27. Specific Heat Capacities of Gases

Short Answer

Answer.1

No, the number of specific heat capacities of gas is infinite as it depends on the thermodynamic process followed by the gas.

The specific heat capacity of a substance is defined as the amount of heat required to raise the temperature of one mole of that substance by 1 degree Celsius, or 1 Kelvin. It is denoted by C.

Gases are compressible substances. They have two well-known specific heat capacities: one at constant pressure(C_p)(isobaric process - constant pressure) and another at constant volume(C_v)(isochoric process - constant volume). However, gases can have many specific heat capacities depending on the other thermodynamic processes they follow, like adiabatic process, isothermal process, etc.

Answer.2

The specific heat capacity of a substance is defined as the amount of heat required to raise the temperature of one mole of that substance by 1 degree Celsius, or 1 Kelvin. It is denoted by C.

Hence,
$$C = \frac{Q}{m \times dT}$$
, ... (i)

where

C = specific heat capacity

Q = heat required to raise the temperature by dT

m = molar mass

dT = change in temperature.

For constant temperature, dT = 0. Putting this value in (i), we get

$$C = \frac{Q}{m \times dT} = \infty$$

Hence, for a process at a constant temperature, the specific heat capacity is infinite.

Answer.3

The specific heat capacity of a substance is defined as the amount of heat required to raise the temperature of one mole of that substance by 1 degree Celsius, or 1 Kelvin. It is denoted by C.

Hence, $C = \frac{Q}{m X dT}$, ... (i) where

C = specific heat capacity

Q = heat required to raise the temperature by dT

m = molar mass

dT = change in temperature.

For an adiabatic process, Q = 0. Substituting this value in (i), we get

C = 0.

Hence, for a process at a constant temperature, the specific heat capacity is zero.

Answer.4

Yes, a solid also has two kinds of molar heat capacities, $C_p(at \text{ constant pressure})$ and $C_v(at \text{ constant volume})$.

Solids are almost incompressible.

Hence, the values of C_p and C_v are such that $C_p > C_v$, but they are almost equal since the dependence on heat capacities is very less in the case of solids.

Since the values of C_p and C_v are not that different, C_p - C_v is much less than R.

We know that, for an ideal gas,

$$C_{P} - C_{V} = R \dots (i)$$

where

 C_p = specific heat constant at constant pressure

 C_v = specific heat constant at constant volume

R = universal gas constant

Multiplying by n x dT on both sides of (i), we get

 $nC_P dT - nC_v dT = nRdT$

which gives

 $(dQ)_{P} - (dQ)_{v} = nRdT ...(ii)$

Since $(dQ)_p = nC_p dT$ and $(dQ)_v = nC_v dT$... (iii)

Where

n = number of moles

dT = change in temperature

 $(dQ)_p$ = change in heat at constant pressure

 $(dQ)_v$ = change in heat at constant volume

However, for a real gas, the internal energy depends on the temperature as well as the volume.

Hence, there will be an additional term on the right-hand side of (ii) which will indicate the change in the internal energy of the gas with volume at constant pressure. Let this term be u.

Hence, for a real gas, (ii) becomes :

 $(dQ)_p - (dQ)_v = nRdT + u ... (iv)$

Again, dividing on both sides by ndT, we get

 $C_p - C_v = R + \frac{u}{ndT} \dots (v),$

which is greater than R.

Here,

 C_p = specific heat constant at constant pressure

 C_v = specific heat constant at constant volume

R = universal gas constant

n = number of moles

dT = change in temperature

Hence, from (v), we get $C_p - C_v > R$.

We conclude that for a real gas, $C_p - C_v > R$.

Answer.6

According to the first law of thermodynamics,

 $dQ = dU + dW = nC_v dT + dW ... (i),$

where

dQ = heat supplied

dU = change in internal energy

dW = work done on the gas

n= number of moles

 C_v = specific heat capacity at constant volume

dT = change in temperature

For an adiabatic process, dQ(heat supplied) = 0.

An adiabatic process occurs without the transfer of heat or mass of substances between the thermodynamic system and the surrounding.

For an isothermal process, dT(change in temperature) = 0

An Isothermal process is a change of system, in which the temperature remains constant $\Delta T=0$.

Putting these values in (i), we get

dW = 0,

which is not possible for either of the processes.

dW = 0 only in the case of a process where the volume is constant that is dV = 0, since dW = PdV,

where P = pressure and dV = change in volume.

Hence, we conclude that a process cannot be both adiabatic and isothermal.

Answer.7

For an isothermal process, the ideal gas equation is given as

PV = constant ... (i),

Where

P = pressure

V = volume.

Differentiating on both sides of (i), we get

PdV + VdP = 0

On solving for $\frac{dp}{dv}$, we get

$$\frac{dP}{dV} = -\frac{P}{V} \dots \text{(ii)}$$

For a graph of P versus V, dP/dV indicates the slope.

Hence, for an isothermal process, the slope of the p-V diagram is given by -P/V.

Now for an adiabatic process, the ideal gas equation is

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PV^{\gamma}= constant ... (iii),
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where

P = pressure,

V = volume,

 γ = ratio of specific heat capacities at constant pressure and constant volume.

Differentiating both sides of (ii), we get

 $V^{\gamma}dP + \gamma V^{\gamma-1}PdV = 0$ which gives

 $\frac{dP}{dV} = -\frac{\gamma P}{V}... \text{ (iv)}$

Hence, for an adiabatic process, the slope of the p-V diagram is given by $-\gamma P/V$.

Since $\gamma > 1$, we find that $\gamma P/V$ is greater than P/V, which concludes that slope of p-V diagram of an adiabatic process is steeper than that of an isothermal process(proved).

Answer.8

An isothermal process is represented by the equation

PV = constant ... (i),

where

P = pressure

V = volume.

To keep this product constant, a small change in V will only produce a small change in P and vice versa. Hence, an isothermal process is usually a slow process.

On the other hand, an adiabatic process is represented as

 PV^{γ} = constant ... (ii), where

P = pressure

V = volume

The γ = ratio of specific heat capacities at constant pressure to constant volume.

Now, $\gamma > 1$. Hence, the term V^{γ} will increase exponentially. Hence, to keep the product constant, a small change in V will cause a large change in P. Hence, an adiabatic process is usually a fast process.

Answer.9

An isothermal process is represented by

PV = constant ... (i),

where P = pressure, V = volume.

This gives

 $P_1 V_1 = P_2 V_2$ (ii),

Where

 P_1 , V_1 = initial values of pressure and volume

 P_2 , V_2 = final values of pressure and volume.

An adiabatic process is represented by

 $PV^{\gamma} = constant...$ (iii),

where

P = pressure,

V = volume,

 γ = ratio of specific heat capacities at constant pressure and constant volume.

From (iii), we get

 $P_1 V_1^{\gamma} = P_2 V_2^{\gamma} \dots$ (iv),

Where P_1 , V_1 = initial values of pressure and volume

 P_2 , V_2 = final values of pressure and volume.

Dividing (iv) by (ii), we get

 $V_1^{\gamma-1} = V_2^{\gamma-1} \dots (v)$

If the condition given by (v) is satisfied, then two states of an ideal gas can be connected by both an isothermal and an adiabatic process.

Answer.10

Given:

$$\gamma = \frac{c_P}{c_v} = 1.29$$

Formula used:

 $\gamma = 1 + \frac{2}{f} \dots$ (i),

where

 γ = ratio of molar heat capacities at constant pressure to constant volume

f = number of degrees of freedom

(i) becomes :

$$\gamma - 1 = \frac{2}{f}$$

$$\Rightarrow f = \frac{2}{\gamma - 1}$$
... (ii)

Substituting γ = 1.29 in (ii), we get

 $f = \frac{2}{(1.29-1)} = 6.89$ which is approximately equal to 7.

Thus, the number of degrees of freedom is approximately equal to 7.

Objective I

Answer.1

- Q = nCdT ... (i), where Q = work done by an ideal gas n = number of moles C = molar heat capacity
- dT = rise in temperature.

For process A, let the value of C be C_A and for B, let it be C_B .

Since the work done by the gas in process A is twice that in B, and the rise in temperature is the same in both the cases, we get two equations:

 $2Q = nC_A dT \dots$ (ii) and

 $Q = nC_B dT \dots$ (iii),

Where

Q = work done in process B

n = number of moles of gas

dT = rise in temperature

Dividing (ii) by (iii), we get

$$2 = \frac{C_A}{C_B}$$

$$=> C_{A} = 2C_{B}$$

This proves that $C_A > C_B$.

Answer.2

For solids which have a small expansion coefficient, the work done on the solid is pretty small. Hence, the specific heat at constant pressure and at a constant volume only slightly different since the work done depends very little on the process. Hence, C_p is slightly greater than C_v but much less than R as in the case of ideal gases.

For gas in state A, $C_p - C_v = R$. This represents an ideal gas. Now, for an ideal gas, we require very high temperature and very low pressure compared to a real gas

For gas in state B, $C_p - C_v = 1.08R$, which represents a real gas. Since gas A was ideal, its pressure must be much lower than that of B and temperature must be much higher than that of A.

Hence, we require the condition $p_A < p_B$ and $T_A > T_B$. This is given by option (a).

Options (b), (c) and (d) are incorrect because none of those satisfies the conditions for an ideal gas.

Answer.4

For an ideal gas, $C_p - C_v = R$.

Here, R is the universal gas constant.

Hence, the correct option is (c).

Option (a) is incorrect since $C_p/C_v = \gamma$, which differs for monoatomic, diatomic or polyatomic gases. Here $\gamma =$ ratio of molar heat capacities at constant pressure and constant volume.

Option (b) is incorrect since $C_p C_v = \gamma C_v^2$, which is not a constant as γ varies.

