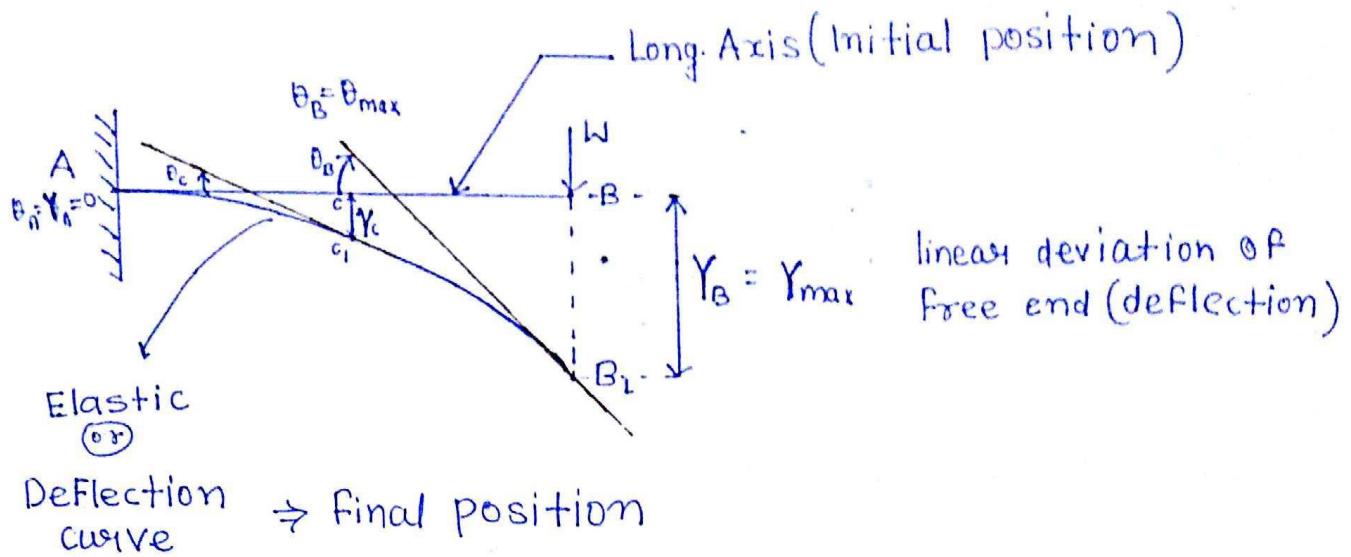


## Chapter -4 "Deflection of Beams"

Aim:-① Aim of this chapter is to derive expression for maximum slope & maximum deflection for various beams under different loading conditions.

- ② The maximum deflection and maximum slope expression are used in the design of beams and design of shaft based on rigidity criterion and in the determination of natural frequencies of shaft under transverse vibration
- ③ Deviation of axis of structural member under T.S.L is known as bending and deviation of axis of structural member under A.C.L is known as buckling.
- ④ Beam undergoes two deviation (linear and angular) while it is bending.
- ⑤ Linear deviation of longitudinal axis of beam ~~is~~ is known as deflection.
- ⑥ Angular deviation of L.A. of beam is known as slope
- ⑦ Slope and deflection due to shear force are neglected because they are smaller in magnitude in comparison to slope & deflection due to bending moment.



- Elastic Curve →
- (i) Circular arc  $[\because S.F. = 0 \text{ & } B.M. = \text{const.}]$
  - (ii) Straight line  $[S.F. = 0 \text{ & } B.M. = 0]$
  - (iii) Parabola  $[S.F. \neq 0 \text{ & } B.M. = \text{Variable}]$

\*  $\theta$  - Always measure in direction of Bending moment

$$(\text{Slope})_B = \tan \theta_B = \left( \frac{dY}{dx} \right)_{AB} = \frac{Y_B - Y_A}{X_B - X_A}$$

$$(\text{Slope})_B = (\theta)_B = \left( \frac{dY}{dx} \right)_{AB} \Rightarrow \tan \theta_B \approx \theta_B$$

$$(\theta)_C = \left( \frac{dY}{dx} \right)_{AC} = \frac{Y_C - Y_A}{X_C - X_A}$$

### Sign Convention

$$\theta_{ACW} = +ve$$

$Y_{\text{upward}} = +ve$

$$\theta_{CW} = -ve$$

$Y_{\text{downward}} = -ve$

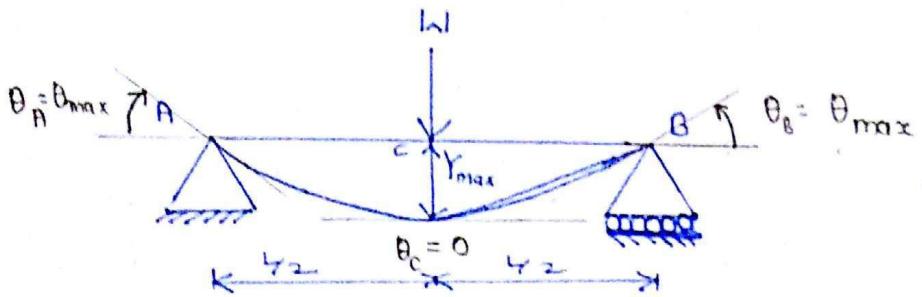
In Cantilever beam

At Fixed end

$$\theta = 0 \text{ & } Y = 0$$

At Free end

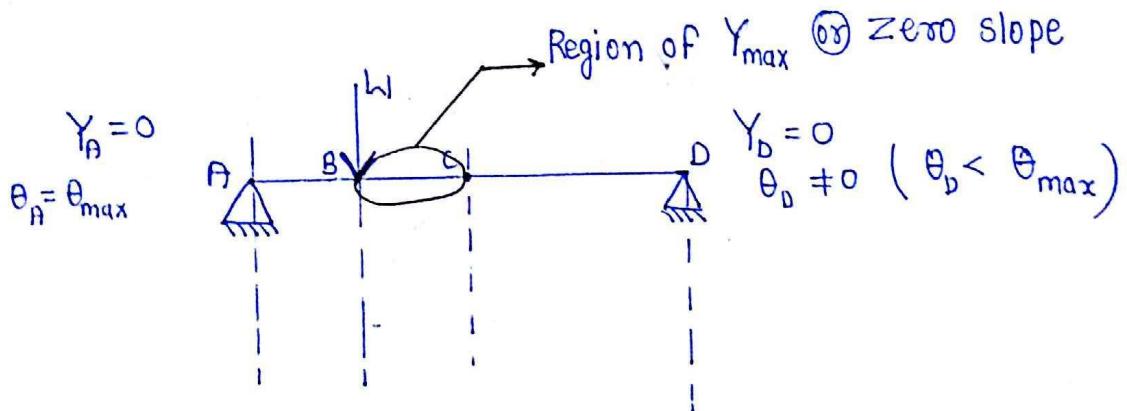
$$\theta = \theta_{\max} \text{ & } Y = Y_{\max}$$



In symmetrically loaded S.S.B.

- \* At mid span,  $\theta = 0$  &  $Y = Y_{\max}$
- \* At s.s. ends,  $Y = 0$  &  $\theta = \text{max.}$
- \* Slope at the ends are equal and unlike

\* Under any loading conditions In S.S.B., Slope is zero at  $x = c$  where deflection is maximum.



In unsymmetrically loaded S.S.B.

- (1) \* Deflection is maximum in the region between point of application of load & mid span.
- \* Slope at the support are unequal & unlike.

To determine  $Y_{\max}$ .

$$1. \quad \Theta_{x-x} = \text{_____} = 0 \Rightarrow x = x^*$$

$$\textcircled{O} \quad \frac{d}{dx}(Y_{x-x}) = 0$$

$$2. \quad Y_{\max} = (Y_{x-x})_{x=x^*} =$$

\*  $\Theta_{\max}$  nearby the applied load.

Double Integration Method:-

$$\boxed{\frac{1}{R_{x-x}} = \frac{d^2 Y_{x-x}}{dx^2}}$$

where  $R_{x-x}$  = Radius of curvature of  
Elastic Curve at  $x-x$

$$R_{x-x} = \boxed{\left( \frac{EI_{N.A.}}{M} \right)_{x-x}}$$

$$\boxed{M_{x-x} = (EI_{N.A.})_{x-x} \frac{d^2 Y_{x-x}}{dx^2}}$$

$\Rightarrow$  Governing diff. equation  
in term of B.M. slope &  
deflection at  $x-S_C$   $x-x$

$M_{x-x}$  = B.M. at  $x-S_C$   $x-x$

$Y_{x-x}$  = def<sup>n</sup> at  $x-S_C$   $x-x$

$\frac{dY_{x-x}}{dx}$  = slope at  $x-S_C$   $x-x$

$$\frac{dY_{x-x}}{dx} = \tan \theta$$

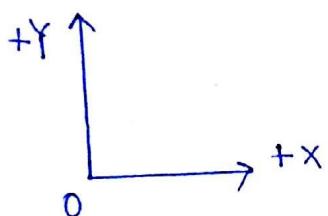
for small angular derivations

$$\tan \theta \approx \theta$$

Hence, slope at  $x - S_C$   $\frac{dY_{x-x}}{dx} = \theta$

$(EI)_{x-x}$  = Flexible Rigidity of  $x - S_C$   $x - x$

Sign Convention:-



(a) Bending moment

Sagging bending = +Ve

Hogging bending = -Ve

(b) Deflection-

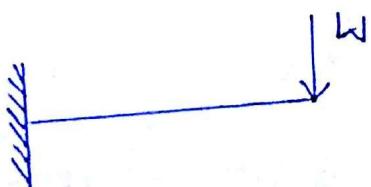
$$Y_{\text{upward}} = +\text{Ve}$$

$$Y_{\text{downward}} = -\text{Ve}$$

(c) Slope ( $\theta$ )

$$\theta_{\text{ACW}} = +\text{Ve}$$

$$\theta_{\text{CW}} = -\text{Ve}$$



Steps:- 1.  $(BM)_{x-x} = M_{x-x}$

2.  $(EI_{\text{N.A.}})_{x-x} \left[ \frac{d^2 Y_{x-x}}{dx_{x-x}^2} \right] = M_{x-x}$

3.  $E I_{\text{N.A.}} \left[ \frac{dY_{x-x}}{dx_{x-x}} \otimes \theta_{x-x} \right] = \int M_{x-x} + C_L$

det. by using B.C. Cons  
slope

$$4. (EI_{N.A.}) [Y_{x-x}] = \iint M_{x-x} + C_1 x + C_2$$

↓

Det by using B.C.  
of deflection

In presence of concentrated Moments ( $M$ )

$$\theta_{max} = \frac{1}{C_1} \left[ \frac{ML}{EI_{N.A.}} \right] ; Y_{max} = \frac{1}{C_2} \left[ \frac{ML^2}{EI_{N.A.}} \right]$$

In presence of concentrated point load ( $w$ )

$$\theta_{max} = \frac{1}{C_1} \left[ \frac{WL^2}{EI_{N.A.}} \right] ; Y_{max} = \frac{1}{C_2} \left[ \frac{WL^3}{EI_{N.A.}} \right]$$

In presence of D.L. ( $w$ )

$$\theta_{max} = \frac{1}{C_1} \left[ \frac{wL^3}{EI_{N.A.}} \right] ; Y_{max} = \frac{1}{C_2} \left[ \frac{wL^4}{EI_{N.A.}} \right]$$

In case of UDL,  $w = \frac{W}{L}$

In case of UVL,  $w = \frac{2W}{L}$

$$\frac{Y_{max}}{\theta_{max}} = \left[ \frac{C_1}{C_2} \right] L$$

\* Deflection is always one order higher than slope

\*  $[\theta_{max} \& Y_{max}] \propto \left( \frac{1}{EI_{N.A.}} \right)$

Inversely proportion to  
Flexural rigidity ( $EI$ )

## Stiffness

$$* \quad k = \frac{\omega}{Y_{max}} = C_2 \left[ \frac{EI_{N.A.}}{L^3} \right]$$

