H.P = wQ$

= Hydraulic power, also called water power

$w$  = Specific weight =  $\rho g$

denoted by ' $\gamma$ '

$\gamma$  = unit wt or specific wt

1 H.P = 746 watt (British H.P)

1 H.P = 736 watt (Metric H.P)

$\rho$  = density,  $Q$  = Discharge,

$H$  = Head =  $H_g - H_f$

$H_g$  = Gross Head,  $H_f$  = Friction loss.

(ii)  $R.P = \rho Q [V_{w1} U_1 - V_{w2} U_2]$

=  $\frac{wQ}{g} [V_{w1} U_1 - V_{w2} U_2]$  R.P = Runner power

For maximum efficiency  $V_{w2} = 0$ .

(a) For maximum efficiency

$$R.P = \frac{wQ}{g} (V_{w1} U_1)$$

$V_{w1}$  = Whirl velocity at Inlet.

$V_{w2}$  = Whirl velocity at outlet.

(iii)  $S.P = \tau \omega$  S.P = R.P - Mechanical friction losses

Where,

$\tau$  = Torque produced by shaft

$\omega$  = Angular velocity of shaft

S.P = Shaft power.

In Francis Turbine H.P > R.P > S.P

### 2. Efficiency

(i)  $\eta_h = \frac{R.P}{H.P}$   $\Rightarrow \eta_h = \frac{(V_{w1} U_1 - V_{w2} U_2)}{gH}$   
 $\eta_h$  = Hydraulic efficiency

(ii)  $\eta_m = \frac{S.P}{R.P}$   $= 1 - \frac{\text{loss}}{R.P}$   
 $\eta_m$  = Mechanical efficiency.

(iii)  $\eta_o = \frac{S.P}{H.P}$   $\eta_o = \eta_m \cdot \eta_h$   
 $\eta_o$  = Overall efficiency.

(iv)  $\eta_{vol} = \frac{Q - \Delta Q}{Q}$   $\eta_{vol}$  = Volumetric efficiency  
 $Q$  = Discharge ( $m^3/s$ )  
and  $\Delta Q$  = Charge in discharge ( $m^3/s$ ).

(v)  $\eta_b = \frac{v_1^2 - v_2^2}{v_1^2}$   $\eta_b$  = Blade efficiency.

$V_1$  and  $V_2$  are absolute velocity of water at inlet and outlet respectively.

(vi)  $\eta_o = \eta_h \cdot \eta_m \cdot \eta_v \cdot \eta_b$   $\eta_o$  = Overall efficiency.

### 3. Design Parameter of Francis Turbine

(i)  $\frac{D_1}{D_2} \sim 2$   $D_1$  and  $D_2$  are dia of inlet and outlet respectively.

(ii) For maximum efficiency,  $V_{w2} = 0$ .

(iii)  $\phi = \frac{U_1}{\sqrt{2gH}} \approx 0.6 + 0.9$   $\phi$  = Speed ratio  
 $U_1$  = Tangential also called circumferential velocity of inlet.  
 $H$  = Head =  $H_g - H_f$   
 $\sqrt{2gH}$  = Spout velocity.

(iv)  $\psi = \frac{V_{f1}}{\sqrt{2gH}} \approx 0.15 \text{ to } 0.30$   $\psi$  = Flow ratio  
 $V_{f1}$  = Flow velocity of inlet.

(v)  $\eta = \frac{B_1}{D_1} \approx 0.1 \text{ to } 0.45$   $\eta$  = width ratio

(vi)  $A_{f1} = (1 - k) \pi D_1 B_1$   $A_{f2} = (1 - k) \pi D_2 B_2$

$A_{f1}$  and  $A_{f2}$  are area of flow at inlet and outlet respectively.

$k$  = Vane thickness coefficient  $\approx 5\%$

$D_1$  and  $D_2$  are diameter at inlet and outlet respectively.