Option (d) is incorrect since $C_p + C_v = (\gamma + 1)C_v$, which is not a constant as γ varies.

We know that, $Q = nC_p dT...$ (i),

Where

Q = heat required to raise the temperature

n = number of moles

 C_p = specific heat capacity at constant pressure

dT = rise in temperature.

In this first case,

Amount of heat required(Q) = 70 cal

Number of moles(n) = 2 mol

Rise in temperature(dT) = $(35-30)^{0}$ C = 5^{0} C

Hence, from (i), we get $C_p = 7 \text{ calmol}^{-1} {}^{0}C^{-1}$

Now, we know, $C_p - C_v = R$,

Where

 C_p = specific heat capacity at constant pressure

C_v = specific heat capacity at constant volume

R = universal gas constant = $1.98 \text{ calmol}^{-1} \text{ }^{0}\text{C}^{-1}$

Therefore, we get $C_v = C_p - R = (7-1.98)$ cal mol⁻¹ $^0C^{-1}$

 $= 5.02 \text{ cal mol}^{-1} \, {}^{0}\text{C}^{-1}$

Now, in the new case, change in temperature $dT = 5^{0}C$ as before. Number of moles(n) = 2, and C_v = 5.02 calmol⁻¹ $^{0}C^{-1}$

Hence, amount of heat required to raise the temperature of the same gas through the same range

 $dQ = nCvdT = (2 \times 5.02 \times 5) cal = 50.2 cal which is approximately equal to 50 cal.$

Hence, the correct option is (b).

Answer.6

Let us consider the figure given below :



We consider the process AB to be composed of two processes, AC(at constant pressure) and CB(at constant volume), such that AB = AC + CB. C is the molar heat capacity of AB, C_p is the molar heat capacity of AC(constant pressure) and Cv is the molar heat capacity of CB(constant volume).

From the first law of thermodynamics, we know that $Q = U + W \dots (i)$,

Where Q = heat supplied, U = change in internal energy, W = work done on the system.

Since the change in internal energy is independent of the path, it will have the same value for paths AB and ACB.

Hence, $U_{AB} = U_{ACB}$... (ii), where U_{AB} = change in internal energy for path AB, U_{ACB} = change in internal energy for path ACB.

Now, the work done by a process is given by the area under the PV diagram. We can clearly see that the area under process AB is greater than the sum of areas under processes AC and CB.

Hence, $W_{AB} > W_{ACB}$... (iii), where W_{AB} = work done for process following path AB, W_{ACB} = work done for process following path ACB.

Adding (ii) and (iii), we get :

 $U_{AB} + W_{AB} > U_{ACB} + W_{ACB} \dots$ (iv)

But, from the first law of thermodynamics, Q = U + W, where Q = heat supplied, U = change in internal energy, W = work done on the system.

Hence, we get, $Q_{AB} > Q_{ACB} \dots (v)$,

We know, $Q_{AB} = nCdT$, $Q_{AC} = nC_pdT$, $Q_{CB} = nC_vdT$,

where Q_{AB} = heat supplied in process AB

 Q_{AC} = heat supplied in process AC

 Q_{CB} = heat supplied in process CB

n = number of moles

C = molar heat capacity

C_p = molar heat capacity at constant pressure

 C_v = molar heat capacity at constant volume

dT = change in temperature.

Also, $Q_{ACB} = Q_{AC} + Q_{CB} \dots$ (vi), where

Q_{ACB} = heat supplied in process ACB

 Q_{AC} = heat supplied in process AC

 Q_{CB} = heat supplied in process CB

Hence, from (v), we get,

 $nCdT > nC_pdT + nC_vdT$

Dividing by ndT on both sides :

 $C > C_p + C_v \Longrightarrow C > C_v$ (option c)

Given : $p = \frac{\kappa}{v^g}$ which gives us

 $pV^g = K$, where p = pressure, V = volume, and g and K are constants. This represents an adiabatic process.

In an adiabatic process, Q(heat exchanged) = 0.

Now we know that Q = nCdT,

where n = number of moles, C = specific molar heat capacity and dT = rise in temperature.

Since Q = 0, we get C = 0.

Hence, the correct option is (d).

Answer.8

Let the initial values of pressure and volume be P_1 and V_1 , and let the final values of pressure and volume be P_2 and V_2 .

Now for an isothermal process, PV = constant .. (i)

This gives us: $P_1V_1 = P_2V_2$.. (ii)

Since the pressure increases by 0.5%, the new pressure is

$$P_2 = P_1(1 + \frac{0.5}{100}) = 1.005P_1$$

Substituting this value in (ii) :

$$P_1V_1 = 1.005P_1V_2$$

=> $V_2 = \frac{V_1}{1.005} = 0.995 V_1.$

Hence, the volume decreases by (1-0.995) = 0.005 = 0.5%

Hence, the correct option is (b).

Answer.9

For an adiabatic process, PV^{γ} = constant ... (i)

If P_1 , V_1 represents the initial values of pressure and volume and P_2 , V_2 represent the final values of pressure and volume,

$$P_1 V_1^{\gamma} = P_2 V_2^{\gamma} \dots$$
 (ii)

Now, $\gamma = 1.4$ (given).

It is given that the pressure increases by 0.5%.

Hence,
$$P_2 = \left(1 + \frac{0.5}{100}\right)P_1 = 1.005P_1$$

Substituting in (ii) :

 $P_1 V_1^{\gamma} = 1.005 \text{ X } P_1 V_2^{\gamma}$

 $=> (V_1/V_2)^{\gamma} = 1.005 \dots (iii)$

Taking log on both sides of (iii), we get

$$\gamma log\left(\frac{V_1}{V_2}\right) = log(1.005) = 0.002$$

$$=> log\left(\frac{V_1}{V_2}\right) = \frac{0.002}{1.4} = 0.0014 \dots (iv)$$

Taking inverse log on both sides of (iv), we get

 $V_1/V_2 = 10^{0.0014} = 1.003$ => $V_2 = V_1/1.003 = 0.997V_1$

Hence, the volume decreases by (1-0.997) = 0.003 which is approximately equal to 0.36%.

Hence, the correct option is (a).

Answer.10

Adiabatic process is represented by : PV^{γ} = constant,

where P = pressure

V = volume

 γ = ratio of specific heat capacities at constant pressure and constant volume

Therefore, we get $\mathsf{P}_A = \text{constant}/\mathsf{V}^{\mathbb{Y}}.$.. (i)

Isothermal process is given by : PV = constant

Hence, we get $P_B = constant/V \dots (ii)$

Since $\gamma > 1$, we get $p_A < p_B$. (proved)

Let $\rm T_a$ and $\rm T_b$ be the final temperatures of the samples A and B respectively in the previous question.

A. $T_a < T_b$

B. $T_a = T_b$

C. $T_a > T_b$

D. The relation between T_{a} and T_{b} cannot be deduced.

Answer.11

Since sample B undergoes isothermal expansion, its temperature remains constant = T_b .

For an adiabatic process, since the heat supplied is 0, the internal energy will change by an amount $dU = nC_v dT$,

where dU = change in internal energy

n = number of moles

 C_v = specific heat capacity at constant volume

dT = change in temperature

This change in internal energy will compensate for the constancy in heat.

Sample B is undergoing expansion through an isothermal process; its initial and final temperatures will be the same.

Sample A will expand at the cost of its internal energy.

Therefore, the final temperature will be less than the initial temperature,

since $dU < 0 \Rightarrow dT < 0$.

 $T_b-T_a < 0$

 $T_b > T_a \text{ or } T_a < T_b$

Hence, we get $T_a > T_b$.

We know that for any given state, the slope of the p-V diagram of an adiabatic process is $-\gamma P/V$, while that of an isothermal process is -P/V.

Hence, the slope of an adiabatic process is more.

Now, the area under the curve of a p-V diagram gives the work done.



From this diagram, we can see that that area under the curve of process A(ACDE) which represents the adiabatic process with greater slope is less than that of the area under the curve of process B(ABDE), which represents the isotherm.

Hence, we can conclude that $\Delta W_a < \Delta W_b$

Answer.13

We know that the molar heat capacity of any gas depends on the degrees of freedom of the gas. On increasing the temperature of the gas, we allow more molecules to vibrate about their equilibrium positions, which in turn increases the degrees of freedom of the gas. For this reason, the molar heat capacity increases as the temperature of the gas are increased.

Hence, the correct answer is an option (d).

Objective II

Since the gas kept in the container is suddenly compressed, there is very little time for any heat to allow to enter or leave the container. Hence, it must be very nearly an adiabatic process. If the temperature remains constant during this process, then the process may also be isothermal. Option (a) is incorrect because it must be very nearly adiabatic but might not be completely depending on how

sudden the process is.

Option (b) is incorrect because if the compression is indeed done suddenly, then the temperature of the gas does not remain constant, and it may not be isothermal. It changes with the pressure and the volume of the gas according to the formulae:

i.
$$\frac{P_1 V_1}{T_1} = \frac{P_2 V_2}{T_2}$$

ii. PV^{γ} = constant

Where,

P1, V1, T1: initial values of pressure, volume and temperature

P2, V2, T2: final values of pressure, volume and temperature

 γ : adiabatic index, varies from gas to gas

Therefore, the correct answers are option (c) and (d).

In an isothermal process, the temperature remains constant. The internal energy of an ideal gas is a state function that depends on temperature. Hence, change in internal energy is zero and from the first law of thermodynamics: $\Delta U = Q - W$, where $\Delta U =$ change in internal energy, Q = amount of heat given and W = work done by it.

Since $\Delta U = 0$ in this case, we get Q = W.

Options (a) and (b) are incorrect because we actually provide a finite amount of heat to the system, and hence work is also not zero.

Option (c) is incorrect because we just showed that Q = W.