\* Stiffness

$$k \propto EI_{N.A.}$$

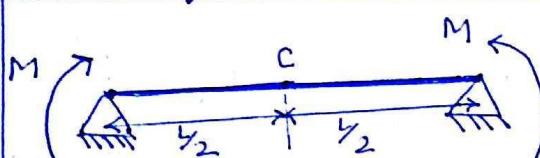
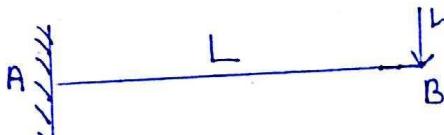
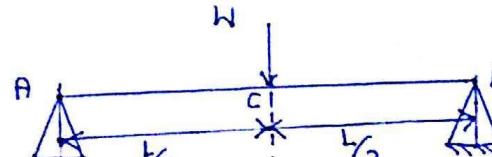
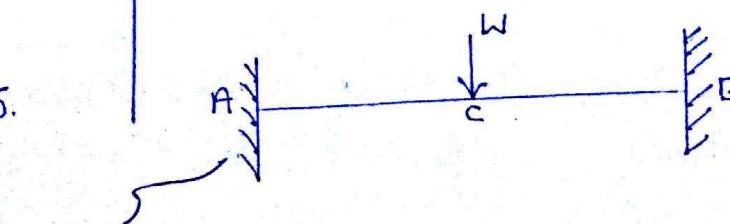
$$k_{\text{axial}} = \frac{AE}{L}$$

$$k_{\text{torsional}} = \frac{GJ}{L}$$

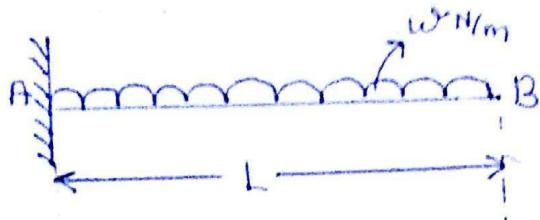
$$* \quad EI_{N.A.} (\uparrow) \Rightarrow k_{\text{beam}} (\uparrow)$$

$$\Rightarrow Y_{max} (\downarrow)$$

$$\Rightarrow \theta_{max} (\downarrow)$$

S. No.	$(EI_{N.A.} = \text{const.})$ Type of Beam	For $\theta_{max}$		For $Y_{max}$	
		$C_1$	$C_2$	$C_1$	$C_2$
1.		1	2		
2.		2	8		
3.		2	3		
4.		16	48		
5.		-	-		192

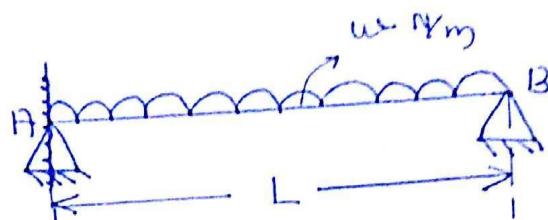
6.



6

8

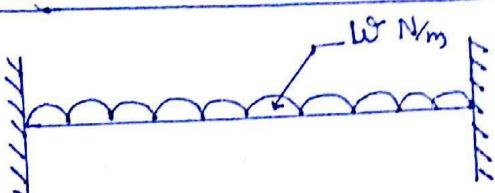
7.



24

$$\frac{384}{5}$$

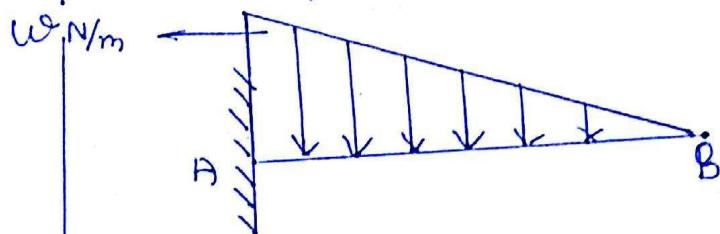
8.



—

$$384$$

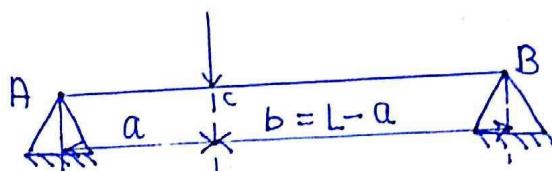
9.



24

$$30$$

10.



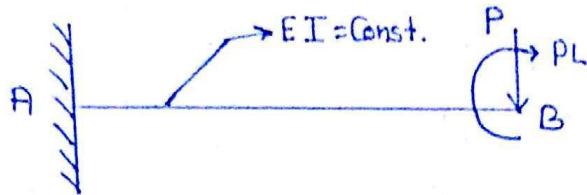
$$\theta_c = \frac{w b}{3 E I_{N.A.} L} [a^2 - ab]$$

$$Y_c = \frac{w a^2 b^2}{3 E I_{N.A.} L}$$

	M	w	UDL	UVL
C.V.	1 2	2 3	6 8	24 30
SSB	2 8	16 48	24 $\frac{384}{5}$	-

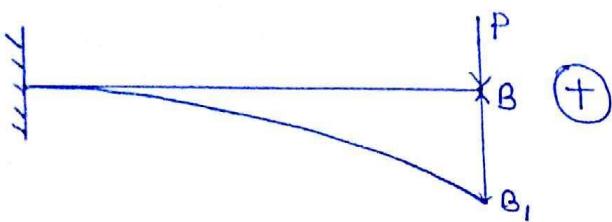
- 192      - 384

## Question

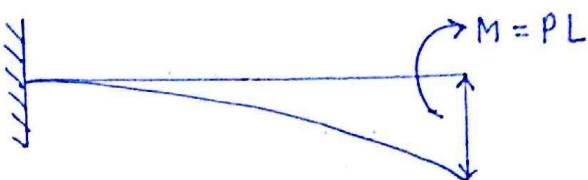


Using principle of Superposition

(1)



(2)



$$\theta_1 = \frac{1}{c_1} \left[ \frac{PL^2}{EI} \right] = \frac{1}{2} \left[ \frac{PL^2}{EI} \right] \quad (2)$$

$$Y_1 = \frac{1}{c_2} \left[ \frac{PL^3}{EI} \right] = \frac{1}{3} \left[ \frac{PL^3}{EI} \right] \quad (\downarrow) \quad (1)$$

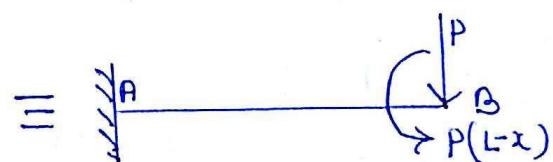
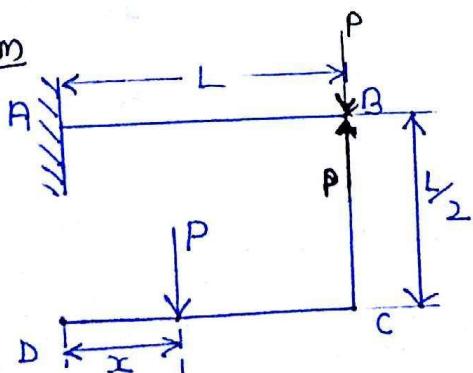
$$\theta_2 = \left[ \frac{1}{c_1} \right] \left[ \frac{ML}{EI} \right] = \frac{1}{L} \left[ \frac{(PL)L}{EI} \right]$$

$$Y_2 = \frac{1}{c_2} \left[ \frac{ML^2}{EI} \right] = \frac{1}{2} \left[ \frac{PL^3}{EI} \right] \quad (\downarrow)$$

$$\theta_B = \theta_1 + \theta_2 = \frac{3}{2} \frac{PL^2}{EI} \quad (2)$$

$$Y_B = Y_1 + Y_2 = \frac{5}{6} \frac{PL^3}{EI} \quad (\downarrow)$$

## Question

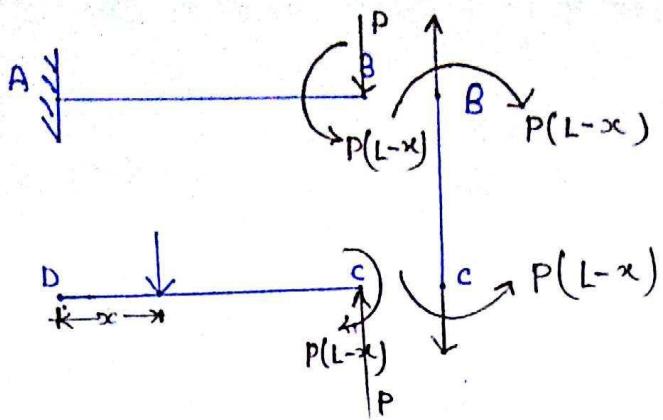


$$Y_B = Y_1 + Y_2 = \left( -\frac{1}{3} \right) \frac{PL^3}{EI} + \frac{ML^2}{2EI} = 0$$

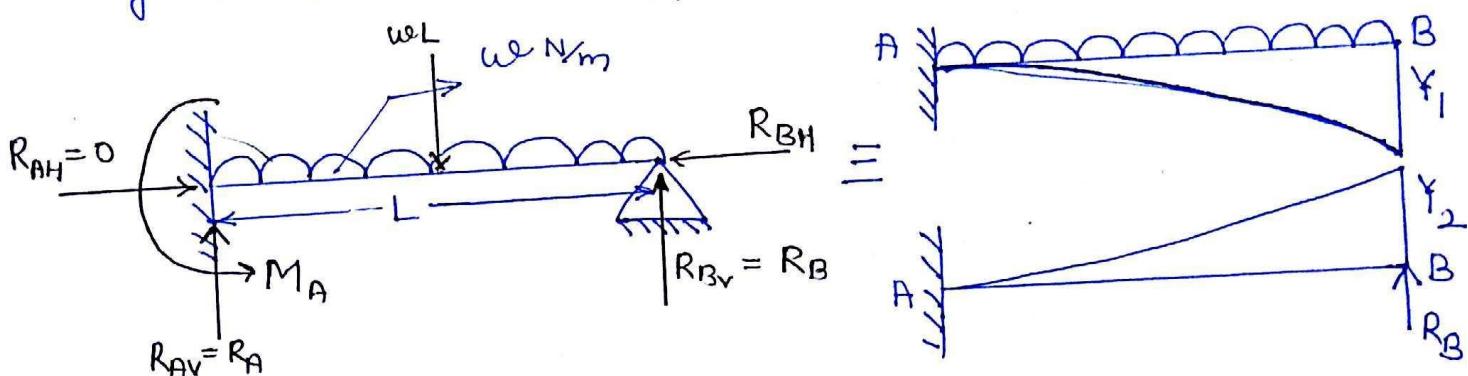
$$-\frac{PL^3}{3EI} + \frac{P(L-x)L^2}{2EI} = 0$$

Determine the value of 'x' if  $Y_B = 0$

$$\frac{PL}{3} = L-x \Rightarrow x = \frac{L}{3}$$



Question For the propped cantilever beam as shown in Fig determine the support reaction.



$$\sum v = 0 \Rightarrow R_A + R_B = wL \quad \text{--- (1)}$$

$$\sum M_A = 0 \Rightarrow -M_A + wL\left(\frac{L}{2}\right) - R_B(L) = 0 \quad \text{--- (2)}$$

$$Y_B = Y_1 + Y_2 = 0$$

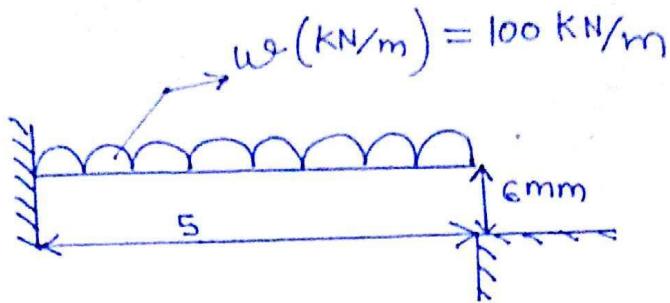
$$Y_B = -\left(\frac{1}{8}\right) \frac{wL^4}{EI} + \frac{1}{3} \left(\frac{R_B L^3}{EI}\right) = 0$$

$$R_B = \frac{3}{8} (wL) \uparrow$$

$$\text{From eqn (1), } R_A = \frac{5}{8} wL \uparrow$$

$$\text{From eqn (2)} \quad M_A = \frac{wL^2}{8}$$

Question: 12  
workbook



No use of  
extra data

Assuming  $R_B = 0$  [ $\because Y_B \leq 6 \text{ mm}$ ]