$B_1$  and  $B_2$  are width of plate at inlet and outlet respectively.

(vii)  $R = \frac{\frac{P_1}{\gamma} - \frac{P_2}{\gamma}}{\left( \frac{V_{w1} U_1 - V_{w2} U_2}{g} \right)}$   $R = \frac{\frac{P_1}{\gamma} - \frac{P_2}{\gamma}}{\left( \frac{V_{w1} U_1}{g} \right)} \rightarrow$  For maximum efficiency

Here,  $R$  = Degree of reaction  
 $\gamma$  = Unit weight or specific wt =  $\rho g$   
 $\rho$  = Density

Also,  $R = 1 - \left( \frac{V_1^2 - V_2^2}{2V_{w1} \cdot U_1} \right)$

(viii)  $H = \frac{V_2^2}{2g} + \frac{V_{w1} U_1}{g}$  (Master Formula) where,  $H$  = Head

Assumption,  $V_{w2} = 0$ ,  $\frac{P_2}{\rho g} = 0$ ,

#### 4. Model Relationship for Turbine

Dimensional Parameter

Dimensionless Parameter

1.  $N_s = \frac{N\sqrt{P}}{(H)^{5/4}}$

1.  $N_s = \frac{\omega \sqrt{\frac{P}{\rho}}}{(gH)^{5/4}}$

2.  $C_H = \frac{H}{N^2 D^2}$

2.  $C_H = \frac{gH}{\omega^2 D^2}$

3.  $C_Q = \frac{Q}{N D^3}$

3.  $C_Q = \frac{Q}{\omega D^3}$

4.  $C_P = \frac{P}{N^3 D^5}$

4.  $C_P = \frac{P}{\rho \omega^3 D^5}$

Here,  $N_s$  = Specific speed

$C_H$  = Head coefficient

$C_Q$  = Discharge coefficient

$C_P$  = Power coefficient

$g \rightarrow$  Accn. due to gravity =  $9.81 \text{ m/s}^2$ .

$\omega \rightarrow$  Angular speed =  $\frac{2\pi N}{60}$

$N \rightarrow$  No. of revolution/minute

$H \rightarrow$  Head (m)

$P \rightarrow$  Pressure ( $\text{N/m}^2$ )

$\rho \rightarrow$  Density ( $\text{kg/m}^3$ )

$D \rightarrow$  Diameter (m)

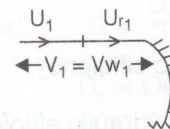
$Q \rightarrow$  Discharge ( $\text{m}^3/\text{sec}$ )

### PELTON WHEEL TURBINE

#### 1. Velocity Triangle

(i) At Inlet

(ii) At outlet

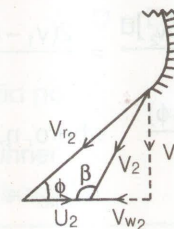


$V_1 = V_{w1}$

$V_{f1} = 0$

Here,  $\alpha = 0^\circ$

$\theta = 0^\circ$



$\beta > 90^\circ$

$V_{r2} = k V_{r1}$

$k$  = Friction factor

$k = 1$

For frictionless vane

$|V_{r2} \cos \phi|$   
 $= |U_2| + |V_{w2}|$

#### 2. Power

(i)

H.P =  $WQH$  ... kW

$= \frac{\rho QH}{75}$  ... H.P

H.P = Hydraulic power.

$$(ii) \text{ Jet power or k.E/sec of Jet} = \frac{1}{2} \rho Q v_1^2 = \frac{1}{2} \cdot \frac{wQ}{g} \cdot v_1^2$$

$$(iii) R.P = \frac{wQ}{g} [V_{w_1} + V_{w_2}] u$$

R.P = Runner power

$$(iv) S.P = R.P - \text{Mechanical friction losses.}$$



H.P > K.E/sec. of Jet > R.P > S.P

### 3. Efficiencies

$$(i) \eta_{\text{nozzle}} = \frac{k \cdot E / \text{sec of jet}}{H \cdot P \text{ available of base of nozzle}}$$

$$= \frac{V_1^2}{2gH} = C_v^2 \quad \left[ \because v_1 = C_v \sqrt{2gH} \right]$$

where,  $\eta_{\text{nozzle}}$  = nozzle efficiency.