Hence, the correct option is option (d).

Answer.3

In an adiabatic process, we assume that there is no heat exchange between the system and surroundings.

Hence, Q = 0. Also, U = W (when Q = 0), so $Q \neq W$ holds.

Option (b) is incorrect because due to change in temperature, there is still a change in internal energy and from the first law of thermodynamics, $\Delta U = W$. Hence $W \neq 0$.

Option (c) is incorrect because the heat exchange is zero, but the work done in general is not zero.

Hence, the correct answers are option (a) and (d).

- A. both the processes are isothermal
- B. both the processes are adiabatic
- C. A is isothermal and B is adiabatic
- D. A is adiabatic and B is isothermal

From the graph, we can see that the slope of process B is steeper than that of process A.

Now in an isothermal process, under constant temperature, Pressure(P) x Volume(V) = constant ... (i) (according to Boyle's law)

Differentiating on both sides:

PdV + VdP = 0.slope = $\frac{dP}{dV} = -\frac{P}{V}$... (ii)

For an adiabatic process,

$$PV^{\gamma} = constant$$

Where,

 γ is the ratio of specific heat of the gas at constant pressure and constant volume.

Differentiating the above question, we get

$$V^{\gamma}dP + \gamma V^{\gamma-1}PdV = 0$$

$$slope = \frac{dP}{dV} = -\frac{\gamma P}{V}$$

which is greater than that in case of the isothermal process.

Thus path (A) is for isothermal process while path (B) is for the adiabatic process.

Options (a) and (b) are incorrect since the slopes of the two paths Differ.

Option (d) is incorrect because the slope of path (A) is less than that of path (B).

Hence, the correct answer is option (c).

Since this is an adiabatic process, Q(heat) = 0.

Hence, dU(change in internal energy) = dW(work done) = $nC_v dT$.

where n = no. of moles, C_v = specific heat of the gas at constant volume, and dT = change in temperature.

 C_v is different for oxygen(diatomic gas) but same for helium and neon(monoatomic gases) so dT for oxygen is different.

W for adiabatic process = $(P_2V_2-P_1V_1)/(1-\gamma)$

In all cases, $V_2 = V_1/2$.

 P_1 , V_1 , T_1 - initial pressure, volume and temperature

 P_2, V_2, T_2 - final pressure, volume and temperature

 V_1 , P_1 are same for all gases and so is W. Only γ for oxygen is different compared to helium and neon since it is a diatomic gas while the rest are monoatomic.

Hence, options (c) and (d) are correct.

Answer.6

Let us use the formula,

Heat supplied to a gas, $dQ = nC_v dT$.

Where,

n is the number of mole of the ags

 C_v is the specific heat of the gas at constant volume T is the temperature

Putting the values in the above formula, we get

$$dQ = 3.0 cal = (3.0 x 4.2)J = 12.6 J.$$

Therefore, C_v is approximately equal to 12.6 J/mol/°C ~ 1.5R = 3R/2

We know,
$$Cp = Cv + R = \frac{3R}{2} + R = \frac{5R}{2}$$

Where C_p = specific heat at constant pressure

R = universal gas constant = 8.314 J/kg/mol

$$\gamma = \frac{Cp}{Cv} = \frac{5}{3}$$
 which holds for monoatomic gases.

Helium and Argon are monoatomic and so options (a) and (b) hold.

Answer.7

According to the law of equipartition of energy, the Kinetic energy associated with each degree of freedom of a molecule is kT/2,

$$KE = \frac{f \kappa T}{2}$$

where k is Boltzmann constant,

T is absolute temperature.

f is the number of degrees of freedom

The number of degrees of freedom of a system is the minimum number of coordinates required to completely describe its position and orientation.

Now argon,

which is a monoatomic gas, can only translate along the x, y and z axes.

Hence, it has 3 degrees of freedom. Now, for a system,

total associated energy is:

Total energy of all the degrees of freedom, $E = \frac{fkT}{2}$

Where

E = total energy,

f = number of degrees of freedom

k = Boltzmann constant = $1.38 \times 10^{-23} \text{m}^2 \text{ kg s}^{-2} \text{ K}^{-1}$

T = absolute temperature (Kelvin)

For argon, f = 3. Therefore, $E = \frac{3kT}{2}$ which is minimum.

Options (b) and (c) are incorrect since they are diatomic and

have more than 3 degrees of freedom(translational and rotational degrees of freedom of 2 molecules).

Option (d) is incorrect since carbon dioxide is triatomic and has

more than 3 degrees of freedom(translational and rotational degrees of freedom of 3 molecules).

Hence, the correct answer is option (a).

Exercises

Answer.1

Given:

number of moles, n = 1

Specific heat at constant temperature, C_v (monoatomic gas) = 3R/2 = 12.471 J/mol/K

R = universal gas constant = 8.314 J/mol/K

Initial velocity(v_i) = 50 m/s

Final velocity $(v_f) = 0$

Molecular weight(m) = 20 g/mol = 0.02 kg/mol

Formula Used:

i. Change in internal energy(dU) = $nC_v dT$,

Where,

n is the number of the moles of the gas,

 $C_{\boldsymbol{v}}$ is the heat capacity at the constant volume

dT = rise in temperature.

ii. Mechanical energy lost,
$$E = -m(\frac{v_i^2 - v_f^2}{2})$$

Where,

m is the molecular weight of the gas in kg

 \boldsymbol{v}_i is the initial velocity,

 $v_{\rm f}$ is the final velocity,

equating equation (i) and (ii), we get

Putting the values in the above equation, we get

$$\frac{0.02X(2500-0)}{2} = 12.471XdT$$

=> dT = rise in temperature \sim 2 K (Answer).

Answer.2

Given:

Mass of the gas(m) = 5g

Change in temperature(dT) = 10° C

Specific heat at constant volume(C_v) = = 0.172 cal g⁻¹°C⁻¹

Mechanical equivalent of heat = 4.2 J cal^{-1}

Formula used:

Change in internal energy = $mC_v dT$

Where,

m = Mass of the gas
C_v = Specific heat at constant volume
dT = Change in temperature
Also, we know,
Heat(Joule)= Mechanical equivalent of heat× Heat(cal)
Putting the values in the above formula, we get
= (5x10x0.172) cal

- = 8.6 cal = (8.6x4.2)J
- = 36.12 J (Answer)

Answer.3

Given:

Mass of piston(m) = 50 kg

Area(A) = 100 cm^2 = $(100 \times 10^{-4})\text{m}^2$ (since 1m = 100 cm)

Acceleration due to gravity, $g = 10 \text{ ms}^{-2}$

Atmospheric pressure = 100 kPa

Distance through which it moves = 20 cm

γ = 1.4

Formula used:

Therefore, pressure exerted by piston = $\frac{Force}{area} = \frac{(m \times g)}{area}$

= ((50x10)/(100x10⁻⁴)) Pa

= 50,000 Pa

Atmospheric pressure = 100 kPa = 1,00,000 Pa.

Therefore, Total pressure(P) = (50,000 + 1,00,000)Pa

=1,50,000Pa

Work done = Pressure x change in volume = P x dV

dV(change in volume) = distance moved by piston x Area

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= (20 \text{ cm x } 100 \text{ cm}^2)
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= 2,000 cm<sup>3</sup> = 2,000 x 10^{-6} m<sup>3</sup> = 2 x 10^{-3} m<sup>3</sup>
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Therefore, Work = $(1,50,000 \times 2 \times 10^{-3})$ J = 300 J

Work done, $W = P\Delta V = n R dT$

We get,

$$dT = \frac{300}{nR}$$

Now, We calculate Q:

 $dQ = nC_p dt = n \times C_p \times \frac{300}{nR} = 300 \times \frac{C_p}{R}$

Given: $\gamma = 1.4 = \frac{c_p}{c_v}$. Also, $C_p + C_v = R$. Solving these two equations, we get $C_p = 7R/2$, $C_v = 5R/2$.

Hence, dQ = $\frac{300X\frac{7R}{2}}{R} = \frac{300X7}{2} = 1050$ J.

Answer.4

Given:

Specific heat at constant volume is $C_v = 2.4$ cal $g^{-10}C^{-1}$

Specific heat at constant pressure is C_p =3.4 cal g^{-1o}C⁻¹

Molecular mass of hydrogen is 2 g mol⁻¹

Gas constant, R= 8.3×10^7 gmol

Formula used:

(mXCp) - (mXCv) = R

Where:

m is the molecular weight of hydrogen

 $C_{\rm p}$ is the specific heat at constant pressure

 C_v is the specific heat at constant volume

R is the gas constant.

Putting the values in the above equation , we get

The gas constant R= (m X C_p) - (m X C_v) = 2 × (Cp-Cv)= 2×J

Now, 2× J=R

 $2 \times J = 8.3 \times 10^7 \text{ erg/ mol}^{-0} \text{C}$

Thus, $J=4.15 \times 10^7$ erg/cal (Answer).

Answer.5

Given:

n = number of moles = 1,

 C_v = specific heat capacity at constant volume,

 C_p = specific heat capacity at constant pressure

dT = change in temperature = 50K.

 γ = Ratio of molar heat capacities = C_P/C_V = 7/6 => C_v = $6C_p/7$.

(a) Formula used:

Pressure constant: Isobaric process. For an isobaric process,

change in internal energy $dU = nC_v dT$,

Where

n = number of moles,

 C_v = specific heat at constant volume,

dT = rise in temperature

Also, C_p - C_v = R.

C_p = specific heat at constant pressure

 C_v = specific heat at constant volume

R = universal gas constant = 8.314 J/mol/K

Substituting: $C_p - 6C_p/7 = C_p/7 = R \implies C_p = 7R$.