Cantilever beam

$$Y_B = \frac{w L^4}{8 E I_{\text{N.A.}}}$$

$$Y_B = \frac{100 \times (5)^4}{8 \times 781250 \times} = 10 \text{ mm}$$

$R_B \neq 0$  [ $\because Y_B > 6 \text{ mm}$ ]  $\Rightarrow$  st. indt. beam

$$Y_B = Y_1 + Y_B = 0$$

$$-\frac{w L^4}{8 E I} + \frac{R_B L^3}{3 E I} = -6 \text{ mm}$$

$$R_B = 75 \text{ kN}$$

$$\sum V = R_A + R_B = 500 \text{ kN}$$

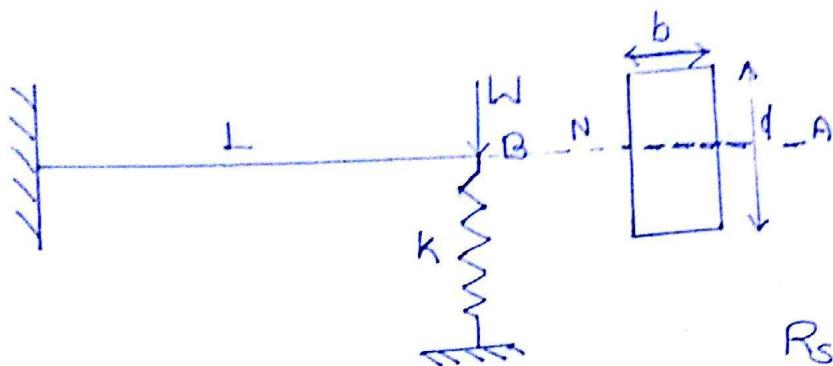
$$R_A = 500 - 75$$

$$R_A = 425 \text{ kN } (\uparrow)$$

Question For the beam as shown in Fig determine  
Data:-

(i) spring reaction

(ii) Deflection of spring.



$$W = 2.0 \text{ kN}$$

$$k = 500 \text{ N/mm}$$

$$E = 200 \text{ GPa}$$

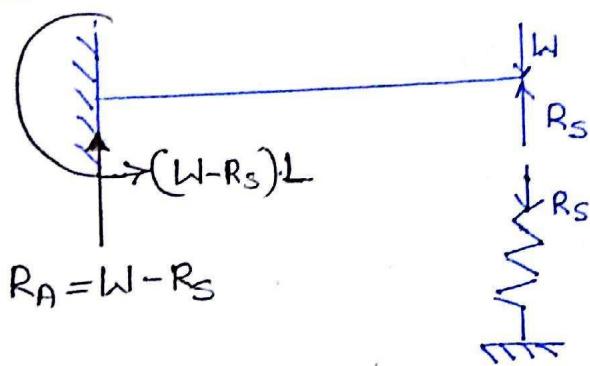
$$L = 2 \text{ m}$$

$$b = 50 \text{ mm}$$

$$d = 100 \text{ mm}$$

$$R_s = kx = k\delta_s$$

SOL<sup>n</sup>



$$R_A = W - R_s$$

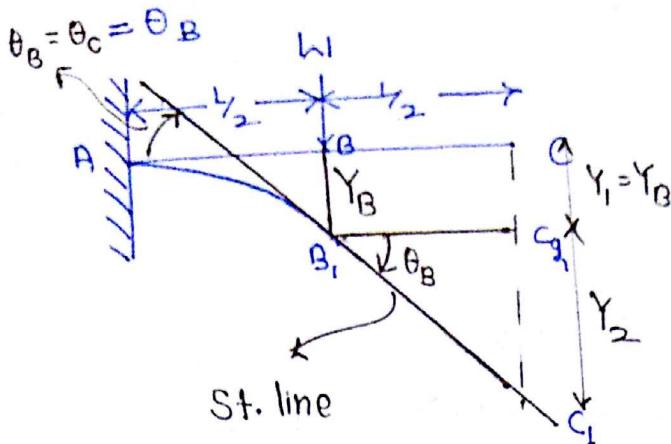
$$Y_B = Y_1 + Y_2 = -\delta_s$$

$$-\frac{WL^3}{3EI} + \frac{R_s L^3}{3EI} = -\frac{R_s}{k}$$

$$R_s = \frac{kWL^3}{kL^3 + 3EI_{N.A.}}$$

$$\delta_s = \frac{WL^3}{kL^3 + 3EI_{N.A.}}$$

Question Find deflection at C



$\Delta B_1 C_2 C_1$

$$\tan \theta_B = \frac{Y_2}{L}$$

$$Y_2 = \theta_B \frac{L}{2} \quad \therefore \tan \theta_B \approx \theta_B$$

$$Y_C = Y_{\max} = Y_1 + Y_2 = Y_B + \theta_B (L/2)$$

$$\theta_B = \frac{1}{2} \frac{w(L/2)^2}{EI} = \frac{wL^2}{8EI}$$

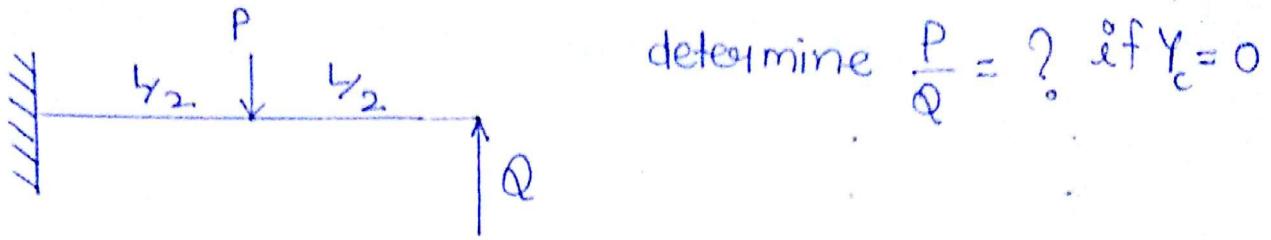
$$Y_B = \frac{1}{3} \frac{w(L/2)^3}{EI} = \frac{wL^3}{24EI}$$

$$\theta_{\max} = \theta_C = \theta_B = \frac{wL^2}{8EI} \quad (?)$$

$$Y_{\max} = \frac{wL^3}{24EI} + \frac{wL^2}{8EI} (L/2)$$

$$Y_{\max} = \frac{5}{48} \left[ \frac{wL^3}{EI_{N.A.}} \right] \quad (?)$$

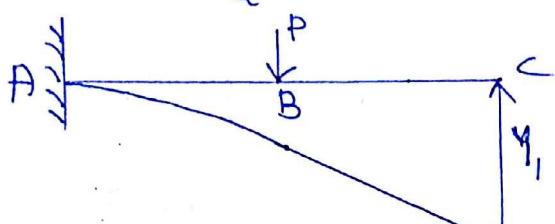
## Question



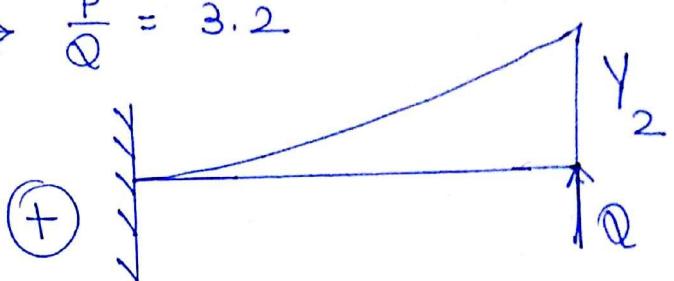
SOLN

$$\frac{5}{48} \frac{PL^3}{EI} = \frac{QL^3}{3EI}$$

$$\frac{P}{Q} = \frac{48}{15} \Rightarrow \frac{P}{Q} = 3.2$$

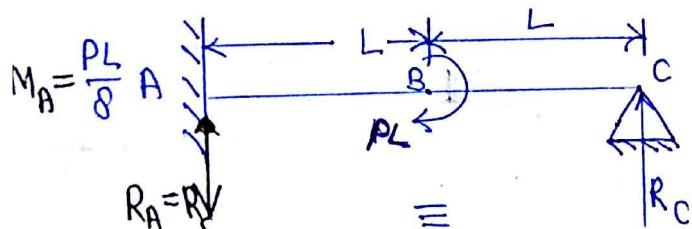


$$Y_1 = Y_B + \theta_B (Y_2)$$

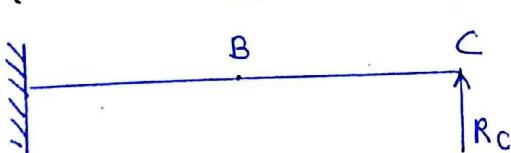


$$Y_1 = \frac{QL^3}{3EI}$$

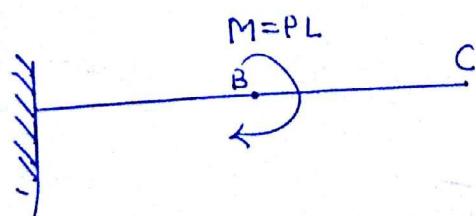
Ans



$$\text{Find } R_C = ?$$



$$Y_1 = \frac{R_c(2L)^3}{3EI} = \frac{8R_c L^3}{3EI}$$



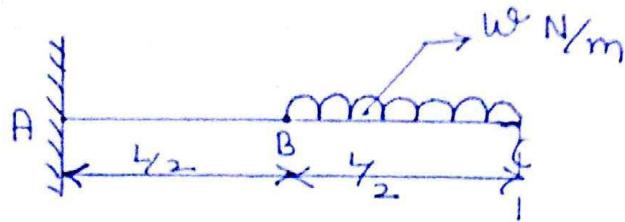
$$Y_2 = Y_B + \theta_B L = \frac{ML^2}{2EI} + \frac{ML}{EI}(L)$$

$$Y_2 = \frac{PL^3}{2EI} + \frac{PL^3}{EI} = \frac{3}{2} \frac{PL^3}{EI}$$

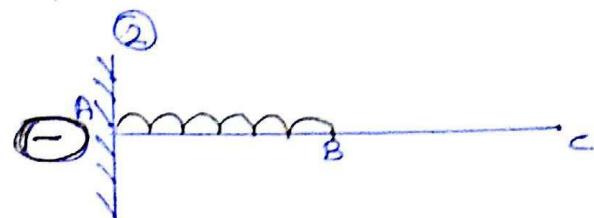
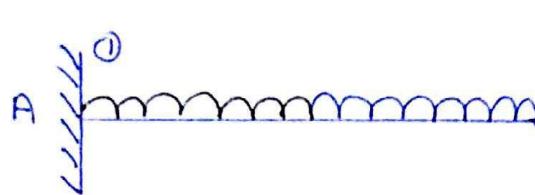
$$Y_C = Y_1 + Y_2 = 0 \Rightarrow \frac{8R_c L^3}{3EI} + \left( -\frac{3}{2} \frac{PL^3}{EI} \right) \Rightarrow R_c = \frac{9}{16} P (\uparrow)$$

$$\Rightarrow PL - \frac{9}{16} P(L \times 2) = M_A = \frac{PL}{8} \Leftrightarrow R_A = \frac{9}{16} P (\downarrow)$$