$$(ii) \eta_h = \frac{R \cdot P}{k \cdot E / \text{sec of Jet}}$$

$$\eta_h = \frac{2[V_{w_1} + V_{w_2}]u}{v_1^2} = \frac{2(v_1 - u)(1 + k \cos \phi)u}{v_1^2}$$

$$\eta_{h|_{\text{max}}} = \frac{(1 + k \cos \phi)}{2} \quad \text{where, } \eta_h = \text{Hydraulic efficiency.}$$

$\eta_{h|_{\text{max}}}$  = Maximum hydraulic efficiency.

$$(iii) \eta_m = \frac{S \cdot P}{H \cdot P} \quad \text{where, } \eta_m = \text{Mechanical efficiency.}$$

$$(iv) \eta_o = \frac{S \cdot P}{H \cdot P} = \eta_{\text{nozzle}} \cdot \eta_h \cdot \eta_{\text{mechanical}} \quad \text{where, } \eta_o = \text{overall efficiency}$$

### 4. Design Criteria

$$(i) \phi = \frac{u_1}{\sqrt{2gH}} \quad 0.45 \text{ to } 0.47 \quad \text{where, } \phi = \text{Speed ratio.}$$

$$(ii) \psi = \frac{V_{f_1}}{\sqrt{2gH}} = \text{zero} \quad \text{where, } \psi = \text{Flow ratio}$$

$$(iii) m = \frac{D}{d} = 11 \text{ to } 15$$

where,  $m$  = Jet ratio

$D$  = Dia of pitch circle

$d$  = Dia of jet.

$$(iv) n = \left( 15 + \frac{m}{2} \right) \quad \text{this is Tygon formula.}$$

= 18 to 25

where,  $\eta$  = Number of vanes

$$(v) \text{ No. of jet} = \frac{\text{Total discharge through penstock}}{\text{Discharge through each jet}} \approx 6$$

## KEPLAN AND PROPELLER TURBINE (AXIAL FLOW REACTION TURBINE)

$$1. Q = A_{F_1} \cdot V_{F_1} = A_{F_2} \cdot V_{F_2} \quad \text{where, } Q = \text{Discharge}$$

$$A_{F_1} = \frac{\pi}{4} (D_o^2 - D_b^2)$$

$$2. U_1 = U_2 = \frac{\pi DN}{60}$$

where,  $D = D_o$  at expressed edge

$D = D_b$  at inner edge

$$\text{and } D = \frac{D_o + D_b}{2} \quad \text{at mid point}$$

$D_o$  = Outer dia of runner

$D_b$  = Dia of hub or boss.



The analysis of velocity triangle, powers. Are Similar to that of francies turbines. In this case generally  $\theta > 90^\circ$  and  $V_{w_2} = 0$ .

Master formular (Beronoulies principle) can be applied.

### 3. Design Parameter

$$(i) \phi = \frac{u_1}{\sqrt{2gH}} \quad \text{is of the order of } 2.0$$

$$(ii) \psi = \frac{V_{f_1}}{\sqrt{2gH}} = 0.5 \text{ to } 0.7$$

(iii) Number of vanes on the runner are generally 3 to 8.

**Impulse turbine (Pelton):** High head and low discharge.

**Francis turbine:** Medium head and medium discharge.

**Kaplan and Propeller turbine:** Low head and high discharge.

Turbine	Specific speed, $N_s$ , (MKS)
1. Pelton wheel turbine (single jet)	10 - 35
2. Pelton wheel turbine (multiple jet)	35 - 60
3. Francis turbine	60 - 300
4. Kaplan turbine	> 300