Therefore $C_v = C_p - R = 6R = (6 \times 8.314)J/mol/K$

Therefore,

dU = 1 mol X (6 X 8.314)J/mol/K X 50K = 2494.2 J(Ans)

(b) Volume constant: Isochoric process, dV = 0(change in volume)

First law of thermodynamics gives us: dU = dQ - dW

Where dU = change in internal energy, dQ = change in heat,

dW = work done = Pressure x change in volume = PdV

Since dV = 0, dU = dQ.

Hence, $dU = nC_v dT$ since $dQ = nC_v dT$.

Where

n = Number of moles,

 C_v = Specific heat at constant volume,

dT =Change in temperature

Putting the values in the above formula, we get

Therefore, dU = 1 mol x (6 x 8.314) J/mol/K x 50K

= 2494.2 J(Ans)

(c) Adiabatic process: dQ(heat change) = 0. Therefore,

dU(change in internal energy) = dW(work done)

Since dQ = dU + dW.

For an adiabatic process, $dW = dT/(\gamma-1)$.

dW = work done, dT = change in temperature

 γ = C_p/C_v = 7/6 , C_p = specific heat capacity at constant

pressure, C_v = specific heat capacity at constant volume

Therefore dU = - dW = - $\frac{nRdT}{\gamma - 1}$ = (8.314X50)/(7/6 - 1)

= 2494.2 J (Ans)

Answer.6

Given:

Pressure, $p = 1.0 \times 10^5$ Pa,

Temperature, T = 300K,

Universal gas constant, R = 8.314 J/kg/mol

V(volume) = 1.0×10^3 cm³ = 0.001 m³

Formula used:

1. Ideal gas equation: PV = nRT.

Where,

P = pressure,

V = volume,

n = number of moles,

R = universal gas constant = 8.314 J/kg/mol,

T = absolute temperature

2. Number of moles n = PV/RT = 100000x0.001/(8.314*300) = 0.04 mol

3. First law of thermodynamics: dQ = dU + dW = dU + PdV,

Where,

dQ = heat supplied,

dU = change in internal energy,

dW = PdV = work done, where P = pressure, dV = change in volume.

Since volume is constant, $dV = 0 \Rightarrow dW = 0$.

Hence, dQ = dU.

Heat(dQ) = 2 cal = $nC_v dT = 0.04 \text{ mol } x C_v x 1K$,

Where n = number of moles, C_v = specific heat at constant volume, dT = rise in temperature.

=> $C_v = 50 \text{ cal/mol/K}$. = $(50 \times 4.2 \times 10^7)$ erg/cal x cal/mol/K = $2.1 \times 10^9 \text{ erg/mol/K}$ = 210 J/mol/K,

Since heat(J) = mechanical equivalent of heat x heat(cal) = 4.2 x heat(cal), and

$$1 J = 10^7 erg$$

We know, $C_p = (C_v + R)$,

Where C_p = specific heat at constant pressure, C_v = specific heat at constant volume, R = universal gas constant = 8.314 J/kg/mol

Therefore, C_p = (210+8.314) J/mol/K = 218.314 J/mol/K.

Therefore, heat required to raise the temperature by 1°C at constant pressure = $nC_p dT$,

Where n = number of moles, C_p = specific heat at constant pressure, dT = rise in temperature.

Hence, substituting, heat = $(0.04 \times 218.314 \times 1) \text{ J} = 8.737 \text{ J} = (8.737/4.2) \text{ cal} = 2.08 \text{ cal} (since 1 \text{ J} = 4.2 \text{ cal})(Ans)$

Given:

(a) Pressure(P) = 2.0×10^5 Pa, dV(change in volume) = (200-100) cm³ = 10^{-4} m³, since 1 m³ = 10^6 cm³

Heat(dQ) = 50 J.

Formula used:

Now we know, dQ = dU(change in internal energy) + dW(work)=

dU + PdV (first law of thermodynamics),

Where P = pressure, dV = change in volume.

 $= dU = dQ - PdV = (50 - (2.0 \times 10^5 \text{ x } 10^{-4})) \text{ J} = 30 \text{ J} (\text{Ans})$

(b) For constant pressure, from equation of state PV/T = constant,

Where P = pressure, V = volume, T = temperature.

Hence we get: $\frac{V_1}{T_1} = \frac{V_2}{T_2}$, where

 V_1 (initial volume) = 100 cm³, V_2 (final volume) = 200 cm³, T_1 = 300K

$$= T_2 = \frac{V_2 T_1}{V_1} = (300 \times \frac{200}{100}) K = 600 K.$$

Therefore, PdV = nRdT (for more than one mole),

Where P = pressure, dV = change in volume, n = number of moles, R = universal gas constant = 8.314 J/kg/mol, dT = change in temperature.

 $= 2.0 \times 10^5 \text{ x } 10^{-4} = \text{n x } 8.314 \text{ x } 300 \text{ (since T2-T1 = dT = 300K)}$

Therefore, $n = 20/(R \times 300) = 0.008 \text{ mol}$ (Ans).

(c) $dQ(heat) = 50 = nC_p dT$ (at constant pressure),

Where n = number of moles, C_p = specific heat at constant pressure, dT = rise in temperature.

 $\Rightarrow 50 = 0.008 \text{ x C}_{p} \text{ x 300}$

=> C_p = 20.83 J/mol/K. (Ans)

(d) At constant volume, dU(change in internal energy) = $dQ(heat) = nC_v dT$ (since work done dW = PdV = 0, where P = pressure, dV = change in volume), from first law of thermodynamics.

n= number of moles, C_v = specific heat capacity at constant volume, dT = change in temperature.

 $\Rightarrow 30 = 0.008 \text{ x C}_{v} \text{ x } 300$

=> C_v = 12.5 J/mol/K. (Ans)

Answer.8

Given: Amount of heat added(dQ) = Q

Amount of work done(dW) = Q/2.

Formula used:

dQ(heat) = dU(internal energy) + dW(work done).

Here, heat = Q and Work = Q/2(given)

$$\Rightarrow U = Q - \frac{Q}{2} = \frac{Q}{2}.$$

We can write U = $nC_v dT$ and Q = nCdT, where n = no of moles, C_v = specific heat capacity at constant volume(when dQ = dU), C = molar heat capacity and dT = change in temperature.

Therefore,
$$nCvdT = \frac{nCdT}{2} \implies C = 2C_v$$
.

For a monoatomic ideal gas, we know that $C_v = (3R/2) J/kg/mol$,

Where R = universal gas constant = 8.314 J/kg/mol

Therefore, $C = 2^{*}(3R/2) = 3R J/kg/mol.$ (Ans)

Given:

 $\mathsf{P}=\mathsf{k}\mathsf{V}\ldots$

Where,

P = pressure,

V = volume,

k = constant.

Formula used:

Equation of state of ideal gas:

PV = nRT = constant ... (ii)

Where,

n is the number of moles of the gas,

R is the gas constant,

T is the temperature,

 $P = pressure, V^2$

T = temperature.

From (i), multiplying by dV on both sides:

PdV = kVdV.

Integrating from V = V1 to V2, we get

$$\int dV = k \int_{V_1}^{V_2} V \, dV$$

= $k \left[\frac{V \times V}{2}\right]$ with lower limit V₁ and upper limit V₂

$$=k\frac{[v_2^2-v_1^2]}{2}$$

Now, we know, PV = nRT - equation of state,

Where P = pressure, V = volume, n = number of moles, R = universal gas constant, T = temperature

Hence we can write, $V_1 = nRT_1/P_1$. Since $P_1 = kV_1$, this becomes:

 $kV_1^2 = nRT_1$. Similarly, $KV_2^2 = nRT_2$,

 P_1 , V_1 , T_1 - Pressure, volume, temperature of first gas

 P_2, V_2, T_2 - Pressure, volume, temperature of second gas

Therefore, subsituting, the above equation becomes:

$$= \frac{k}{2} \times nR \frac{[T_2 - T_1]}{k} = \frac{nR}{2} dT (dT = T_2 - T_1) \dots (iii)$$

Now, $V = \frac{nRT}{P} \Rightarrow V^2 = \frac{nRT}{k}$ (since P = kV) ... (iv)

But $Q = U + \int PdV$ (first law of thermodynamics), where Q = heat, U = change in internal energy, W = total work done = $\int PdV$

$$=$$
 nCdT = nC_vdT + (nR/2)dT

(since
$$Q = nCdT$$
 and $U = nC_v dT$)

$$=> C = C_v + nR/2$$
 (proved),

Where

n = number of moles,

C = specific heat capacity,

 C_v = specific heat capacity at constant volume,

R = universal gas constant,

dT = rise in temperature.

Answer.10

Given:

 $p = aV^b$

p = pressure, V = volume, a and b are constants.

Formula used:

We know, $Q = U + \int p dV$ from first law of thermodynamics,

where Q = change in heat, U = change in internal energy and $\int pdV = W = total work$ done, p = pressure, V = volume.

Since Q = nCdT, and $U = nC_v dT$, we get

$$nCdT = nCvdT + \int_{V1}^{V2} aV^b dV$$
... (ii), n = no. of moles, C = specific

heat capacity, and C_v = specific heat capacity at constant volume,

dT = change in temperature,

Since specific heat capacity is 0(given),

$$nCvdT + \frac{a}{b+1}X[V_2^{b+1} - V_2^{b+1}] = 0$$
...(iii)

(after integration of
$$\int a V^b dV$$
 from V₁ to V₂)

Now, from equation of state, PV = nRT,

Where

V = volume,

n = number of moles,

R = universal gas constant,

T = temperature.