Ques Find slope and deflection at C i.e.  $\theta_c = ?$   
 $\gamma_c = ?$



Soln



$$\theta_1 = \frac{wL^3}{6EI} \text{ (2)} ; \gamma_1 = \frac{wL^4}{8EI} \text{ (↓)} \quad \theta_2 = \frac{w(L/2)^3}{6EI} = \frac{wL^3}{48EI} \text{ (↔)}$$

$$\gamma_1 = \frac{wL^4}{8EI} \text{ (↓)}$$

$$\boxed{\gamma_2 = \gamma_B + \theta_B(L/2)}$$

$$\gamma_2 = \frac{w(L/2)^2}{8EI} + \frac{wL^3}{48EI} \left(\frac{L}{2}\right)$$

$$\gamma_c = \gamma_1 - \gamma_2$$

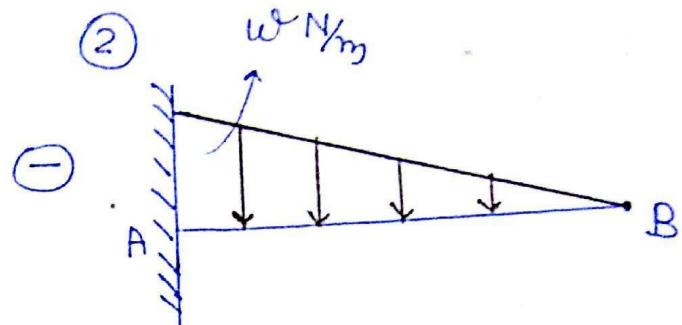
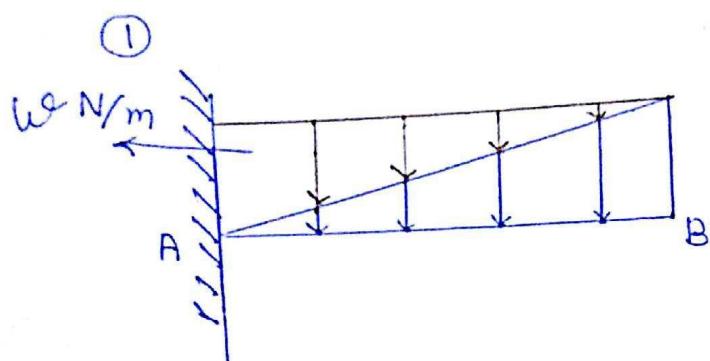
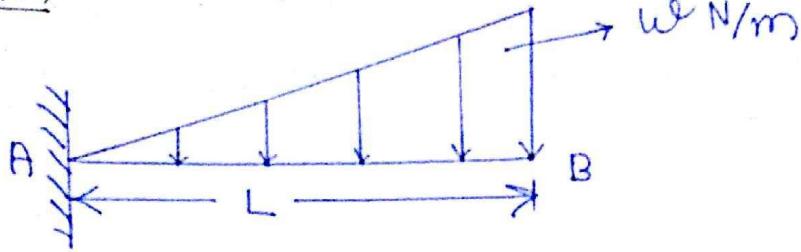
$$\gamma_2 = \frac{7}{384} \frac{wL^4}{EI}$$

$$\gamma_c = \frac{wL^4}{EI} \left( \frac{1}{8} - \frac{7}{384} \right) = \cancel{\left( \frac{41}{384} \right)} \left( \frac{wL^4}{EI} \right) \text{ (↓)}$$

$$\theta_c = \theta_1 - \theta_2$$

$$\theta_c = \frac{wL^3}{EI} \left( \frac{1}{6} - \frac{1}{48} \right) = \frac{7}{48} \frac{wL^3}{EI} \text{ (↔)}$$

Question



$$(Y_B)_I = \frac{wL^4}{8EI} (\downarrow)$$

$$(Y_B)_{II} = \frac{wL^4}{30EI} (\downarrow)$$

$$Y_B = (Y_B)_I - (Y_B)_{II}$$

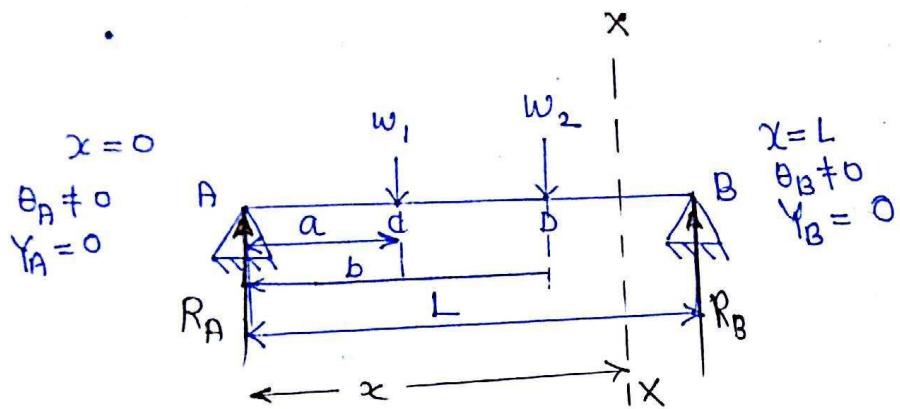
$$Y_B = \frac{wL^4}{EI} \left( \frac{1}{8} - \frac{1}{30} \right) = \frac{11}{120} \frac{wL^4}{EI}$$

$$\theta_B = \frac{wL^3}{6EI} - \frac{wL^3}{24EI} = \frac{wL^3}{8EI}$$

# Macaulay's Method : —

(Modified double Integration method)

$$M_{x-x} = \left( EI_{N.A.} \right)_{x-x} \frac{d^2 Y_{x-x}}{dx^2}$$



✳ Consider a  $x$ -sec in last segment and write BM

$$M_{x-x} = R_A(x) - w_1(x-a) - w_2(x-b)$$

✳ For any  $x$  value if term is -ve then discard it

e.g. For  $x = a_2$

$$M_{x-x} = R_A(a_2) \quad \because (a_2 < a < b)$$

Neglect two

Now

$$EI_{N.A.} \frac{d^2 Y_{x-x}}{dx^2} = R_A x - w_1(x-a) - w_2(x-b)$$

$$EI_{N.A.} \left[ \frac{dy}{dx^2} \text{ } \textcircled{m} \text{ } \theta_{x-x} \right] = R_A \frac{x^2}{2} - \frac{w_1(x-a)^2}{2} + \frac{w_2(x-b)^2}{2} + C_1$$

in macaulay's method

$$\int (x-a) dx = \frac{x^2}{2} - ax \quad x$$

$$\int (x-a)^2 dx = \frac{(x-a)^2}{2} \quad \checkmark$$

$$EI_{N.A.} [Y_{x-x}] = R_A \frac{x^3}{6} - W_1 \frac{(x-a)^3}{6} - \frac{W_2(x-b)^3}{6} + C_1 x + C_2$$

$$\Rightarrow x=0 \Rightarrow Y=0$$

$$EI_{N.A.}[0] = 0 - 0 - 0 + 0 + C_2$$

at  $x=0$   $(x-a)$  &  $(x-b)$  become -ve so  
discard them

$$\Rightarrow x=L \Rightarrow Y=0$$

$$EI_{N.A.}[0] = \frac{R_A L^3}{6} - \frac{W_1(L-a)^3}{6} - \frac{W_2(L-b)^3}{6} + C_1 L$$

$$C_1 = \beta \text{ Const.}$$

$\Rightarrow$  Equation for Slope

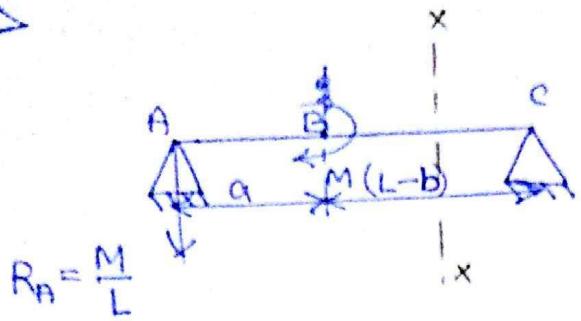
$$EI_{N.A.} \left[ \frac{dY_{x-x}}{dx} \right] = \frac{R_A x^2}{2} - \frac{W_1(x-a)^2}{2} - \frac{W_2(x-b)^2}{2} + \beta$$

$\Rightarrow$  Equation for deflection

$$EI_{N.A.} [Y_{x-x}] = \frac{R_A x^3}{6} - \frac{W_1(x-a)^3}{6} - \frac{W_2(x-b)^3}{6} + \beta x$$

$\boxed{\begin{array}{l} \text{if } x < a \text{ discard } (x-a) \text{ & } (x-b) \\ a < x < b \text{ discard } (x-b) \end{array}}$

W



$$\Rightarrow M_{x-x} = R_A x - M(x-a) \stackrel{0}{=}$$

$$R_A = \frac{M}{L}$$

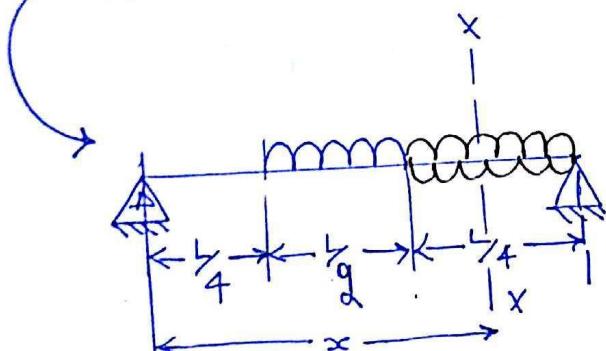
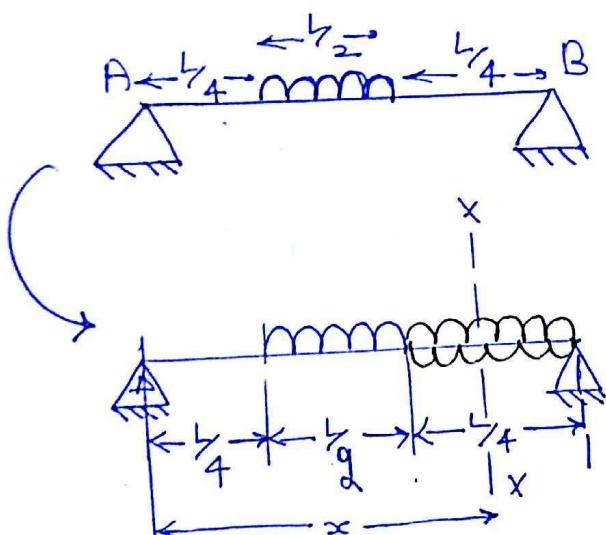
$$M_{x-x} = R_A x - M \quad x$$

~~R<sub>A</sub> = M/L~~

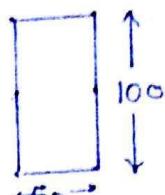
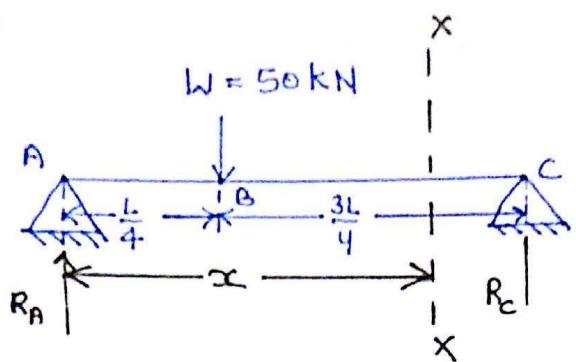
$$\boxed{M_{x-x} = R_A x - M(x-a)}$$

✓

⇒



Question



Determine

$$(a) \theta_A \text{ & } \theta_C$$

$$(b) Y_B \text{ & } Y_{max}$$

$$EI = 2m; E = 200 \text{ GPa}$$

$$M_{x-x} = \frac{3}{4}W(x) - W\left(x - \frac{L}{4}\right)$$

$$R_A = \frac{W \times \frac{3L}{4}}{L} = \frac{3W}{4}$$

$$EI \frac{d^2y}{dx^2} = \frac{3}{4}Wx - W\left(x - \frac{L}{4}\right)$$

$$EI \frac{dy}{dx} = \frac{3}{4}W \frac{x^2}{2} - \frac{|W|}{2} \left(x - \frac{L}{4}\right)^2 + C_1 \quad -①$$

$$EI[Y] = \frac{3}{24}Wx^3 - \frac{|W|}{6} \left(x - \frac{L}{4}\right)^3 + C_1 x + C_2 \quad -②$$

$$\begin{aligned} x=0 & \quad 0 = 0 - 0 + 0 + C_2 \\ Y=0 & \quad C_2 = 0 \end{aligned}$$

$$\begin{aligned} x=L & \Rightarrow 0 - \frac{3}{18}|W|L^3 - \frac{27}{64}|W|L^3 + C_1 L \\ Y=0 & \end{aligned}$$

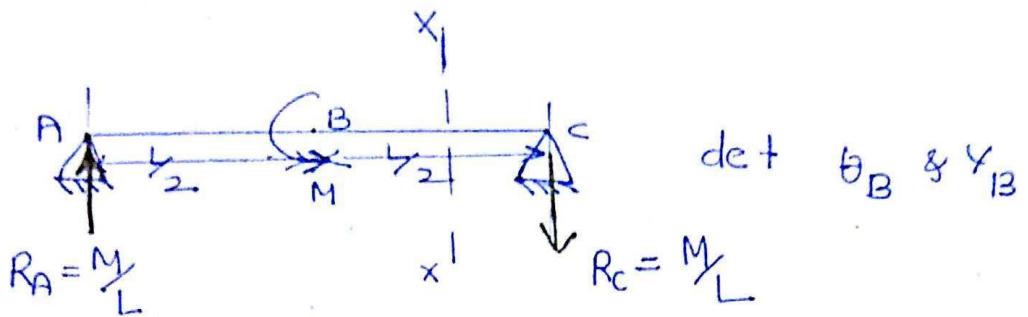
$$\begin{aligned} C_1 &= WL^2 \left( \frac{1}{18} + \frac{27}{64} \right) \\ W L^2 \left( \frac{8+27}{64} \right) & \end{aligned}$$

$$\psi = \frac{\theta}{EI} = \frac{1}{EI} \left[ \frac{3}{8}Wx^3 - \frac{|W|}{6} \left(x - \frac{L}{4}\right)^3 + \frac{35}{64}WL^2x \right]$$

$$\theta = \frac{1}{EI} \left[ \frac{3Wx^2}{8} - \frac{|W|}{2} \left(x - \frac{L}{4}\right)^2 + \frac{35}{64}WL^2 \right]$$

$$C_1 = \frac{35}{64}WL^2$$

## Question



$$M_{x-x} = \frac{M}{L}x - M(x - \frac{L}{2})^0$$

$$EI \frac{dy}{dx} = \frac{Mx^2}{2L} - M(x - \frac{L}{2}) + c_1$$

$$EI[y] = \frac{Mx^3}{6L} - \frac{M}{2}(x - \frac{L}{2})^2 + c_1 x + c_2$$

$$\begin{aligned} x=0 \\ y=0 \end{aligned} \Rightarrow \begin{aligned} c_2 &= 0 \\ x=L \\ y=0 \end{aligned} \Rightarrow 0 = \frac{ML^3}{6L} - \frac{M}{2}\left(\frac{L}{2}\right)^2 + c_1 L = 0$$

$$\text{EQ} \quad \frac{d^2y}{dx^2} = \frac{1}{EI} \left[ \frac{Mx^2}{2L} - M(x - \frac{L}{2}) - \frac{ML}{24} \right]$$

$$c_1 = ML^2 \left( \frac{1}{8} - \frac{1}{6} \right)$$

$$c_1 = -\frac{ML^2}{24}$$

$$\theta_B \Big|_{x=\frac{L}{2}} = \frac{1}{EI} \left[ \frac{M}{2L} \frac{L^2}{4} - \frac{ML}{24} \right] = \frac{ML}{EI} \left( \frac{1}{8} - \frac{1}{24} \right) = \frac{ML}{EI} \frac{3-1}{24}$$

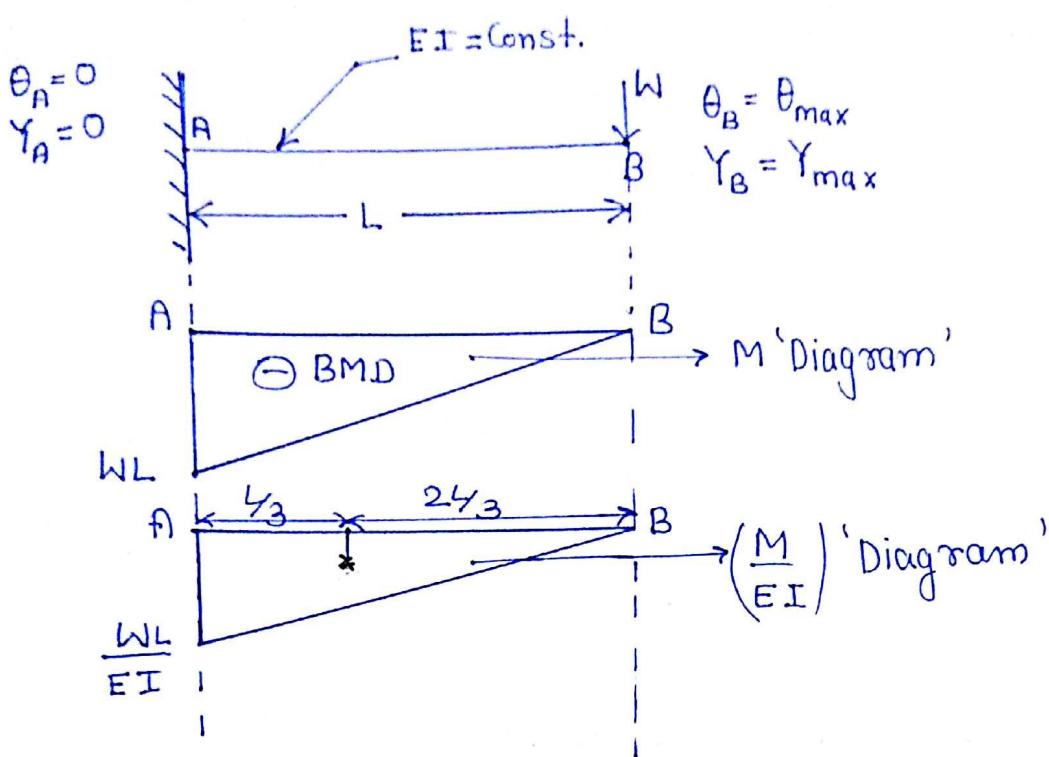
$$\theta_B = -\frac{ML}{12EI} \quad \textcircled{2} \quad \frac{ML}{12EI} \text{ (cw)}$$

$$EIy = \frac{1}{EI} \left[ \frac{Mx^3}{6L} - \frac{M}{2}(x - \frac{L}{2})^2 + \frac{ML}{24}x \right]$$

$$y \Big|_{x=\frac{L}{2}} = \frac{ML^2}{EI} \left[ \frac{1}{48} - \right]$$

## Moment Area Method :-

(i) Draw B.M.D.



### Mohr's Theorems

First theorem:- Difference of two slope between two x-s/c (one is zero slope and other one is equal to area of  $\frac{M}{EI}$  diagram) is equal to area of  $\frac{M}{EI}$  diagram

$$\theta_B - \theta_A = \text{Area of } \frac{M}{EI} \text{ diagram between B & A}$$

$$\theta_{\max} - 0 = \frac{1}{2} \times L \times \left( \frac{-WL}{EI} \right)$$

$$\theta_{\max} = -\frac{WL^2}{2EI} \quad \text{or} \quad \frac{WL^2}{2EI} \quad (\rightarrow) \text{ clockwise}$$

Second theorem:- Diff. of deflection between two x-s/c (simplified) moment of area of  $\frac{M}{EI}$  Diagram.

↳ is valid when point of zero slope is known -

$Y_B - Y_A = \text{moment of Area of } \frac{M}{EI} \text{ diag. bet^n B & A}$

②  $Y_B - Y_A = A\bar{x} \text{ of } \frac{M}{EI} \text{ diag. bet^n B & A}$

$$Y_{\max} - 0 = -\frac{WL^2}{3EI} \times \frac{2L}{3} \quad \bar{x} = \text{From non zero slope}$$

$$Y_{\max} = -\frac{WL^3}{3EI} \text{ or } \frac{WL^3}{3EI} (\downarrow)$$

If  $EI_{N.A.} = \text{constant}$

$$\text{I theorem} \Rightarrow \theta_B - \theta_A = \frac{1}{EI_{N.A.}} [\text{Area of BMD bet^n B & A}]$$

$$\text{II theorem} \Rightarrow Y_B - Y_A = \frac{1}{EI_{N.A.}} [A\bar{x} \text{ of BMD bet^n B & A}]$$

Actual II theorem:  $\Rightarrow$  when point of zero slope unknown,

$$(Y_B - Y_A) + (x_A \theta_A - x_B \theta_B) = A\bar{x} \text{ of } \frac{M}{EI} \text{ dia. bet^n B & A}$$

In previous example  $x_A = L ; \theta_A = 0$

$x_B = 0 ; \theta_B \neq 0$

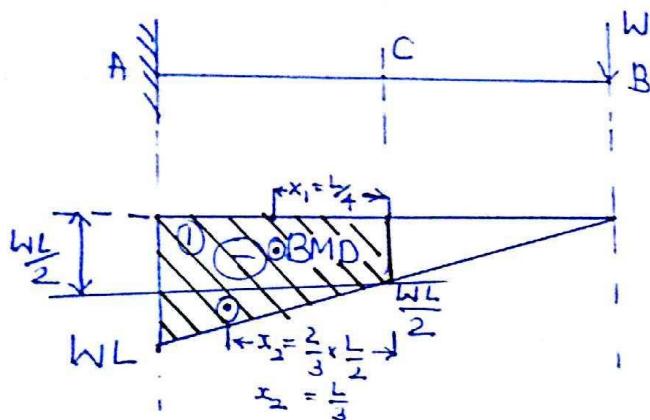
$$\text{So } Y_B - Y_A = A\bar{x} \text{ of } \frac{M}{EI} \text{ dia. bet^n B & A}$$

Conclusion:-

- (i) Simplified second theorem should be used when point of zero slope is known (for all cantilever beam and SSB under symmetric loading conditions).
- (ii) When simplified II theorem is used two  $x-s/c$  should be selected in such a way that one should be ~~be~~ a zero slope  $x-s/c$  and another one is non-zero slope  $x-s/c$ .
- (iii) Non-zero  $x-s/c$  should be selected in such a way that it should be  $x-s/c$  where slope and deflection are to be determined.
- (iv) Non-zero slope  $x-s/c$  should be considered as origin and zero slope  $x-s/c$  is considered as reference  $x-s/c$ .
- (v)  $\bar{x}$  should be measured from origin (i.e. non-zero slope  $x-s/c$ )
- (vi) Area of BMD should be used for the calculation of slope and deflection of beam whose flexure rigidity is constant.
- (vii) Area of  $M/EI$  diagram should be used for the calculation of slope & deflection of beam whose flexure rigidity is variable.
- (viii) Actual Second theorem should be used when zero slope  $x-s/c$  is unknown (i.e. for SSB under unsymmetric loading)

Ques +

Determine slope & deflection at a  $x-s/c$  located at a distance  $y_2$  from fixed end of a cantilever beam when it is loaded by concentrated point load at the free end in the downward dir<sup>n</sup>



Assume  $EI$  Const.

$$\theta_A = 0$$

$$Y_A = 0$$

Ref.  $x-s/c \rightarrow$  zero slope  $x-s/c$  (A)