Substituting p = aV^b from (i):

 $aV^{b+1} = nRT$

$$= V^{b+1} = nRT/a \dots (iv)$$

Substituting (iv) in (iii),

$$nCvdT = -\frac{a}{b+1}X\frac{nR}{a}X(T2 - T1)$$
$$nCvdT = -\frac{nR}{b+1}XdT \text{ (Since } (T_2 - T_1) = dT)$$

=>
$$b + 1 = -\frac{R}{Cv} = -\frac{Cp - Cv}{Cv} = -(\frac{Cp}{Cv} - 1) = -(\gamma - 1) = 1 - \gamma$$

 \Rightarrow b = - γ (Ans)

Answer.11

We know Cp/Cv= γ , R=Cp-Cv,

where the molar heat capacity C, at constant pressure, is represented by Cp, at constant volume, the molar heat capacity C is represented by Cv and R is the universal gas constant.

Now,

$$Cv = \frac{R}{\gamma - 1}$$

For the first ideal gas,

$$Cv1 = \frac{R}{\gamma - 1}$$
$$Cp1 = \frac{\gamma R}{\gamma - 1}$$

Where C_{p1} and C_{V1} is the molar heat capacity at constant pressure and constant volume

Similarly, for the second ideal gas

$$Cv2 = \frac{R}{\gamma - 1}$$
$$Cp2 = \frac{\gamma R}{\gamma - 1}$$

Where $C_{\rm p2}$ and $C_{\rm V2}$ is the molar heat capacity at constant pressure and constant volume

Given,

n1:n2 = 1:2

i.e

 $dU_1 = nC_{V1}dT$

 $dU_2=2nC_{V2}Dt$

When gas is mixed,

$$nC_{v1}dT + 2nC_{v2}dT = 3nC_{v}dT$$
$$C_{v} = \frac{C_{v1} + 2C_{v2}}{3}$$
$$C_{v} = \frac{\left(\frac{R}{\gamma - 1} + \frac{2R}{\gamma - 1}\right)}{3} (1)$$

Also,

$$C_p = \gamma C_v = \frac{\gamma R}{\gamma - 1} \ (2)$$

From (1) and (2)

$$\frac{C_p}{C_v} = \gamma$$

Answer.12

Given,

 $C_{\rm P1}$ = 2.5 R, $C_{\rm V1}$ = 1.5R for helium

 C_{P2} = 3.5 R, C_{V2} = 2.5 R for hydrogen

n1=n2=1

We know $dU = nC_v dT$

Where dU is the change in internal energy, n is the number of moles, Cv is the molar heat capacity at constant volume and dT is the change in temperature.

For the mixture,

$$(n1 + n2)C_{v}dT = n1C_{v1}dT + n2C_{v2}dT$$
$$C_{v} = \frac{n1C_{v1}dT + n2C_{v2}dT}{(n1 + n2)dT}$$
$$C_{v} = \frac{1.5RdT + 2.5dT}{(2)dT} = 2R$$

Also,

$$C_p = C_v + R = 3R$$
$$\gamma = \frac{C_p}{C_v} = 1.5$$

Answer.13

Given, n=1/2, γ =5/3, R=25/3 J/Kmol

a) By ideal gas equation,

PV = nRT, where P, V and T are the pressure, volume and absolute temperature; n is the number of moles of gas; R is the ideal gas constant.

Here temperature at a, $Ta = \frac{PV}{nR} = \frac{5000 \times 10^{-6} \times 100 \times 10^{3}}{\frac{1}{2} \times \frac{25}{3}} = 120 K$

Here temperature at b,

$$Tb = \frac{PV}{nR} = \frac{10000 \times 10^{-6} \times 100 \times 10^{3}}{\frac{1}{2} \times \frac{25}{3}} = 240 \, K$$

Here temperature at c,

$$Tc = \frac{PV}{nR} = \frac{1000 \times 10^{-6} \times 200 \times 10^{3}}{\frac{1}{2} \times \frac{25}{3}} = 480 K$$

Here temperature at d,

$$Td = \frac{PV}{nR} = \frac{5000 \times 10^{-6} \times 200 \times 10^{3}}{\frac{1}{2} \times \frac{25}{3}} = 240 K$$

b) Here ab is an isobaric process where heat supplied dQ can be expressed as

_ _

$$dQ = nCpdT = \frac{1}{2}\frac{R\gamma}{\gamma - 1}(Tb - Ta) = \frac{1}{2}\frac{\left(\frac{25}{3} \times \frac{5}{3}\right)}{\frac{5}{3} - 1}(240 - 120) = 1250J$$

Here bc is an isochoric process where heat supplied dQ is

$$dQ = nCvdT = \frac{1}{2}\frac{R}{\gamma - 1}(Tc - Tb) = \frac{1}{2}\frac{\frac{25}{3}}{\frac{5}{3} - 1} \times 240 = 1500 J$$

c) Heat liberated in cd, isobaric process dQ is

$$dQ = -nCpdT = \frac{1}{2}\frac{R\gamma}{\gamma - 1}(Td - Tc) = -\frac{1}{2}\frac{\left(\frac{25}{3} \times \frac{5}{3}\right)}{\frac{5}{3} - 1}(-240) = 2500 J$$

Heat liberated in da, isochoric process dQ is

$$dQ = -nCvdT = \frac{1}{2}\frac{R}{\gamma - 1}(Ta - Td) = -\frac{1}{2}\frac{\frac{25}{3}}{\frac{5}{3} - 1}(120 - 240) = 750 J$$

Answer.14

Let (P1, V1, T1), (P2, V2, T2), (P3, V3, T3) denote the pressure, volume and temperature at a, b and c respectively.

(a)For the process ab volume is constant

I.e. by ideal gas equation,

$$\frac{P1}{T1} = \frac{P2}{T2}$$

i. e. $\frac{100}{300} = \frac{200}{T2}$

T2 = 600 K, temperature at b

For the process bc, pressure is constant.

By ideal gas equation,

$$\frac{V2}{T2} = \frac{V3}{T3}$$

$$I.\,e.\frac{100}{600} = \frac{150}{T3}$$

T3 = 900 K, temperature at c

(b)Here process ab is isochoric i.e. $W_{ab}=0$

For process bc, P=200 kPa, change in volume is 50 cm^3 from b to c

Work done = $PdV = 200 \times 10^3 \times 50 \times 10^{-6} = 10 J$

(c) From the first law of thermodynamics,

dQ = dU + dW

Where dQ is the amount of heat supplied

As ab is isochoric process dW=0

$$dQ_{ab} = dU = nC_{v}dT$$

= $\frac{PV}{RT} \times \frac{R}{\gamma - 1} dT$
= $\frac{200 \times 10^{3} \times 150 \times 10^{-6}}{600 \times .67} \times 300 = 14.925$

Here bc is an isobaric process where heat supplied dQ by first law of

Thermodynamics is

 $dQ_{bc} = dU = nC_p dT$

$$= \frac{PV}{RT} \times \frac{\gamma R}{\gamma - 1} dT$$

$$= \frac{200 \times 10^{3} \times 150 \times 10^{-6}}{900 \times 8.3} \times \frac{1.67 \times 8.3}{.67} \times 300 = 24.925$$
(d) dQ = dU + W
Now, dU = dQ - W
$$= Heat \ supplied - Work \ done$$

$$= (24.925 + 14.925) - 10$$

$$= 29.85$$

Here,

m1 = Mass of gas present = 3 g,
$$\theta$$
1 = 20°C, θ 2 = 100°C

m2 = Mass of steam condensed = 0.095 g, L = 540 Cal/g = $540 \times 4.2 \text{ J/g}$

In Joly's differential steam calorimeter,

$$C_{v} = \frac{m2L}{m1(\theta 2 - \theta 1)}$$
$$Cv = \frac{0.095 \times 540 \times 4.2}{3 \times 80} = 0.89 J/gK$$

Answer.16

Here γ = 1.5, V1=4, V2=3.Let P1 and P2 be the initial and final pressure

(a)Since it is an adiabatic process, So PV^{γ} = const.

I.e.
$$\frac{P2}{P1} = \left(\frac{V1}{V2}\right)^{\gamma} = \left(\frac{4}{3}\right)^{1.5} = 1.54$$

(b)Also for an adiabatic process, $TV^{\gamma - 1}$ =constant

I.e.

$$T1V1^{\gamma-1} = T2V2^{\gamma-1}$$
$$\Rightarrow \frac{T2}{T1} = \left(\frac{V1}{V2}\right)^{\gamma-1}$$
$$= \left(\frac{4}{3}\right)^{1.5-1} = 1.154$$

Answer.17

Here given, $P1 = 2.5 \times 10^5$ Pa, V1 = 100 cc, T1 = 300 K, V2=50 cc

(a)Since it is an adiabatic process, So PV^{γ} = const.

I.e.
$$\frac{P2}{P1} = \left(\frac{V1}{V2}\right)^{\gamma} = \left(\frac{100}{50}\right)^{1.5}$$

 $P2 = 2.5 \times 10^5 \times 2^{1.5} = 7.1 \times 10^5 \ Pa$

(b) Also for an adiabatic process, $TV^{\gamma - 1}$ =constant

I.e.