Origin  $\rightarrow$  Non zero slope  $x-s_c$  (C)

$$A_1 = -\frac{WL}{2} \times \frac{L}{2}$$

$$A_2 = -\frac{1}{2} \times \frac{WL}{2} \times \frac{L}{2}$$


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$$\theta_C - \theta_A = -\frac{1}{2} \times \left(\frac{L}{2}\right) \times \left(\frac{3}{2} \frac{WL}{EI}\right)$$

$$\theta_C = -\frac{3}{8} WL^2 \quad \text{(or)} \quad \frac{3}{8} \frac{WL^2}{EI} \quad (\text{?})$$

$$\text{or } \theta_C - 0 = \frac{1}{EI} (A_1 + A_2)$$

$$= \frac{-WL^2}{EI} \left[ \frac{1}{4} + \frac{1}{8} \right]$$

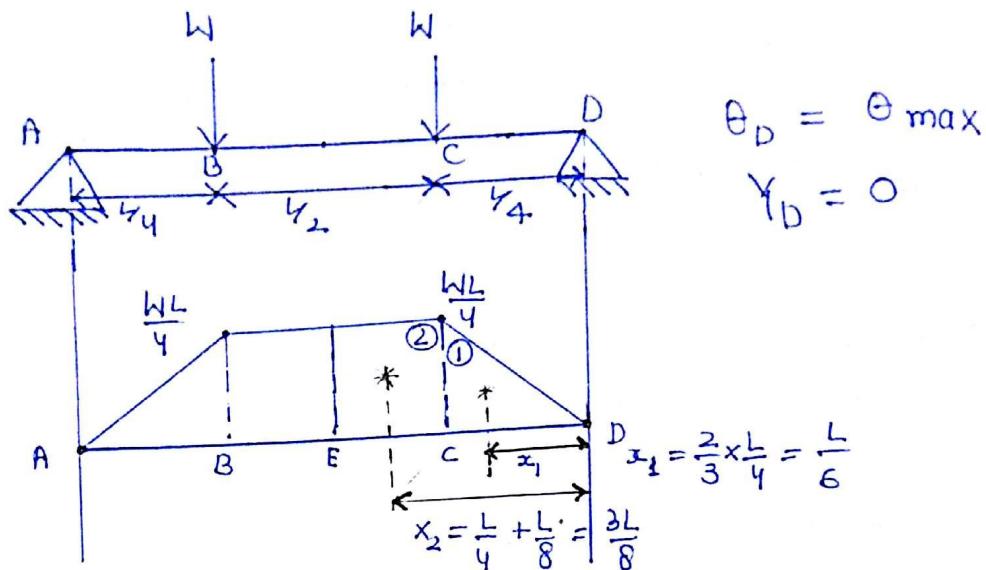
$$\theta_C = \frac{3}{8} WL^2 \quad (\text{?})$$

$$Y_C - Y_A = \frac{1}{EI_{N.A.}} \left[ A_1 \bar{x}_1 + A_2 \bar{x}_2 \right]$$

$$Y_C = \frac{1}{EI} \left[ \left( -\frac{WL^2}{4} \times \frac{L}{4} \right) + \left( -\frac{WL^2}{8} \times \frac{L}{3} \right) \right]$$

$$Y_C = -\frac{5}{48} \frac{WL^3}{EI_{N.A.}} \quad \text{(or)} \quad \frac{5 WL^3}{48 EI_{N.A.}} \quad (\downarrow)$$

Prob. For SSB as shown in fig determine slope at the support and max. deflection.



Ref  $x-s/C \rightarrow$  zero slope  $x-s/C (E)$

origin  $\rightarrow$  non zero slope  $x-s/C (D)$

$$\theta_D - \theta_E = \frac{1}{EI} (A_1 + A_2)$$

$$= \frac{1}{EI} \left( \frac{1}{2} \times \frac{WL}{4} \times \frac{L}{4} + \left( \frac{WL}{4} \times \frac{L}{4} \right) \right)$$

$$\theta_D = \frac{3}{32} \frac{WL^2}{EI} (\curvearrowleft)$$

$$\theta_H = -\theta_D = \frac{3}{32} \frac{WL^2}{EI} (\curvearrowleft)$$

$$Y_D - Y_E = \frac{1}{EI} (A_1 \bar{x}_1 + A_2 \bar{x}_2) = \frac{1}{EI} \left[ \frac{WL^2}{32} \times \frac{L}{6} + \frac{WL^2}{16} \times \frac{3L}{8} \right]$$

$$Y_{max} = \frac{11}{384} \left( \frac{WL^3}{EI} \right) (\downarrow)$$

Find deflection at C  $\gamma_c = ?$

origin  $\rightarrow C$

ref.  $\rightarrow E$

$$\gamma_c - \gamma_E = \frac{1}{EI} (A\bar{x})$$

Consider Area b/w C & E

$$A = \frac{WL}{4} \times \frac{L}{4} = \frac{WL^2}{16}$$

$\bar{x} = \frac{L}{8}$  (From c)

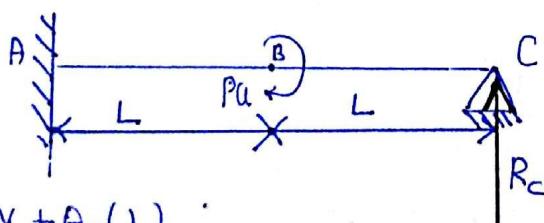
$$\gamma_c - \left( -\frac{11}{384} \frac{WL^3}{EI} \right) = \frac{1}{EI} \left( \frac{WL^2}{16} \times \frac{L}{8} \right)$$

$$\gamma_c = \frac{WL^3}{EI} \left( \frac{1}{128} - \frac{11}{384} \right)$$

$$\gamma_c = \frac{WL^3}{EI} \left( \frac{384 - 11}{384 \cdot 128} \right)$$

$$\gamma_c = + \frac{WL^3}{EI} \left( -\frac{8}{384} \right) = \frac{8}{384} \left( \frac{WL^3}{384} \right) (\downarrow)$$

Ques For the propped Cantilever beam as shown in Fig determine slope at B ( $\theta_B = ?^\circ$ )

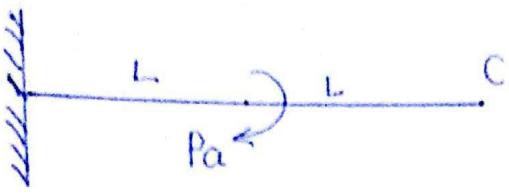


$$(\gamma_c)_1 = \gamma_B + \theta_B(L)$$

$$\frac{Pa L^2}{2EI} + \frac{Pa L (L)}{EI} = \frac{R_c (2L)^3}{3EI}$$

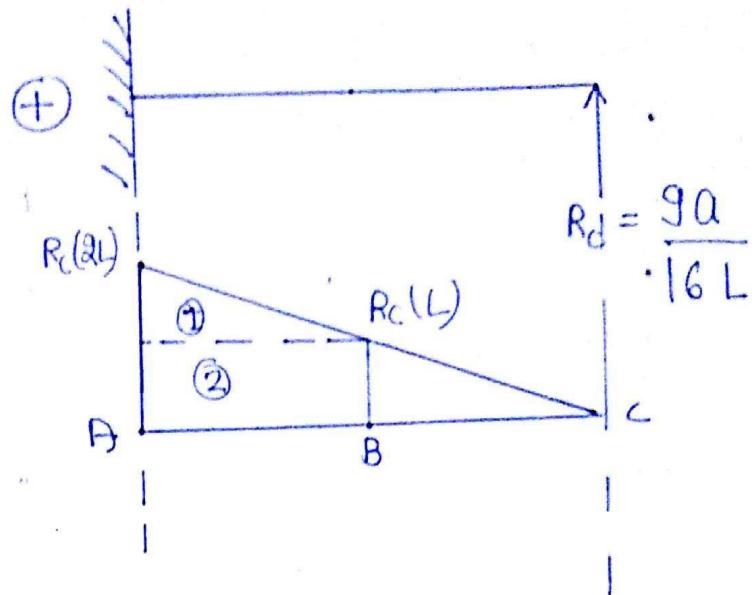
use principle of superposition  
defn at C  $\gamma_c = \gamma_{c1} + \gamma_{c2} = 0$

$$\frac{3Pa L^2}{8EI} = \frac{8 R_c L^3}{3EI} \Rightarrow R_c = \frac{9 Pa}{16 L}$$



$$\theta_B = \frac{ML}{EI}$$

$$(\theta_B)_1 = \frac{PaL}{EI} (2)$$



Region  $\rightarrow B$

Ref.  $\rightarrow A$

$$(\theta_B)_2 - (\theta_A)_1 = \frac{1}{EI} \left[ \left( \frac{1}{2} R_c L \times L \right) + (R_c L)^2 \right]$$

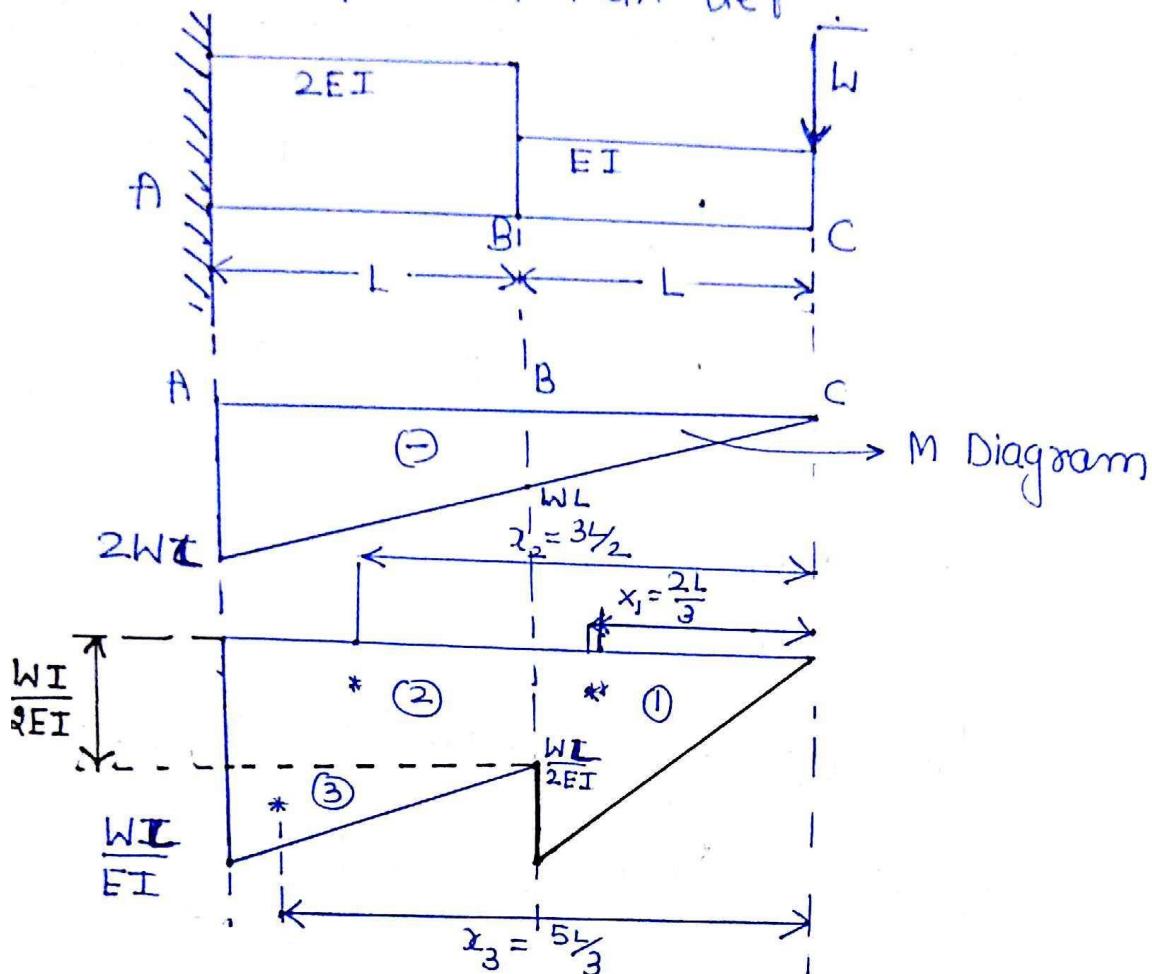
$$(\theta_B)_2 - 0 = \frac{3}{2} \frac{R_c L^2}{EI} = \frac{27 PaL}{32 EI} \cdot (2)$$

$$\theta_B = (\theta_B)_1 + (\theta_B)_2$$

$$\theta_B = -\frac{PaL}{EI} + \frac{27 PaL}{32 EI}$$

$$\theta_B = \frac{5 PaL}{32 EI} (2)$$

Ques. For the cantilever beam as shown in fig determine max. Slope and max. def'n



origin  $\rightarrow$  C Ref:- A

$$\theta_C - \theta_A = [A_1 + A_2 + A_3]$$

$$\theta_{\max} = \left[ \left( -\frac{1}{2} \frac{WL}{2EI} \times L \right) + \left( -\frac{WL}{2EI} \times L \right) + \left( -\frac{1}{2} \times \frac{WL}{2EI} \times L \right) \right]$$

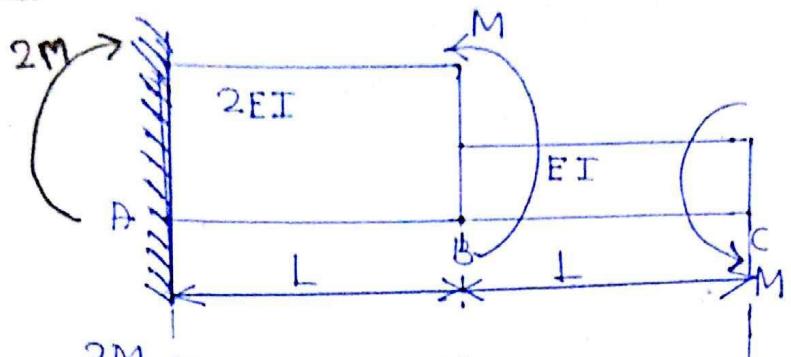
$$\theta_{\max} = \frac{5}{4} \frac{WL^2}{EI} \quad (\textcircled{2})$$

$$Y_c - Y_A = A_1 \bar{x}_1 + A_2 \bar{x}_2 + A_3 \bar{x}_3$$

$$Y_{\max} - 0 = \left[ \left( -\frac{WL^2}{2EI} \times \frac{2L}{3} \right) + \left( -\frac{WL^2}{2EI} \times \frac{3L}{2} \right) + \left( -\frac{WL^2}{4EI} \times \frac{5L}{3} \right) \right]$$

$$Y_{\max} = -\frac{3}{2} \left( \frac{WL^3}{EI} \right) \textcircled{1} \quad \frac{3}{2} \frac{WL^3}{EI} \textcircled{2}$$

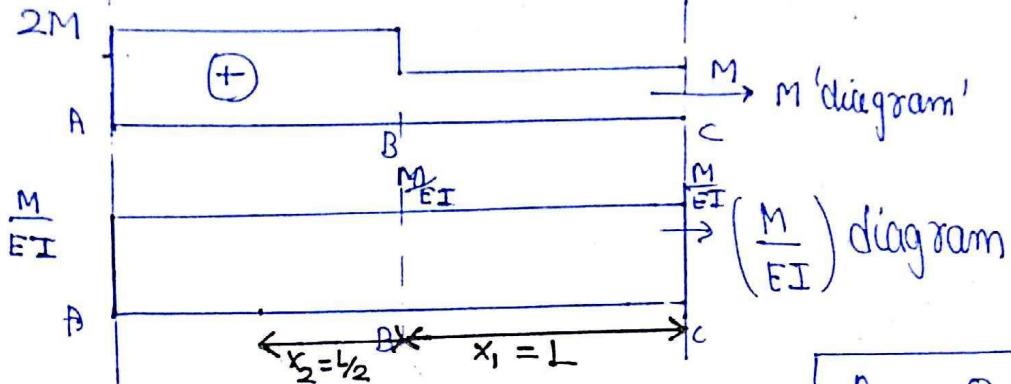
Ques.



determine

$$\textcircled{a} \quad Y_C \text{ & } \theta_C = ?$$

$$\textcircled{b} \quad \theta_B \text{ & } Y_B = ?$$



Origion - C, Ref - A

$$\theta_C - \theta_A = (2L) \frac{M}{EI}$$

$$\theta_C = \frac{2ML}{EI} (\rightarrow)$$

$$\bar{x}_1 = L \text{ (from C)}$$

$$Y_C - Y_A = \frac{2ML^2}{EI} (L)$$

$$Y_C = \frac{2ML^2}{EI} (\uparrow)$$

$$\bar{x}_2 = \frac{L}{2} \text{ (from B)}$$

$$\begin{cases} \theta_A = 0 \\ Y_A = 0 \end{cases}$$

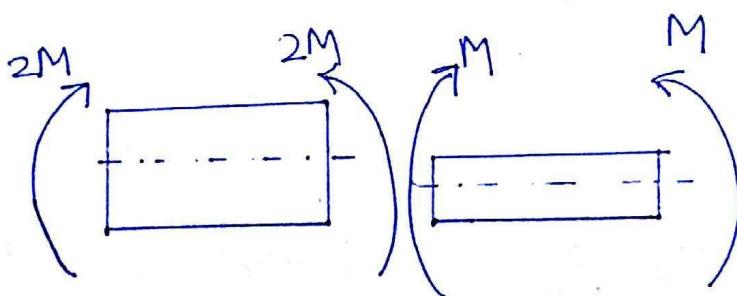
Origion - B

Ref - A

$$\theta_B - \theta_A = L \frac{M}{EI}$$

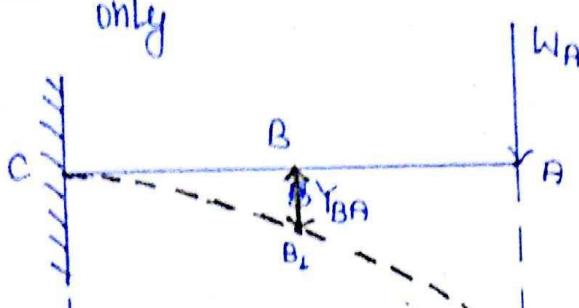
$$\theta_B = \frac{ML}{EI} (\leftarrow)$$

$$Y_B = \frac{ML}{EI} \left( \frac{L}{2} \right) = \frac{ML^2}{2EI} (\uparrow)$$

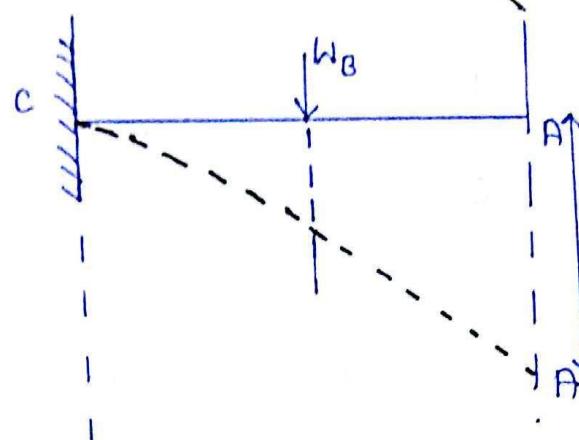


## Maxwell's Reciprocal Theorem: -

is valid for def<sup>n</sup> in presence concentrated point load



$Y_{BA}$  → load at A  
 $w_A$  → def<sup>n</sup> at B

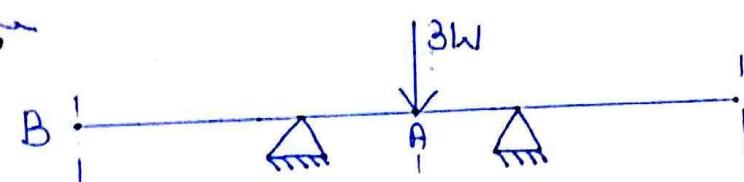


load is applied at 'B'  
 $Y_{AB}$  → Def<sup>n</sup> at x-s/c 'A'

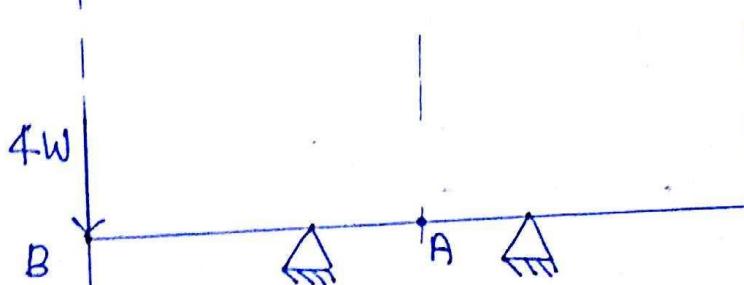
$$W_A \times Y_{AB} = W_B \times Y_{BA}$$

$$\text{if } W_A = W_B \Rightarrow Y_{AB} = Y_{BA}$$

Eg:-



$$\text{if } Y_B = \delta = \delta_{BA}$$



$$\text{def } Y_A = Y_{AB} = ?$$

$$W_A \times Y_{AB} = Y_B \times Y_{BA}$$

$$3W \times Y_{AB} = 4W \times \delta$$

$$Y_{AB} = \frac{4}{3} \delta = Y_A =$$

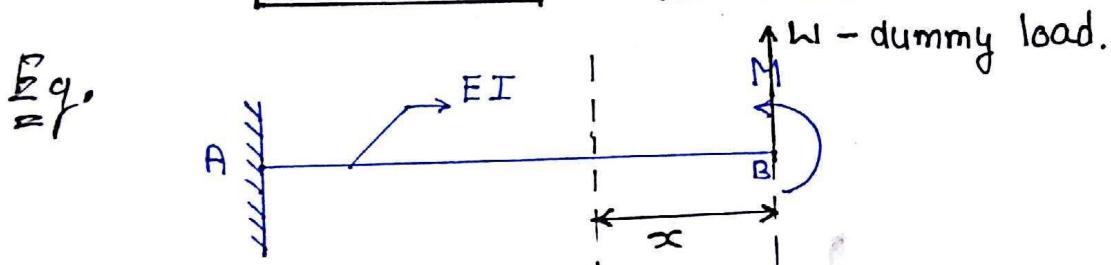
## Castigliano's Theorem! —

let  $U = S.E.$  due to B.M.

$$U = \int_a^b \frac{(M_{x-x})^2 dx}{2(EI_{N.A.})_{x-x}}$$

As per this theorem

$$\Theta_A = \frac{\partial U}{\partial M_A} : Y_A = \frac{\partial U}{\partial W_A}$$



$$U = \frac{M^2 L}{2 E I_{N.A.}}$$

$$\Theta_B = \frac{\partial U}{\partial M_B} = \frac{\partial}{\partial M} \left[ \frac{M^2 L}{2 E I_{N.A.}} \right]$$

$$\Theta_B = \frac{ML}{E I_{N.A.}} \quad \text{Conc. Pt.}$$

To det.  $Y_B$ , introduce a dummy load such that it will cause B.M. in same dirn. at B

$$M_{x-x} = M + Wx$$

$$U_{x-x} = \int_0^L \frac{(M + Wx)^2 dx}{2EI}$$

$$U = \frac{1}{2EI} \int_0^L (M^2 + W^2 x^2 + 2MWx) dx$$

$$U = \frac{1}{2EI} \left[ M^2 L + \frac{W^2 L^3}{3} + MWL^2 \right]$$

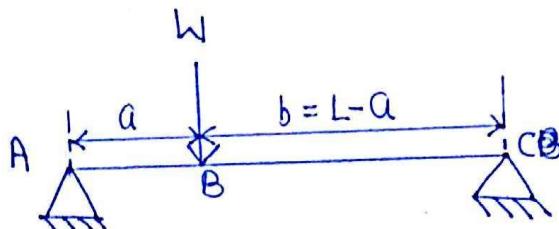
$$Y_B = \frac{\partial U}{\partial W_B} = \frac{1}{2EI} \frac{\partial}{\partial W} \left[ M^2 L + \frac{W^2 L^3}{3} + MWL^2 \right]$$

$$Y_B = \frac{1}{2EI} \left[ 0 + \frac{2WL^3}{3} + ML^2 \right]$$

To get actual  $Y_B$ , sub  $W = 0$

$$Y_B = \frac{ML^2}{2EI}$$

Eg.