 $T1V1^{\gamma-1}=T2V2^{\gamma-1}$

$$\Rightarrow \frac{T2}{T1} = \left(\frac{V1}{V2}\right)^{\gamma-1}$$

 $I.e.T2 = 300 \times 2^{0.5} = 424 K$

(c) Work done by the gas in the process

$$W = \frac{nR(T2 - T1)}{\gamma - 1} = \frac{P1V1(T2 - T1)}{T1(\gamma - 1)}$$
$$= \frac{2.5 \times 10^5 \times 100 \times 10^{-6}(424 - 300)}{300(1.5 - 1)} = 21J$$

Given $\gamma = 1.4$,

Initial pressure, P1=2 atm

Final pressure, P2=1 atm

Initial Temperature, T1=20°C=293 K

Here bursting of tire is an adiabatic process,

 $T1^{\gamma}P1^{1-\gamma} = T2^{\gamma}P2^{1-\gamma}$

$$\Rightarrow T2^{\gamma} = \frac{T1^{\gamma} P 1^{1-\gamma}}{P2^{1-\gamma}} = 293^{1.4} \times \left(\frac{2}{1}\right)^{-0.4}$$
$$\Rightarrow T2 = 240.3 K$$

Answer.19

Given, P1 = 100 KPa = 10^5 Pa, V1 = 400 cm³, T1 = 300 K,

 $C_{\rm P}/C_{\rm V} = 1.5$

(a)Suddenly compressed to $V2 = 100 \text{ cm}^3$ I.e. it is an adiabatic process

 $\therefore PV^{\gamma} = const.$

I.e.
$$\frac{P2}{P1} = \left(\frac{V1}{V2}\right)^{\gamma} = \left(\frac{400}{100}\right)^{1.5}$$

 $P2 = 10^5 \times 4^{1.5} = 800 \, KPa$

Also,

 $T1P1^{1-\gamma} = T2P2^{1-\gamma}$

$$\Rightarrow T2 = \frac{T1P1^{1-\gamma}}{P2^{1-\gamma}} = 300 \times \left(\frac{100}{400}\right)^{-0.5}$$

 $\Rightarrow T2 = 600 K$

(b) Even though the container is slowly compressed the walls are adiabatic so heat transferred is 0.

Thus the values remain same

i.e. P2 = 800 KPa, T2 = 600 K.

Answer.20

Given $C_P/C_V = \gamma$

Let $P1=P_0$ be the Initial Pressure, $V1=V_0$ be the Initial Volume

(a)Since the volume is slowly compressed, temperature remains constant I.e. Isothermal compression. Let P2 and V2= $V_0/2$ be the pressure and volume after slow compression

$$\therefore P1V1 = P2V2$$

 $\Rightarrow P_0 V_0 = P 2 V_0 / 2$

$$\Rightarrow P2 = 2P_0$$

When volume is suddenly compressed, it is an adiabatic process .Let P3 and V3 be the pressure and volume after sudden compression .i.e.

 $PV^{\gamma} = const.$

I.e.
$$\frac{P3}{P2} = \left(\frac{V2}{V3}\right)^{\gamma} = \left(\frac{V_0/2}{V_0/4}\right)^{\gamma} = (2)^{\gamma}$$

Substituting value of P2=2P₀

$$P3 = P_0 2^{\gamma+1}$$

(b) Since the volume is suddenly compressed, I.e. it is an adiabatic process Let P2 and $V2=V_0/2$ be the pressure and volume after sudden compression, then

 $PV^{\gamma} = const.$

I.e.
$$\frac{P2}{P1} = \left(\frac{V1}{V2}\right)^{\gamma} = \left(\frac{V_0}{V_0/2}\right)^{\gamma} = (2)^{\gamma}$$

Substituting value of P1=P0

$$P2 = P_0 2^{\gamma}$$

Now, since the volume is slowly compressed, temperature remains constant I.e. Isothermal compression. Let P3 and $V3=V_0/4$ be the pressure and volume after slow compression.

$$\begin{array}{l} \therefore P2V2 \ = \ P3V23 \\ \Rightarrow P_0 \times 2^{\gamma} (V_0/2) \ = P3V_0/4 \\ \Rightarrow P3 \ = P_0 2^{\gamma+1} \end{array}$$

Answer.21

Given $C_P/C_V = \gamma$, initial pressure P1=P₀ and initial volume V1= V₀

a) Since the gas is isothermally taken to pressure $P2=P_0/2$

- $\therefore P1V1 = P2V2$
- $\Rightarrow P_0 V_0 = P_0 / 2V2$
- $\Rightarrow V2 = 2V_0$

Let $P3=P_0/4$ and V3 be the pressure and volume after adiabatic compression.

Then,

 $PV^{\gamma} = const.$

$$I.e \frac{P3}{P2} = \left(\frac{V2}{V3}\right)^{\gamma}$$
$$\frac{P_0 / 4}{P_0 / 2} = \left(\frac{V2}{V3}\right)^{\gamma}$$

Substituting value of V2=2V $_0$

 $V3 = 2^{1/\gamma}(2) V_0 = 2^{(1+1/\gamma)} V_0$

(b)Here $P1=P_0$, $P2=P_0/2$ and the process is adiabatic. Let $V1=V_0$ be the initial volume and V2 be the volume after process.

Then,

 $PV^{\gamma} = const.$

$$I.e.\frac{P2}{P1} = \left(\frac{V1}{V2}\right)^{\gamma}$$
$$\frac{P_0/2}{P_0} = \left(\frac{V_0}{V2}\right)^{\gamma}$$

I.e.
$$V2 = V_0(2)^{1/\gamma}$$

Let P3= $P_0/4$ and V3 be the pressure and volume after isothermal process Then,

$$P2V2 = P3V3$$

$$\Rightarrow \frac{P_0}{2}V_0(2)^{1/\gamma} = \frac{P_0}{4}V3$$

$$\Rightarrow V3 = 2^{(1+1/\gamma)}V_0$$

Given P1 = 150 KPa = 150×10^3 Pa, V1 = 150 cm³, V2=50 cm³, T1 = 300 K

(a)By ideal gas equation,

PV = nRT, where P, V and T are the pressure, volume and absolute temperature; n is the number of moles of gas; R is the ideal gas constant

I.e.

$$n = \frac{PV}{RT} = \frac{150 \times 10^3 \times 150 \times 10^{-6}}{8.3 \times 300} = .009 \text{ moles}$$

(b)We know Cp/Cv=γ R=Cp-Cv,

where the molar heat capacity C, at constant pressure, is represented by Cp, at constant volume, the molar heat capacity C is represented by Cv

Now,

$$Cv = \frac{R}{\gamma - 1}$$
$$Cv = \frac{8.3}{0.5} = 16.6 J$$

(c) Since the process is adiabatic,

 $PV^{\gamma} = const.$

$$I.e \frac{P2}{P1} = \left(\frac{V1}{V2}\right)^{\gamma}$$

$$P2 = P1 \times \left(\frac{V1}{V2}\right)^{\gamma}$$

$$P2 = 150 \times 10^{3} \times \left(\frac{150}{50}\right)^{1.5} = 780 \, Kpa$$

Also, as the process is adiabatic,

$$T1^{\gamma}P1^{1-\gamma} = T2^{\gamma}P2^{1-\gamma}$$

$$\Rightarrow T2^{\gamma} = \frac{T1^{\gamma}P1^{1-\gamma}}{P2^{1-\gamma}} = 300^{1.5} \times \left(\frac{150}{780}\right)^{-0.5}$$

$$\Rightarrow T2 = 519.74 K$$

(d) From the first law of thermodynamics,

dQ = dU + dW, where dQ is the amount of heat supplied which is zero in an adiabatic process.

i.e.

dW = -dU

dW = -nCvdT, where n is the number of moles, Cv is the molar heat capacity at constant volume and dT is the change in temperature

 $dW = -0.009 \times 16.6 \times (520 - 300)$

dW = -33 J

(e)Change in internal energy, dU is

dU = nCvdT, where n is the number of moles, Cv is the molar heat capacity at constant volume and dT is the change in temperature.

$$dU = 0.009 \times 16.6 \times (520 - 300)$$

dU = 33 J

Answer.23

Let V_A , VB, VC be the volume of three gases and T_A , TB, TCbe the temperature of A, B, C gas

Given, $T_A=T_B=T_C$, $V_A=V_B=V_C$

Here A is undergoing an isothermal process, where V1= V_A, V2=2V_A

Let $P1 = P1_A$ and $P2 = P2_A$ be the initial and final pressures,

Then,

P1V1 = P2V2

$$\Rightarrow P1_A V_A = P2_A(2V_A)$$

$$\Rightarrow P2_A = P1_A/2$$

Here B is adiabatic,

 PV^{γ} = const, where V1= V_B, V2=2V_B

Let $P1 = P1_B$ and $P2 = P2_B$ be the initial and final pressures,

$$I.e_{\overline{P1}}^{\underline{P2}} = \left(\frac{V_B}{2V_B}\right)^{\gamma}$$
$$\frac{P2_B}{P1_B} = \left(\frac{1}{2}\right)^{1.5}$$
$$P2_B = P1_B \left(\frac{1}{2}\right)^{1.5}$$

Here C is isobaric ,the pressure remains constant and equal to $P1_c$

Now, as the final pressures are equal for all the gases

$$\frac{P1_A}{2} = P1_B \left(\frac{1}{2}\right)^{1.5} = P1_C$$

 $P1_B: P1_B: P1_C = 2: 2^{1.5}: 1$, ratio of the initial pressures

Answer.24

Let P1 = Initial Pressure, V1 = Initial Volume, P2 = Final Pressure, V2 = Final Volume

Here A is expanded isothermally,

I.e. the work done,

$$W_A = nRT \ln\left(\frac{V2}{V1}\right)$$

Also, B is expanded adiabatically, i.e.

$$W_B = \frac{P1V1 - P2V2}{\gamma - 1}$$

Given $W_A = W_B$

i.e.

$$nRT1\ln\left(\frac{V2}{V1}\right) = \frac{P1V1 - P2V2}{\gamma - 1}$$
(1)

In an adiabatic process,

 $PV^{\gamma} = const,$

$$I.e \frac{P2}{P1} = \left(\frac{V1}{2V1}\right)^{\gamma}$$
$$P2 = P1 \left(\frac{1}{2}\right)^{\gamma}$$

Substituting in (1)

$$nRT1\ln\left(\frac{V2}{V1}\right) = \frac{P1V1 - P2V2}{\gamma - 1}$$
$$nRT1\ln(2) = \frac{P1V1(1 - \frac{1}{2\gamma} \times 2)}{\gamma - 1}$$

We know, PV= nRT by ideal gas equation i.e.

$$\ln(2) = \frac{\left(1 - \frac{1}{2\gamma} \times 2\right)}{\gamma - 1}$$

 $\ln 2(\gamma - 1) = 1 - 2^{1-\gamma}$, the required relation

Given $\gamma = 1.5$ T=300 K, initial volume V1=1 L, Final volume V2=1/2 L. Let P1 and P2 be the initial and final pressures

(a)Here the process is adiabatic since volume is changed suddenly,

i.e.

 $\frac{P2}{P1} = \left(\frac{V1}{V2}\right)^{\gamma}$ $P2 = P1 \times (2)^{\gamma}$ $\frac{P2}{P1} = (2)^{1.5}$

Work done in adiabatic process,

$$W = \frac{P1V1 - P2V2}{\gamma - 1}$$
$$W = \frac{10^5 \times 10^{-3} - 2^{1.5}(10^5)500 \times 10^{-3}}{1.5 - 1} = -82J$$

(c) Here dQ=0, as it an adiabatic process

By first law of thermodynamics,

$$dQ = dU + dW$$

i.e. dU = -dW = 82 J

(d)For an adiabatic process, let TI and T2 be initial and final temperature

 $TV^{\gamma -1}$ =constant

I.e.

 $T1V1^{\gamma-1} = T2V2^{\gamma-1}$

$$\Rightarrow \frac{T2}{T1} = \left(\frac{V1}{V2}\right)^{\gamma-1}$$
$$\Rightarrow T2 = T1 \left(\frac{V1}{V2}\right)^{1.5-1} = 300 \times 2^{0.5}$$
$$\Rightarrow T2 = 424 K$$

(e)Here the pressure is kept constant, i.e. isobaric

Work done in an isobaric process , $W = P \bigtriangleup V = nRdT$

Here,
$$n = \frac{PV}{RT} = \frac{(10^5 \times 10^{-3})}{R \times 300} = \frac{1}{3R}$$

Work done, $W = nRdT = \frac{1}{3R}R(300 - 424) = -41.4J$

(f)Here the process is isothermal.

Work done,
$$W = nRT ln\left(\frac{V2}{V1}\right) = \frac{1}{3R} \times R \times ln(2)$$

W = 103 J

(g)Work done in the cycle, $W_{total} = -82 - 41.4 + 103 = -20.4 J$

Answer.26

Given:

The walls of the cylindrical tube and the separator are made with

adiabatic material. The separator can be slid in the tube by

external mechanism.

An ideal gas of $\gamma = 1.5$ is injected in the two aides of at equal pressure.

It is now slid to a position where it divides tube in the ratio 1:3.

The initial volume of the two aides are equal let's say V/2,

Where, the total volume of the tube is V.

Now say the, left part of tube has V/4 volume and the right side has 3V/4 volume so that the ratio between them is 1:3.

In adiabatic process, $PV^{\gamma} = K$ (K = non zero constant)

Where P is the pressure of the gas and V is the volume and $\gamma = \frac{c_p}{c_v}$

For ideal gas, PV = nRT

Where P is the pressure, V is the volume, T is the temperature of the gas and R is the gas constant and n is the number of moles of the gas.

Putting this in the adiabatic process condition we get,

 $TV^{\gamma-1} = K'$ (K' is a non-zero constant)

Therefore, $T_1 V_1^{\gamma - 1} = T_2 V_2^{\gamma - 1}$ $\Rightarrow T_1 \left(\frac{v}{2}\right)^{(1.5-1)} = T_2 \left(\frac{v}{4}\right)^{(1.5-1)}$ $\Rightarrow T_1 \left(\frac{v}{2}\right)^{(0.5)} = T_2 \left(\frac{v}{4}\right)^{(0.5)}$ $\Rightarrow T_1 \left(\frac{1}{2}\right)^{(0.5)} = T_2 \left(\frac{1}{4}\right)^{(0.5)}$ $\Rightarrow T_1 \left(\frac{1}{2}\right)^{(0.5)} = T_2 \left(\frac{1}{2}\right)$ $\Rightarrow T_2 = \sqrt{2}T_1$

Again for the other part of the tube,

$$\Rightarrow T_1' \left(\frac{v}{2}\right)^{(1.5-1)} = T_2' \left(\frac{3v}{4}\right)^{(1.5-1)}$$

$$\Rightarrow T_1' \left(\frac{v}{2}\right)^{(0.5)} = T_2' \left(\frac{3v}{4}\right)^{(0.5)}$$

$$\Rightarrow T_1' \left(\frac{1}{2}\right)^{(0.5)} = T_2' \left(\frac{v}{4}\right)^{(0.5)}$$

$$\Rightarrow T_1'_2 = \sqrt{\frac{2}{3}} T_1'_1$$

As initially the gases were at the same pressure and volume, the temperatures would be the same as well.

Therefore, $T_1 = T'_1$

Therefore, ,
$$T_2 = T_2' = \sqrt{2} : \sqrt{\frac{2}{3}} = \sqrt{3} : 1$$

Therefore the ratio of the final temperatures will be $\sqrt{3}$: 1

Given:

Gasses in both the vessels are at pressure of 75cm of mercury.

Therefore, $P = 0.75 \times 13600 \times 9.8 Pa = 99960 Pa$

The volume of the vessel is $200 cm^3 = 0.0002 m^3$

The temperature of the gas is 300K

(a) For Ideal gasses,

PV = nRT where P,V and T are the pressure, volume and temperature of the gas, n is the number of moles, and R is the gas constant.

$$\Rightarrow n = \frac{PV}{RT} = \frac{99960 \times 0.0002}{8.314 \times 300} = 0.008$$

Therefore, number of moles = 0.008

(b) The specific heat of the gas at constant volume is $C_V = 12.5 \text{ J K}^{-1} \text{ mol}^{-1}$

Therefore, $nC_V T = Q$

Where n is the number of moles, T is the rise in temperature, Q is the heat given.

Therefore, at constant volume, if we supply 5J and 10J heat to the vessels, the rise of temperature will be $\frac{5}{0.008 \times 12.5} = 50K$ and $\frac{10}{0.008 \times 12.5} = 100K$

So the change in pressure in the vessels will be governed by

$$P = \frac{nRT}{V}$$

So for the first vessel, Change in pressure

$$P = \frac{0.008 \times 8.314 \times 50}{0.0002} = 16628 \, Pa$$

For the second vessel $P = \frac{0.008 \times 8.314 \times 100}{0.0002} = 33256 Pa$

Therefore, the difference of pressure of the two vessels is

33256 - 16628 = 16628Pa

Which is equivalent to $\frac{16628}{13600 \times 9.8} = 0.125m = 12.5cm$ of mercury.

Therefore, the height of the mercury in the manometer tube is 12.5cm

Answer.28

Given:

Two vessels with adiabatic walls, one contains 0.1g of helium ($\gamma = 1.67$, M = 4 g mol⁻¹) and the other contains some amount of hydrogen ($\gamma = 1.4$, M = 2g mol⁻¹)

The gasses are given the same amount of heat.

The temperature rises through the same amount.

0.1g of helium = 0.1/4 mole = 0.025mole

Let there be n moles of hydrogen in the other vessel.

$$\gamma = \frac{c_p}{c_v}$$
 and $C_p - C_V = R$ so, $C_V = \frac{R}{\gamma - 1}$

As the vessels are of constant volume there will be no work done by the gasses. The heat supplied will totally be used to increase internal energy.

Therefore, $Q = nC_v T$ where Q is the heat supplied, n is the number of moles, C_v is the specific heat capacity of gas at constant volume, T is the change in temperature.

For helium, $Q = \frac{0.025 \times 8.314}{1.67^{-1}}T = 0.31 T$

For hydrogen, $Q = \frac{n \times 8.314}{1.4^{-1}}T = 20.8 nT$ we assume for both cases the rise of temperature is T.

As per question,

$$0.31T = 20.8 nT$$

$$\Rightarrow n = \frac{0.31}{20.8}$$

 \Rightarrow n = 0.015

Again, Molar mass of hydrogen = $2g \text{ mol}^{-1}$

Therefore, 0.015 mole of hydrogen = $0.015 \times 2 = 0.03g$ hydrogen

Thus, there is 0.03g of hydrogen in the vessel.

Answer.29

Given:

Two vessels A and B of equal volume V_0 are connected by a narrow tube which can be closed by a valve.

Vessels contain an ideal gas $\left(\frac{c_P}{c_V} = \gamma\right)$ at atmospheric pressure p_0 and atmospheric temperature T_0 .

The walls of the vessel A are diathermic and those of B are adiabatic.

The pistons are slowly pulled out to increase the volumes of the vessels to double the original value.

(a) As the pistons are moved slowly to increase the volume, the expansion of gas in the diathermic vessel will be an isothermic process thus the temperature will be fixed at T_0 . P,V and T represent the pressure, volume and temperature of the gasses and subscripts 1 and 2 denote initial and final state respectively.

Thus,

$$P_1V_1 = P_2V_2$$

$$\Rightarrow P_0V = 2P_2V$$

$$\Rightarrow P_2 = \frac{P_0}{2}$$

For the adiabatic vessel,

$$P_1 V_1^{\gamma} = P_2 V_2^{\gamma}$$
$$\Rightarrow P_0 V^{\gamma} = P_2 (2V)^{\gamma}$$
$$\Rightarrow P_2 = \frac{P_0}{2^{\gamma}}$$

Again for ideal gasses, $\frac{P_1V_1}{T_1} = \frac{P_2V_2}{T_2}$

$$\Rightarrow \frac{P_0 V}{T_0} = \frac{2P_0 V}{2^{\gamma} T_2}$$
$$\Rightarrow T_2 = \frac{T_0}{2^{\gamma - 1}}$$

Thus the temperature and pressure in the diathermic vessel will T₀ and P₀/2 and in the adiabatic vessel, $\frac{T_0}{2^{\gamma-1}}$ and $\frac{P_0}{2^{\gamma}}$.