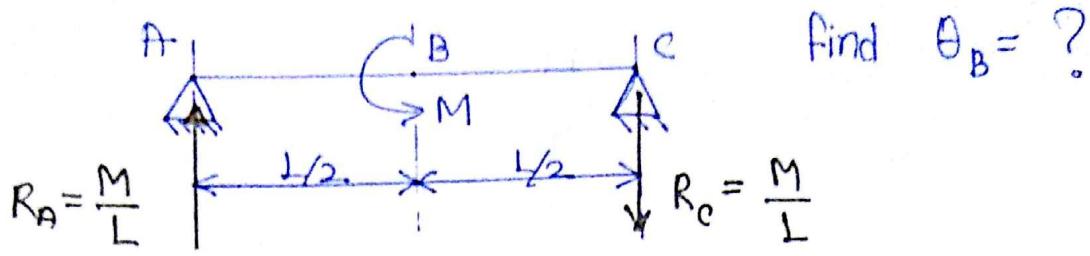


$$U = \frac{W^2 a^2 b^2}{6EI_{N.A.} L} \quad U = U_{AB} + U_{BC}$$

$\Rightarrow$  Go back & check notes  
"strain energy"

$$Y_B = \frac{\partial U}{\partial W} = \frac{Wa^2 b^2}{3EI_{N.A.} L}$$

Eg.



$$U = U_{AB} + U_{BC}$$

$$M_{x-x} = \frac{M}{L} x$$

$$U = \frac{2}{4} U_{AB}$$

$$U_{x-x} = \int_0^x \frac{(M_{x-x})^2 dx}{2EI}$$

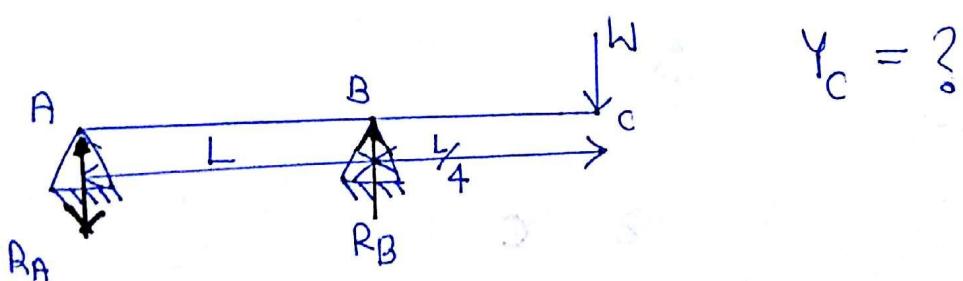
$$U = \frac{2}{2} \int_0^{\frac{L}{2}} \frac{\left(\frac{M}{L}x\right)^2 dx}{2EI}$$

$$U = \frac{M^2}{EI L^2} \frac{(L/2)^3}{3}$$

$$U = \frac{M^2 L}{24EI}$$

$$\theta_B = \frac{\partial u}{\partial M} = \frac{\partial ML}{24EI} = \frac{ML}{12EI}$$

Eg



$$R_A + R_B = WL \quad \text{---(1)}$$

$$\sum M_A = 0 \Rightarrow W\left(L + \frac{L}{4}\right) - R_B L = 0$$

$$\frac{5WL}{4} = R_B L$$

$$\Rightarrow R_B = \frac{5W}{4} (\uparrow); R_A = \frac{WL}{4} (\downarrow)$$

$$U = U_{AB} + U_{BC}$$

$$U = \int_0^L \frac{(-\omega x)^2}{2EI} dx + \int_0^{l_4} \frac{(\omega x)^2}{2EI} dx$$

$$U = \frac{\omega^2}{32EI} \frac{L^3}{3} + \frac{\omega^2}{2EI} \times \cancel{\frac{(L^3)}{3}} \frac{(l_4)^3}{3}$$

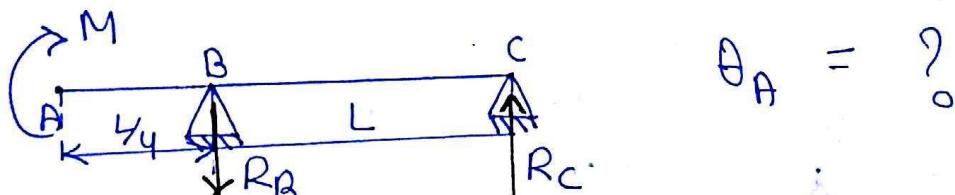
$$U = \cancel{\frac{\omega^2 L^3}{EI}} \left( \cancel{\frac{1}{96}} + \cancel{\frac{1}{6}} \right)$$

$$\cancel{\frac{1+16}{96}}$$

$$U = \frac{5}{384} \frac{\omega^2 L^3}{EI}$$

$$Y_C = \frac{dU}{d\omega} = \frac{5}{192} \frac{\omega L^3}{EI}$$

Eg



$$\theta_A = ?$$

$$R_{AC} = R_B = \frac{M}{5l_4} \Rightarrow R_C = R_B = \frac{4M}{5L}$$

$$U = U_{AB} + U_{BC}$$

$$U = \frac{M^2(l_4)}{2EI} + \int_0^L \frac{(R_C x)^2 dx}{2EI}$$

$$U = \frac{M^2 L}{8EI} + \frac{R_C^2}{2EI} \frac{L^3}{3}$$

$$U = \frac{M^2 L}{8EI} + \left(\frac{4M}{5L}\right)^2 \frac{L^3}{6EI}$$

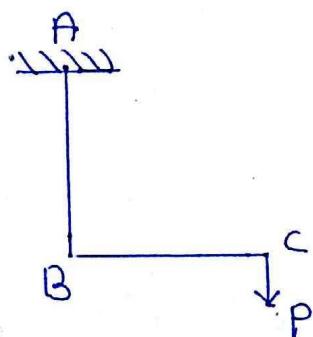
$$U = \frac{M^2 L}{EI} \left( \frac{1}{8} + \frac{16}{150} \right)$$

$$U = \frac{M^2 L}{EI} \left( \frac{150 + 128}{8 \times 150} \right)$$

$$U = \frac{\frac{139}{600}}{\frac{278}{600}} \frac{M^2 L}{EI} = \frac{139}{600} \frac{M^2 L}{EI}$$

$$\theta_A = \frac{139}{300} \frac{ML}{EI}$$

Eg.

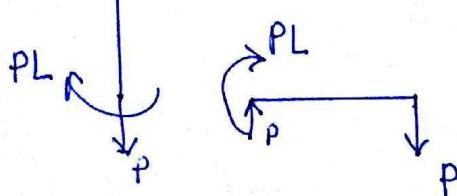


$$U = U_{AB} + U_{BC}$$

$$U = \left( \frac{M^2 L}{2EI} \right)_{AB} + \int_0^L \frac{(Px)^2 dx}{2EI}$$

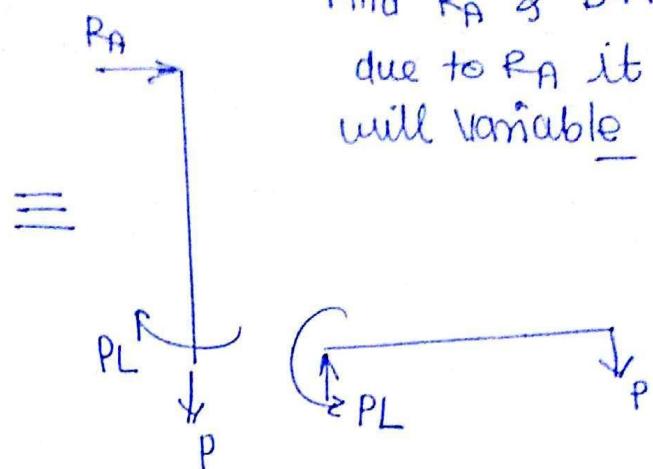
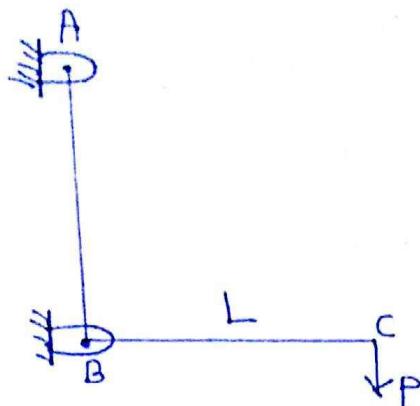
$$U = \frac{(PL)^2 L}{2EI} + \frac{P^2 L^3}{6EI} = \frac{2}{3} \frac{P^2 L^3}{EI}$$

$$Y_C = \frac{\partial U}{\partial P} = \frac{4PL^3}{3EI}$$

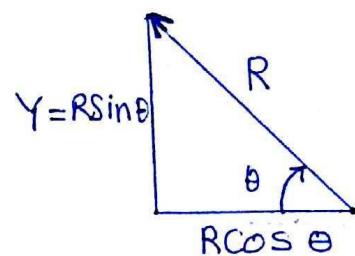
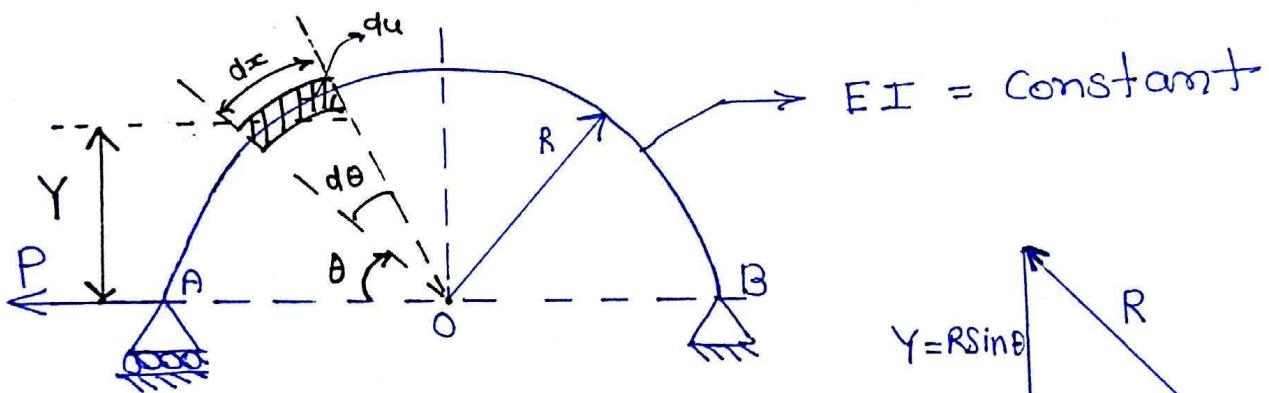


J.F.F

Gate  
see sol<sup>n</sup>



Quest Determine axial displacement at A, Semicircular ring.



$$\theta = \theta_0 + \frac{\pi}{2}$$

$$M_{x-x} = PY = PR \sin \theta$$

$$dx = R d\theta$$

$$U = \int_0^{\frac{\pi}{2}} \frac{(PR \sin \theta)^2 (R d\theta)}{2EI_{N.A.}} = \frac{P^2 R^3}{2EI} \int_0^{\frac{\pi}{2}} \left( \frac{1 - \cos 2\theta}{2} \right) d\theta$$

$$U = \frac{P^2 R^3}{2EI} \left( \frac{\pi}{2} \right) = \frac{\pi P^2 R^3}{4EI_{N.A.}}$$

$$\delta_A = \frac{\partial U}{\partial P} \Rightarrow \delta_A = \frac{\pi P R^3}{2EI_{N.A.}}$$