(b) When the valve is open, the temperature will remain T_0 throughout. Thus, there will be no change in temperature in the diathermic vessel so there will be change in pressure as well. For the gas in the diathermic vessel,

 $P_1 = \frac{nRT_0}{4V}$ and for the adiabatic vessel $P_2 = \frac{nRT_0}{4V}$ Therefore $P_1 = P_2$ Again, $P_0 = P_1 + P_2$ Thus, $P_1 = P_2 = \frac{P_0}{2}$ Thus the final temperature, when the valve is open will be T_0 and the final pressure will be $\frac{P_0}{2}$.

Answer.30

Given:

An adiabatic cylindrical tube of volume V_0 is divided in two parts by a frictionless adiabatic separator.

An ideal gas at pressure p_1 and temperature T_1 is injected into the left part and another ideal gas at pressure p_2 and temperature T_2 is injected into the right part.

(a) When the piston is slowly moved to the equilibrium position, one side increases in volume when the other side decreases.

The processes will be adiabatic,

For the left part,

 $P_1 V_1^{\gamma} = P_2 V_2^{\gamma}$ Where, subscript 1 and 2 represent the initial and the final state.

$$\Rightarrow P_1(\frac{V_0}{2})^{\gamma} = PV_2^{\gamma}....(1)$$

And for the right part,

$$P_2(\frac{V_0}{2})^{\gamma} = PV_1^{\gamma}.....(2)$$

We are assuming P to be the common pressure.

Dividing (1) by (2) we get,

$$\frac{V_2^{\gamma}}{V_1^{\gamma}} = \frac{P_1}{P_2}$$

Again,

$$V_1 + V_2 = V_0$$

So, $\frac{V_2}{V_1} = \left(\frac{P_1}{P_2}\right)^{\frac{1}{\gamma}}$
 $\Rightarrow \frac{V_0}{V_1} - 1 = \left(\frac{P_1}{P_2}\right)^{\frac{1}{\gamma}}$
 $\Rightarrow V_1 = \frac{V_0}{1 + \left(\frac{P_1}{P_2}\right)^{\frac{1}{\gamma}}}$

Therefore, $V_2 = \frac{V_0 \left(\frac{P_1}{P_2}\right)^{\frac{1}{\gamma}}}{1 + \left(\frac{P_1}{P_2}\right)^{\frac{1}{\gamma}}}$

The final volume of the left and the right side will be $\frac{V_0}{1 + \left(\frac{P_1}{P_2}\right)^{\frac{1}{\gamma}}}$ and $\frac{V_0 \left(\frac{P_1}{P_2}\right)^{\frac{1}{\gamma}}}{1 + \left(\frac{P_1}{P_2}\right)^{\frac{1}{\gamma}}}$

respectively.

(b) The heat given will be zero as the whole process is taking place in an adiabatic surrounding.

(c) So putting the above result in (1) we get,

$$P_{1}\left(\frac{V_{0}}{2}\right)^{\gamma} = P\left(\frac{V_{0}\left(\frac{P_{1}}{P_{2}}\right)^{\frac{1}{\gamma}}}{1+\left(\frac{P_{1}}{P_{2}}\right)^{\frac{1}{\gamma}}}\right)^{\gamma}$$
$$\Rightarrow P = P_{2} \frac{\left(1+\left(\frac{P_{1}}{P_{2}}\right)^{\frac{1}{\gamma}}\right)^{\gamma}}{2^{\gamma}} = \left(\frac{P_{1}\frac{1}{\gamma}+P_{2}\frac{1}{\gamma}}{2}\right)^{\gamma}$$

Thus the final common pressure of the gasses will be $\left(\frac{p_1 \frac{1}{Y} + p_2 \frac{1}{Y}}{2}\right)^{\gamma}$

Given:

An adiabatic cylindrical tube of cross-sectional area 1 cm^2 is closed at one end and fitted with a piston at the other end.

The tube contains 0.03g of an ideal gas at 1 atm pressure and at the temperature of the surrounding.

The length of the gas column is 40 cm.

The piston is suddenly pulled out to double the length of the column and the pressure of the gas falls to 0.355 atm.

The expansion process of the gas is adiabatic, so $P_1V_1^{\gamma} = P_2V_2^{\gamma}$ Where, subscript 1 and 2 represent the initial and the final state.

When the length is increased to double the volume is also doubled.

So,

$$1 \times V_1^{\gamma} = P_2 (2V_1)^{\gamma}$$

$$\Rightarrow 1 = 0.355(2)^{\gamma}$$

$$\Rightarrow (2)^{\gamma} = 2.82$$

$$\Rightarrow \gamma = 1.49$$

The speed of sound in gas at atmospheric pressure is given as

$$v = \sqrt{\frac{\gamma P}{\rho}}$$

where γ is the adiabatic constant, P is pressure and ρ is the volume density. The speed is

$$v = \sqrt{\frac{\gamma P}{\rho}}$$

$$v = \sqrt{\frac{1.49 \times 10^5}{\frac{m}{v}}}$$

$$v = \sqrt{\frac{\frac{1.49 \times 10^5}{0.03 \times 10^{-3}}}{\frac{0.03 \times 10^{-3}}{0.4 \times 10^{-4}}}} \approx 447 \frac{m}{s}$$

Given:

The speed of sound in hydrogen at 0° C is 1280 m s⁻¹.

The density of hydrogen at STP is 0.089 kg m⁻³.

At STP the pressure P is 1.013×10^5 Pa.

So the speed of sound in hydrogen $v = \sqrt{\frac{\gamma P}{\rho}}$ where ρ is the density of the gas.

So, putting data in, we get, $\gamma = \frac{1280^2 \times 0.089}{1.013 \times 10^5} = 1.44$

So,
$$C_P = \frac{\gamma R}{\gamma - 1} = \frac{1.44 \times 8.314}{0.44} = 27.21 J K^{-1} mol^{-1};$$

$$C_V = \frac{R}{\gamma - 1} = \frac{8.314}{0.44} = 18.89 \, J \, K^{-1} \, mol^{-1}$$

Answer.33

Given:

4.0 g of helium occupies 22400 cm^3 at STP.

The specific heat capacity of helium at constant pressure is 5.0 cal K^{-1} mol⁻¹ = 21J K^{-1} mol⁻¹

$$C_{P} = \frac{\gamma R}{\gamma - 1}$$
$$\Rightarrow 21\gamma - 21 = 8.314\gamma$$

$$\Rightarrow \gamma = \frac{21}{12.686} = 1.65$$

At STP the pressure P is 1.013×10^5 Pa.

The velocity of sound will be $v = \sqrt{\frac{\gamma P V}{m}}$ where P is the pressure of the gas, V is the volume and M is the mass of the gas.

Thus putting the values given, we get,

$$v = \sqrt{\frac{1.65 \times 1.013 \times 10^5 \times 22400 \times 10^{-6}}{0.004}} = 969.05 \frac{m}{s}$$

The speed of sound in helium is 969.05m/s.

Answer.34

Given:

An ideal gas having density 1.7×10^{-3} g cm⁻³ = 1.7 kg m⁻³ at a pressure 1.5×10^{5} Pa is filled in a Kundt tube.

When the gas is resonated at a frequency of 3.0 kHz, nodes are formed at a separation of 6.0 cm.

the node separation is given by $\frac{l}{2}$ which is 6.0cm. Therefore,

$$l = 12cm = 0.12m$$

the frequency of the sound (f) is 3kHz

thus velocity of sound = v = fl = 360 m/s

again,

 $v = \sqrt{\frac{\gamma P}{\rho}}$ where P is the pressure and ρ is the density of the gas.

Thus,
$$\gamma = \frac{360^2 \times 1.7 \times 10^{-3}}{1.5 \times 10^5} = 1.4688$$

$$C_V = \frac{R}{\gamma - 1} = \frac{8.3}{0.4688} = 17.7 \, J \, mol^{-1} K^{-1}$$

$$C_P = \frac{\gamma R}{\gamma - 1} = 1.4688 \times C_V = 1.4688 \times 17.07 = 26 J \ mol^{-1} K^{-1}$$

Given:

Standing waves of frequency 5.0 kHz are produced in a tube filled with oxygen 300 K. at

The separation between the consecutive nodes is 3.3 cm.

the node separation is given by $\frac{l}{2}$ which is 3.3cm. Therefore,

l = 6.6cm = 0.066m

the frequency of the sound is 5kHz

thus the velocity of sound will be $v = 5 \times 10^3 \times 0.066 = 330 \frac{m}{s}$

again,

 $v = \sqrt{\gamma \frac{RT}{M}}$ where R is the gas constant, T is the temperature of the gas in Kelvin scale and M is the molar mass of the gas.

The molar mass of oxygen is $32 \frac{g}{mole} = \frac{0.032 \, kg}{mole}$

Thus putting in the values in the above expression,

$$\gamma = \frac{330^2 \times 0.032}{8.314 \times 300} = 1.397$$
Again, $C_P = \frac{\gamma R}{\gamma - 1} = \frac{1.397 \times 8.314}{0.397} = 29.26 J \, mol^{-1} K^{-1}$

$$C_v = \frac{R}{\gamma - 1} = \frac{8.314}{0.397} = 20.94 J \, mol^{-1} K^{-1}$$

Therefore, the C_P and C_V for oxygen are 29.26 *J* $mol^{-1}K^{-1}$ and 20.94 *J* $mol^{-1}K^{-1}$ respectively.